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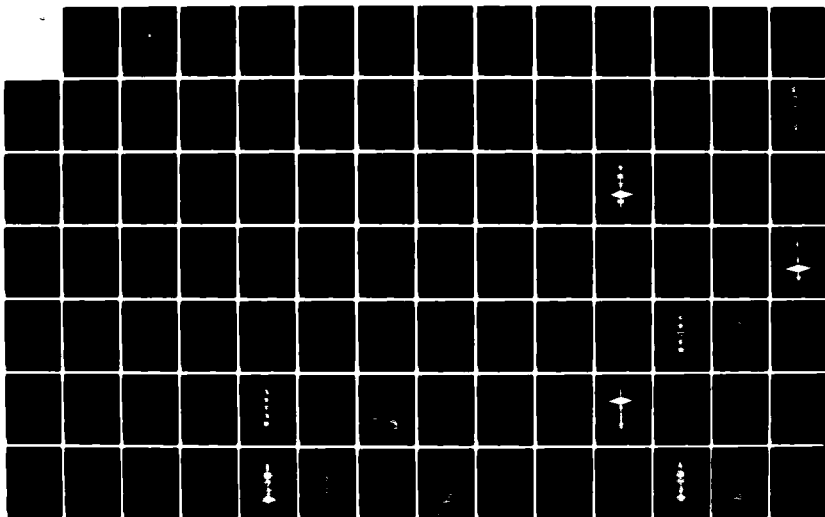
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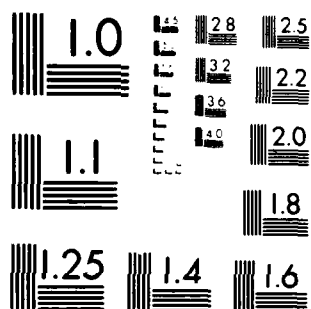
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MULTIPLE TARGET IDENTIFICATION
AND DIRECTION FINDING
USING MATCHED FILTERING TECHNIQUES

by

James L. Johnston

December 1983

Thesis Advisor:

H. A. Titus

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Multiple Target Identification and Direction Finding
Using Matched Filtering Techniques

by

James L. Johnston
Captain, United States Marine Corps
B.S.E.E., San Diego State University, 1975

Submitted in partial fulfillment of the
requirements for the degree of

MASTER OF SCIENCE IN ELECTRICAL ENGINEERING

from the

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ABSTRACT

This research investigates seismic signal processing techniques for battlefield target classification and acquisition. Multiple target classification is performed by discrete time domain matched filtering. Multiple target directions are determined using the responses of the matched filters and least mean squares polynomial curve fitting. The least mean squares polynomial curve fitting procedure is also used for direction finding for recoil/blast sources, using the unfiltered seismic signals.

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I. INTRODUCTION

Timely and accurate combat intelligence is an integral part of the modern battlefield. A primary goal of combat intelligence is target acquisition. Hostile targets can be acquired by either passive or active means. An example of passive target acquisition is visual target identification. Radar, on the other hand, is an active target acquisition device. An effective combat intelligence system will include a mix of both active and passive target acquisition methods.

Rapidly advancing technology in the fields of electronic counter-measures and radiation-seeking weapons has enhanced interest in passive target acquisition methods. To be cost effective, as an additional target acquisition system, a passive system must be able to provide swift and accurate target identification, location and tracking information on hostile targets. A variety of seismic sensor systems have been used in this roll with varying degrees of success.

Naval Ocean System Command (NOSC) in San Diego has developed a system based on a circular ring of sensors with data collection managed by an array-processor/minicomputer system [Ref. 1]. The observation of enemy movements and activity beyond the Forward Edge of the Battle Area (FEBA) is the design objective of this system. This thesis uses data collected during a test of this system at the Marine Corps Air-Ground Combat Center at Twenty-nine Palms, California. Investigated are various methods of processing the seismic data collected. The objective of this research is to try to provide viable methods of satisfying system design objectives through signal processing techniques.

The following chapter addresses the design objectives and requirements for such a seismic sensor system. Additionally, capabilities and deficiencies of current systems and research are detailed. In order to intelligently address solutions to these requirements and deficiencies, an understanding of seismic theory and sensors is needed. General seismic theory is presented with emphasis on the constraining parameters for the use of the earth's surface as a medium for gathering seismic intelligence. Also highlighted are the similarities of the earth's surface to electro/optical phenomena and the resulting simplifying assumptions.

The sequence of design solutions investigated followed from the analogies and simplifying assumptions addressed in the study of seismic sensor theory. The problem of target identification or classification is approached using digital matched filtering of the time domain amplitude data. Frequency domain matched filtering was not considered, based upon the conclusion by NOSC that there appeared to be no consistent spectral lines for any of the possible target types, except for artillery [Ref. 1]. Matched filtering was used to identify single and multiple target classes occurring during a sample period. The chapter detailing the matched filter procedures and implementation also includes a description of support and validation software used in the analysis of the seismic data.

The validation software is primarily used to check the accuracy of the direction finding routines. These direction finding routines are the time domain phase difference procedure (TDPD) and a least mean squares polynomial (LMSP) curve fitting procedure. The combining of the matched filtering procedure and these direction routines allowed for multiple target direction finding. The theory and derivations of the two direction finding algorithms are presented in the multiple target direction chapter.

Application of these algorithms is performed first on simulated targets for validation of the procedures. The experimental data is then analyzed. A user's manual, which includes procedures for tape and mass storage operations, is provided as an appendix. This appendix describes how to set up and use the software system.

II. THEORY OF SEISMIC SENSORS

Elastic waves result from the stressing of an elastic media. The elastic media for seismic theory is the earth. Seismic theory is the study of the earth as a wave propagating media. Elastic waves propagate away from the source of seismic stress, e.g. an explosion [Ref. 2]. The energy, which propagates through the earth, travels via particle deformations. The elastic properties and densities of the earth media determine the velocities of these seismic waves. [Ref. 3]. Seismic wave sources of interest may be impulsive or continuous. Impulsive or short duration sources are artillery recoil or shell blasts. The time-limited nature of this type of seismic signal produces a broad range of frequencies. Continuous wave signals may be produced by tanks, trucks and low flying aircraft. These continuous wave signals may be described by narrow band frequency characteristics. The spectral power of a seismic source is a function of several parameters. A non-inclusive list of these parameters includes:

1. The vehicle's velocity and mass
2. The size of the explosive charge associated with the artillery or shell blast
3. The degree of coupling into the earth's surface
4. The geological structure over the wave's path

Information about the seismic source is contained in the waves which it generates. For example, in an array of seismic sensors, directional information is contained in the relative received signal phases. In otherwords, the relative phase differences between the signals received by the individual array elements can be used to compute the direction to the seismic source [Ref. 4]. These relative phase

differences represent the time delays of the waves as they pass the array's elements. The response of the array's elements is proportional to the amplitude and velocity of the earth's motion, relative to the geophone's sensing axis for the waves. [Ref. 3]

Assumptions about the propagation of seismic waves must be made to assist with their analysis. The earth is assumed to be made up of horizontal, homogeneous, and isotropic layers of material. These layers are assumed to be discontinuous in their elastic properties at their borders. This variance in the elasticity between the layers leads to an optical analogy for the wave propagation. Propagation paths may now be viewed as being direct, refracted, or reflected versions of the source's seismic waves. [Ref. 3]

There are four basic types of seismic waves. These types are compressional, shear, Love, and Rayleigh waves. Compressional waves are generated by impulsive sources such as shell blasts. Particle motion is along the direction of travel. Shear waves are characterized by particle motion orthogonal to the direction of motion. The Love wave is a surface wave which may occur as a result of the layering of the earth's surface. This layering effect acts as a wave guide for this type of wave. Particle motion is orthogonal to the direction of wave propagation. The Rayleigh wave is generally the strongest of the seismic waves. The Rayleigh wave travels along the free surface of the earth. Its particle motion direction is always in the vertical direction. It is the strongest wave generated by a compressional source. Its amplitude attenuates at a rate only inversely proportional to the square root of the distance. [Ref. 3]

The waves of primary interest for a seismic system are the Rayleigh and Love waves. This is due to the long-range propagation of these waves. Since these two wave types are orthogonal to each other, they must be sensed by different

geophones. Rayleigh waves may be sensed by vertical geophones and the Love waves by horizontal geophones. Since the Rayleigh wave is normally the strongest, and in order to reduce computational complexity, only verticle sensor data is used.

Rayleigh waves experience absorbtion losses, particularly at higher frequencies. This phenomenon occurs because of the lowpass filter effect of the earth. This filtering effect is further compounded due to the fact that the cutoff frequency of the earth diminishes with range. Further complications arise due to the dependence of wave velocity on frequency. The result being that the wave train may change with distance, reducing the correlation of the wave shape between its source and distant points. Other sources of error occur because of the weathering of the surface layer, irregularities in the sub-surface composition, variances in the earth's layers and surface geometry. [Ref. 3]

III. DEVELOPMENTAL REQUIREMENTS AND PROBLEM DEFINITION

A. PASSIVE TARGET ACQUISITION AND SURVEILLANCE

A battlefield commander possesses a definite requirement for real time combat intelligence. A significant tactical advantage is held by the commander who is able to integrate his available combat intelligence sources with his supporting arms, i.e., target acquisition and engagement. It follows that to be part of the target acquisition process, any real time, seismic sensing system must provide swift and accurate information on detected enemy targets. The specific requirements for such a system are the ability to detect, identify, and locate these targets [Ref. 5]. Additionally, the target's rate and direction of movement should be provided or made easily discernible.

Any seismic sensing system must be designed around the target acquisition cycle. The target acquisition cycle, as given by Dublin [Ref. 5], is as follows:

1. Search Time
2. Target Sensing
3. Information Processing
4. Display of Target Information
5. Analysis of Target Information
6. Time required to make a Decision
7. Time Required for Supporting Weapons to Respond

For a seismic system, a prioritized list of possible targets are as follows:

1. Artillery
2. Helicopters and Aircraft
3. Tracked and Wheeled Vehicles
4. Personnel

As may be expected, the relative amplitudes of these seismic targets vary widely. A seismic targeting system is therefore constrained as to the targets it can or can not be expected to effectively engage.

The variance in the relative amplitudes of seismic targets suggests a range of specifications for detection of these targets. As summarized by Dublin [Ref. 5], possible detection radii may be as shown in Table I. Radii are given for both short and long targeting systems.

TABLE I		
Target Detection Radii		
Target	Short Range System	Long Range System
Personnel	100M	None
Vehicles	1KM	10-20KM
Low Flying Aircraft	1KM	10-20KM
Hostile Weapons	1KM	15-20KM

The timeliness requirement is ancillary to the radii of detection specifications. Timeliness, as used here, refers to the total time commencing when the seismic sensor system first detects and processes the seismic target data and ending with the dissemination of the targeting information to command elements for disposition. This timeliness requirement ranges between five to fifteen minutes, depending upon the mobility of the target [Ref. 5].

The parameters having a direct effect on the timeliness of a system are the probabilities of false alarm and detection. These parameters directly relate to a system's value. Increased probability of detection with reasonable false alarm performance, combined with the ability to disregard

friendly targets, are practical design objectives for any targeting system. The ability to incorporate such design features into a seismic sensor system will reduce both the time wasted on invalid targets and the danger of undetected targets.

Once a valid target has been detected, target location information must be obtained. Stringent specifications for target locations allow for the system to support or enhance the effectiveness of: fire support systems, blind bombing, Harassing and Interdiction fire (H and I), and observerless artillery engagement.

B. CURRENT CAPABILITIES AND DEFICIENCIES

To date, numerous successful algorithms have been developed to determine direction to single targets. These algorithms include both time and frequency domain methods. Target identification of long range targets via seismic sensing has not yet met with equal success.

The modern battlefield is seldom a single target type environment. The complicated, real world problems of multiple target identification and multiple target engagement require solution before practical seismic sensor systems can be integrated into the target acquisition process.

IV. MATCHED FILTER CONCEPT AND DISCRETE ALGORITHM

A. MATCHED FILTER THEORY

As previously addressed, there exists a requirement for battlefield target identification/classification. The recovery and classification of target signals suggests a filtering requirement. Previous works and implementations have used frequency domain techniques [Ref. 4]. The Air Force's SKEET system and the U.S Army's Remote Battlefield Surveillance System (REMBASS) both have successfully implemented a spectral power approach for classification of seismic data. These systems, however, are for short range applications. Time domain approaches to target identification have been for the most part left unexplored.

The discrete matched filtering technique is an attempt to classify targets by their time domain amplitude pattern i.e., their seismic amplitude signature. The heuristic basis for this method evolved from the observation of visual differences between the amplitude versus time signals for the various classes of targets. The matched filter, being the optimum filter for detecting known signals, was selected [Ref. 6].

The discrete matched filter is described by equation 4.1, where $h(t)$ is the impulse response of a filter whose output signal to noise ratio at time t_0 is maximized. The unit step $u(t)$ has been added to assure causality for the system. The matched filter is [Ref. 7].

$$h(t) = s(t_0 - t)u(t) \quad (4.1)$$

The output signal is given by

$$s_{out} = \int_{-\infty}^{\infty} h(v) s(t - v) dv \quad (4.2)$$

The maximum signal to noise ratio is given by

$$(s_{out}^2/M^2)_{maxoutput} = E(t_0)/N_0 \quad (4.3)$$

Where M is the noise level at the filter output, N is the input noise level, and $E(t_0)$ is the energy in $s(t)$ up to time t_0 . In equation 4.2, the replica of the original known signal is reversed and translated in time to be convolved with the signal input to the filter, producing the optimum output signal to noise ratio.

For discrete realization of the matched filter as implemented, equation 4.2 becomes equation 4.4, where $h(k)$ is the reflected and translated known signal.

$$s_{out}(j) = \sum_{k=1}^N h(k) s(j - k) \quad (4.4)$$

Where N is the number of data points per sample period.

B. MATCHED FILTER ALGORITHM

Samples of known signals are stored in a data file and read into a 5120 array at the start of program execution. Five sample, or known filter signals, are recorded in this array. Each sample signal comprises 1024 of the 5120

elements of the matched filter array. To perform the matched filtering, the 1024 seismic data samples (unknown signal) are copied over five times on an array of 11264 size. These copies of the experimental data are separated by 1024 zeros on either side. Additionally, the leading and trailing elements of the experimental data are set to zero to eliminate switching spikes present in the data. Hence forth, this 11264 element array will be referred to as the working array for brevity. This working array will contain the results of the matched filtering. For M , given as the number of input signal and filter signal array elements and also the number of zeros, the requirement set forth in [Ref. 8] for nonoverlapping convolution of length L is satisfied. Equation 4.5 establishes the minimum length for nonoverlapping convolution of one matched filter segment.

$$L = 2M - 1$$

(4.5)

Figure 4.1 depicts the working array layout. Notice that the first filter signal, "Tracked Vehicle" is loaded into the $h(t)$ array and is convoluted with the working array elements one through 2048. Once the leading data point of $h(t)$ reaches the working array's element 2048, a new known signal is loaded into $h(t)$, "Wheeled Vehicle", and the convolution is continued for working array elements 2048 to 4096. This process continues until the last of the five $h(t)$ filter signals has been read in and the convolution has been performed on all copies of the data. Notice that working data array elements 10240 to 11264 are for array length over-run protection.

Prior to convolution, each filter sample signal and the input signal are equalized to the same power level through

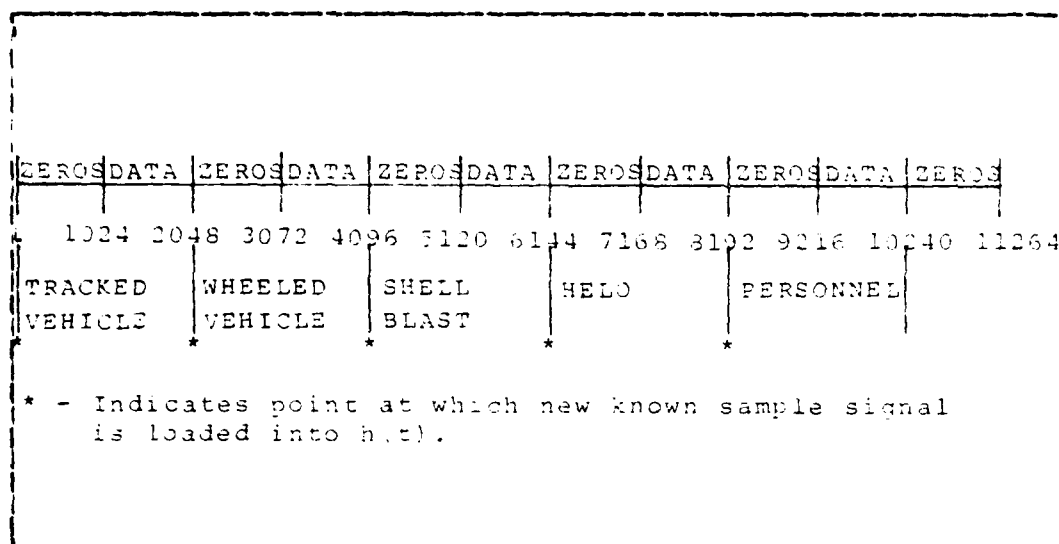


Figure 4.1 Working Array Configuration

division by their respective root mean square values. Additionally, after the convolution, the entire working array is normalized with respect to its maximum amplitude element. The maximum value in each target classification section is then compared with the interactively selected matched filter threshold. If the section's peak value is above the threshold value, that class of target is declared to be present. This method allows for the simultaneous detection and classification of multiple targets. It should be noted that this equalization forces both high and low amplitude signals to an equivalent average power level. This is felt to be justified in that signal amplification is not the goal, rather target detection and classification is. In the case of a single target, the known signal that most closely matches the unknown signal is anticipated to have the greatest amplitude matched filter spike. Figure 4.2 is a sample output.

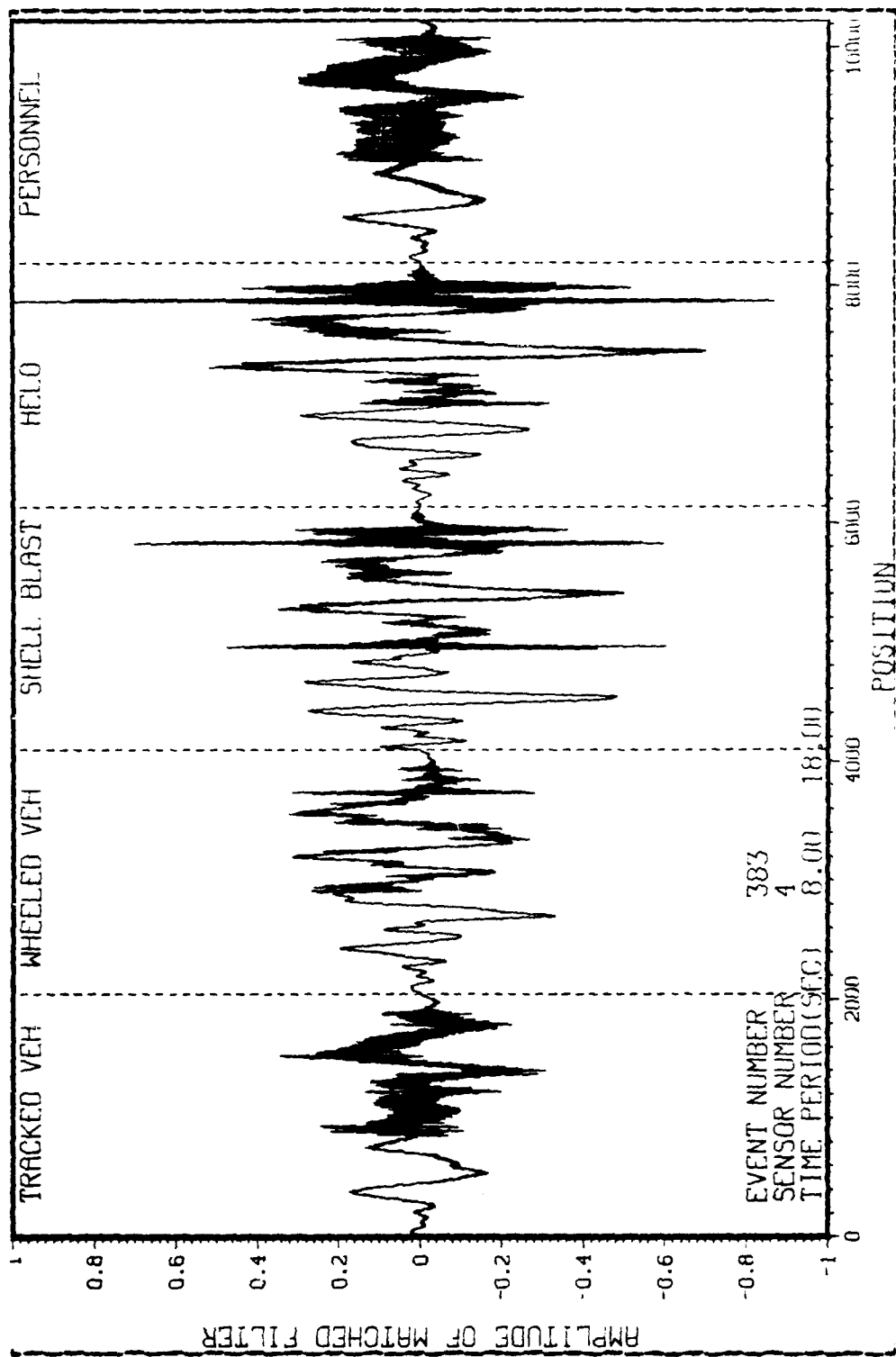


Figure 4.2 Sample Matched Filter Output

C. MULTIPLE TARGET MATCHED FILTERING

The upper limit of equation 4.4 may be selected to be from one to 1024 when called by the multiple target direction routine. This allows for the selection of reduced program execution times. Target identification, however, is always made by the full 1024 element buffer. When the matched filter target identification routine is used by the multiple target direction finding routine, data windows of less than 1024 are formed from a segment of the 1024 elements by extracting the segment size required around the maximum signal value. Figure 4.3 illustrates this procedure.

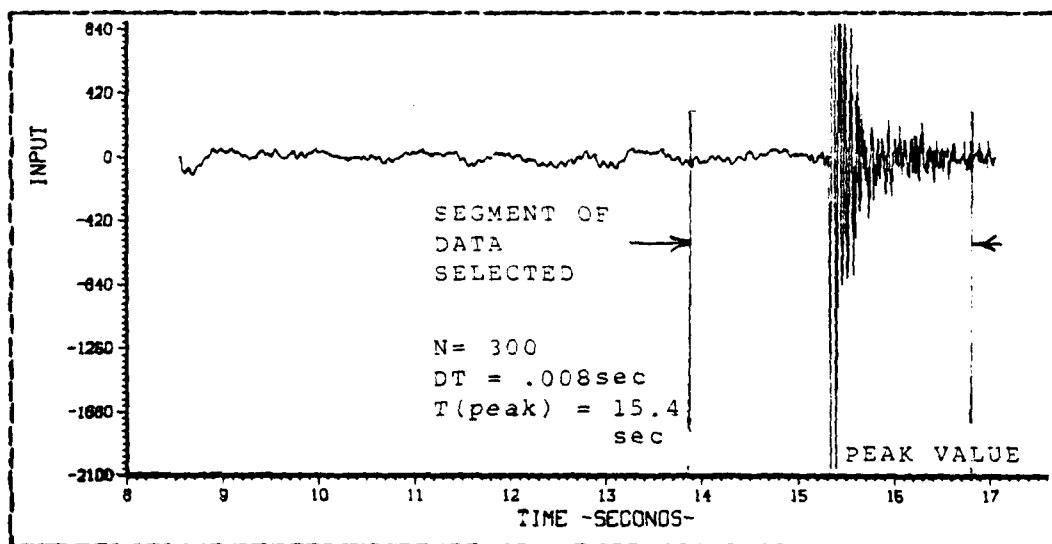


Figure 4.3 Windowing of Experimental Data

D. COMPUTATIONAL REQUIREMENTS

The computational requirements for this procedure are described by equation 4.6, where N_0 is the total number of operations required by the matched filter routine.

$$NO = (Ndf^2)(Nc) + (Ndf - 1)(Ndf)(Nc) \quad (4.6)$$

Ndf is the number of data elements and also the number of filter elements. Nc is the total number of target classes. The term Ndf^2Nc is the total number of multiplications required, while $(Ndf - 1)NdfNc$ is the total number of additions.

When used for target identification, Ndf equals 1024 and Nc equals five. Equation 4.6 gives a total of 10,480,640 operations for these array sizes. When the matched filter routine is initiated by the multiple target direction routine, the value of Ndf can be selected to range from one to 1024. Figure 4.4 is a plot of equation 4.6 and shows the computational consequence for selection of large values for Ndf. Note that equation 4.6 must be multiplied by the number of sensors in the ring when computing the number of operations for the multiple direction routine.

Window sizes of 100 and 200 were found to provide acceptable accuracy with greatly reduced computation times. As shown by figure 4.4, a window size of 200 requires 399,000 operations, while a window size of 100 requires only 99,500 operations. The number of operations required was found, as expected, to be proportional to the execution time of this routine.

B. SUPPORT SOFTWARE

1. Amplitude Analysis (Timeout)

A graphical output is provided by this amplitude routine which displays the relative amplitude versus time for a selected sensor. The initial target direction and target classifications found are also displayed. This routine allows for the interactive selection of the

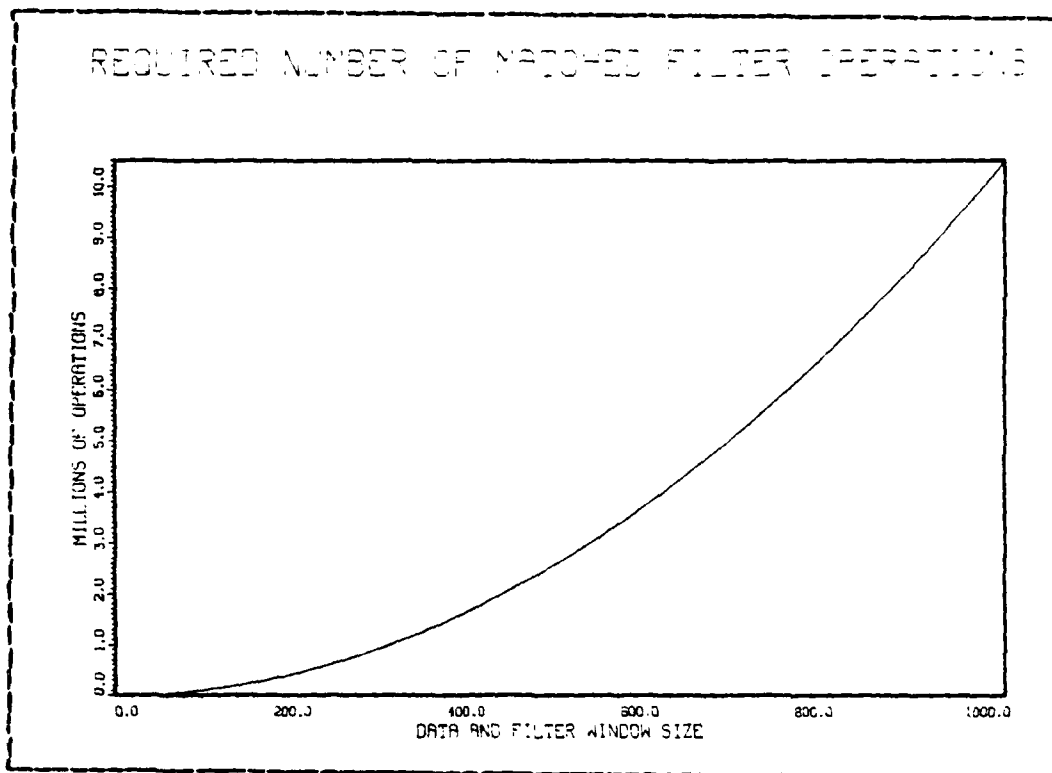


Figure 4.4 Data Window Size vs Number of Operations

amplitude response of any sensor as a sample target for later use in the matched filter analysis. In this way sample, signals can be catalogued and evaluated as filter signals. The axes of the graphical output adapts to the data's maximum amplitude and to the time period involved. Figure 4.5 is an example of amplitude analysis output.

2. Frequency Analysis (Freqot)

The frequency routine, as in the amplitude analysis routine, allows for initial primary target direction, target classifications, and for the time period of the experimental data to be displayed. Normalized spectral power versus frequency is graphically displayed as shown in figure 4.6.

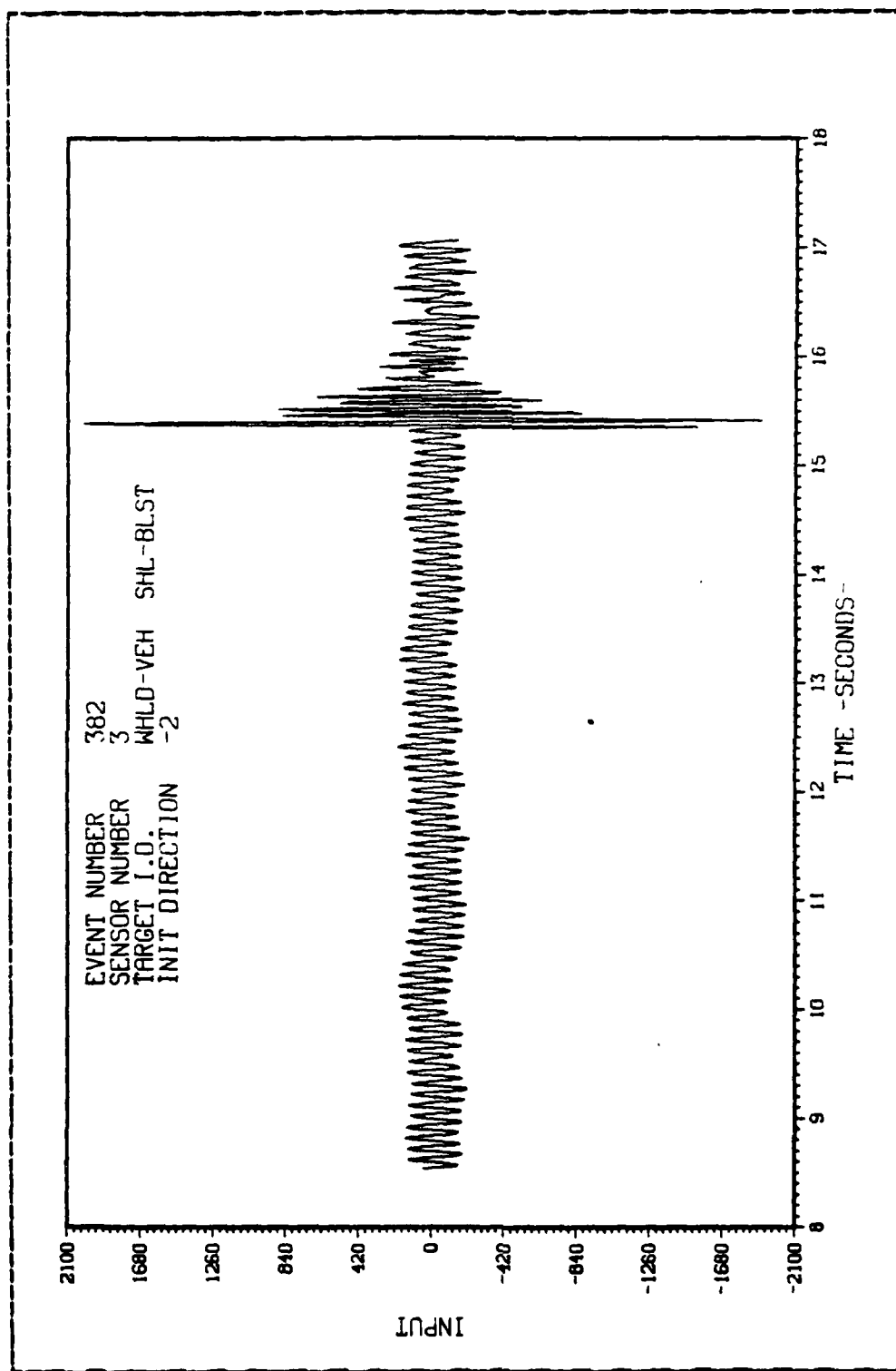


Figure 4.5 Sample Amplitude versus Time Output

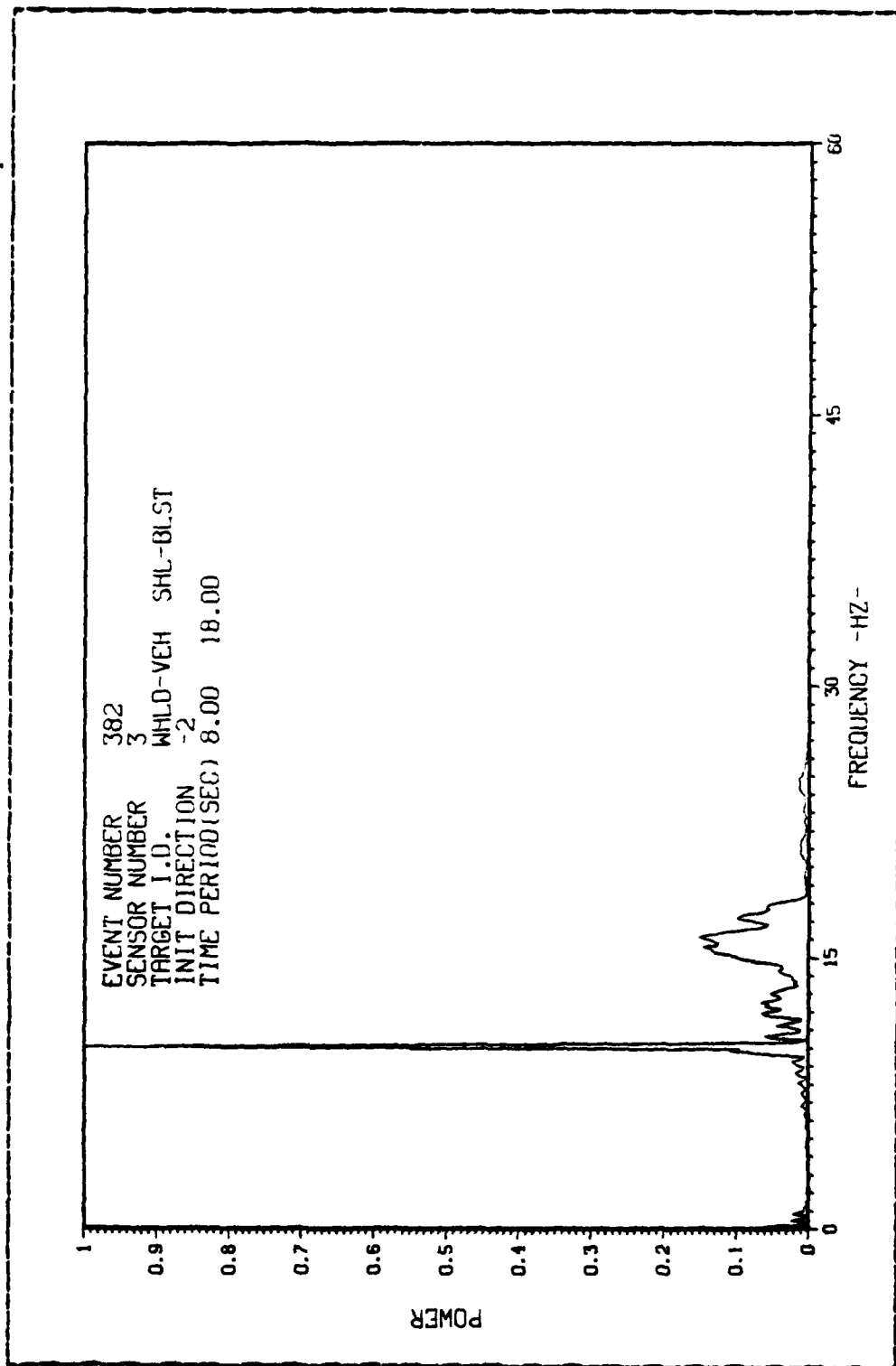


Figure 4.6 Sample Frequency versus Power Output

3. Simulation and Validation Software (SIMULT)

In order to validate the various algorithms and their implementation in software, a testing procedure was required. The specific algorithms which the simulation routine was designed to validate are the initial angle, phase difference and least mean square target direction routines. These routines, as previously described, use the relative time differences of the seismic signal's peak amplitude response or the peak matched filter response respectively. Validation of these routines is performed by allowing the creation of simulated targets with selected arrival angles.

Simulated targets can be created in the routine SIMULT. Up to four sine wave targets of selected frequency, amplitude, and direction can be input during each sample period. These simulated targets are added to the experimental seismic signal data for the sample period. Correspondingly, zero direction filter data must have been written into the matched filter data file at these selected sample target frequencies.

The directions for the simulated targets are created by introducing relative phase delays between the sine waves that are added to each sensor's seismic data. This is implemented by introducing a zero phase to the sensor in the desired direction of the simulated target. The phases of the other sensors are increased proportionally by their distance on the circular array away from the zero phase shift sensor. Figure 4.7 illustrates the case of a zero degree simulated target direction.

A summary of the target directions found by the multiple target direction routine and the simulated targets entered is provided by the multiple direction plotting routine. Figure 4.8 is a sample output.

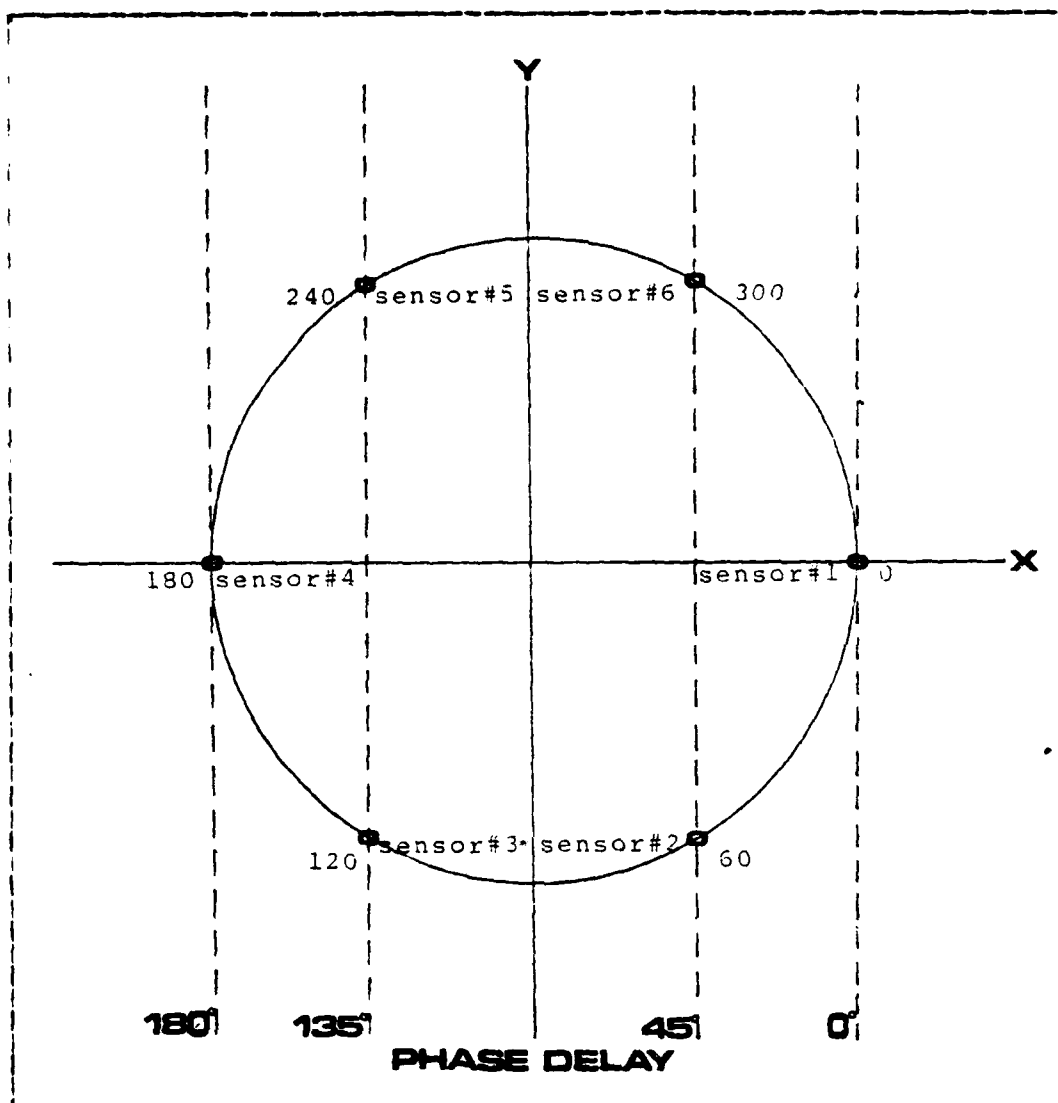


Figure 4.7 Simulation of a Zero Degree Target

MULTIPLE TARGET - MATCHED FILTER OUTPUT

EVENT NUMBER	375
TIME PERIOD(SEC)	25.00 35.00
WHEELED VEHICLE	DIRECTION - -59.00
SHELL BLAST	DIRECTION - 315.00
PERSONNEL	DIRECTION - -59.00
SIMULATED TRKD VEHICLE	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED WHLD VEHICLE	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED HELICOPTER	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED PERSONNEL	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000

Figure 4.8 Sample Multiple Target and Simulated Target Output

V. MULTIPLE TARGET DIRECTION

A. THEORY AND DESIGN CRITERION

The modern battlefield is comprised of many classes and quantities of seismic signals. Using these signals for target identification and acquisition is the purpose of battlefield seismic sensors. As presented in the last chapter, matched filters can be used for target identification. In this chapter, it will be shown that matched filter information may also be used to obtain target bearing.

Multiple target acquisition requires the concurrent separation of the target classes and the computation of direction for each of the target classes found. The output of the matched filter is a spike at t_0 , the time of peak signal detection. If matched filtering is performed for each of the sensors in the ring, the value of their respective t_0 , for each filter output, would be different. Since the seismic waves impinge upon each sensor at different times, dependent upon the target direction, the values for t_0 for each sensor may be expected to be directly related to the arrival angle of the seismic wave. The t_0 spike of the matched filter output may be thought of as a signal compression for both the continuous (tank, truck etc.) and the time limited (shell blast, artillery recoil etc.) seismic signals [Ref. 6].

Time domain methods [Ref. 1] use the positional differences of known wave points to geometrically estimate direction. The times associated with a sensor ring's peak amplitude responses may be explained by way of an illustrative example of a shell blast. A rough direction to the origin of this shell blast can be computed using the

relative time differences associated with the peak amplitudes of all of the ring's sensors [Ref. 9].

The enhanced time positional information, which is a by-product of the matched filtering, can be used to perform just such a time domain approach. The motivation for the method is that, unlike other time and frequency domain methods, which are very susceptible to noise corruption inaccuracies, matched filtering pulls the signal out of the noise and optimally detects the signal at time t_0 . Two time domain methods are evaluated for finding arrival angles of the seismic signal. The first being the time domain phase difference (TDPD) method. The second being a least mean squares polynomial (LMSP) curve fitting approach.

B. MULTIPLE TARGET FILTERING ALGORITHM (MULTI)

The routine MULTI calls the matched filter routine for each sensor's amplitude signal. Returned are the target classes found with their relative peak filter response positions. Figure 5.1 illustrates a two target case. The matched filter response and the relative time position for the two classes of targets can be seen. The figure shows that a shell blast target is present. The relative time for this target class is 5800. The relative time returned for the simulated wheeled vehicle target is 3000. These times or positions are relative since each sensor's filter response peaks are offset in time with respect to the peaks of the other sensors. This allows for simultaneous direction finding for each class of target. This algorithm allows selection of either a time domain phase difference or or a least means square polynomial algorithm for finding the direction to the targets. The time domain phase difference algorithm is derived first [Ref. 1]. The least mean squares polynomial direction algorithm then follows and is believed to be an original application to this field [Ref. 10].

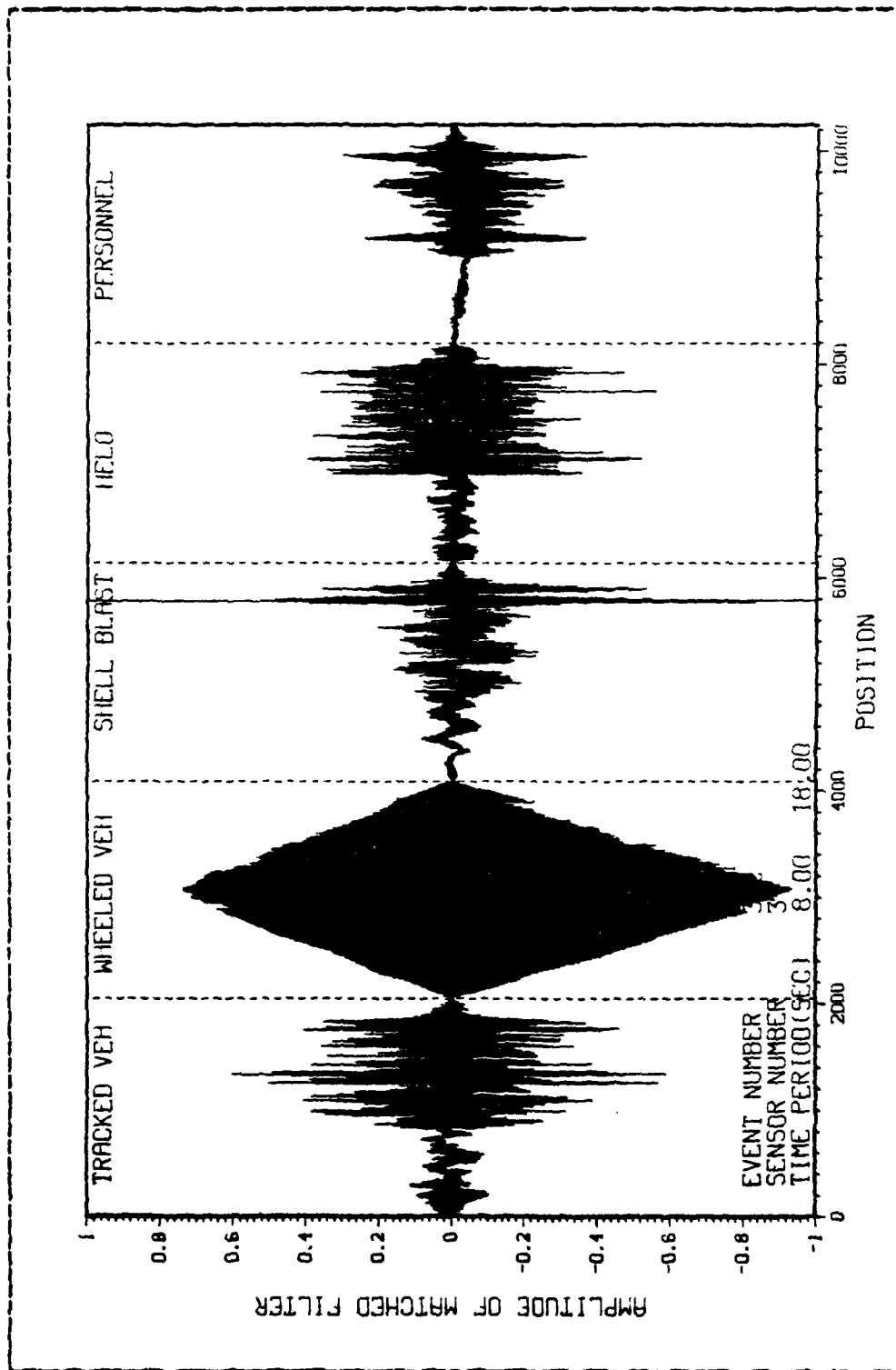


Figure 5.1 Two Target Matched Filter Response

1. Multiple Target Direction Phase Difference Algorithm

T_{ij} - the arrival time of the seismic signal at the I-th sensor for the J-th target class

θ_i - the angle of the I-th sensor to the x-axis

D_{ji} - the distance from the origin to where the wave front of the J-th target passes the I-th sensor

T_{0j} - the time when the J-th target class passes the origin

X_i, Y_i - the position of the I-th sensor

B_j - the arrival angle of the seismic wave for the J-th target class

V - the seismic wave velocity

R - the radius of the sensor ring

I - the sensor number where I has integer values from one to nine

J - the J-th target class

N - the number of sensors in the ring

Figure 5.2 illustrates these parameters and their interdependence. The derivation of the algorithm follows:

$$\theta_i = 2 (I - 1) / N$$

where zero degrees is set parallel to the x-axis

$$X_i = R \cos \theta_i$$

$$Y_i = R \sin \theta_i$$

$$D_{ji} = R \cos(\theta_i - B_j)$$

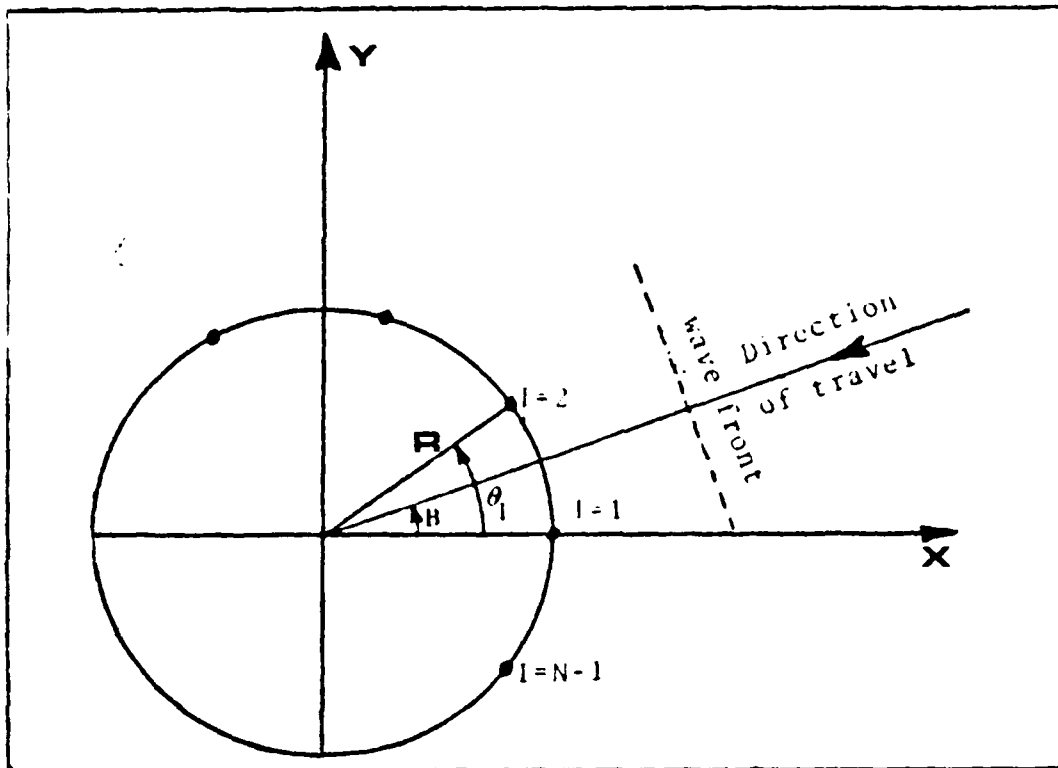


Figure 5.2 Circular Sensor Array Geometry

or equivalently $D_{ji} = (X_i) \cos B_j + (Y_i) \sin B_j$

$$Tc_j = (1/N) \sum_{i=1}^N T_{ji}$$

$$\tau_{ji} = Tc_j - T_{ji}$$

$$V = D_{ji} / \tau_{ji}$$

$$V = ((X_i) \cos B_j + (Y_i) \sin B_j) / (Tc_j - T_{ji}) \quad (5.1)$$

Since the wave velocity can be assumed to be constant when passing all sensors, then for any sensor I and K where $I \neq K$, equation 5.1 leads to

$$\frac{((X_i) \cos B_j + (Y_i) \sin B_j) / (T_{cj} - T_{ji})}{((X_i) \cos B_j + (Y_i) \sin B_j) / (T_{cj} - T_{jk})} =$$

Where $I \neq K$

Now solving for the arrival angle B_{jik} of the wave

$$B_{jik} = \arctan \left(\frac{((T_{cj} - T_{jk}) X_i - (T_{cj} - T_{ji}) X_k)}{((T_{cj} - T_{ji}) Y_k - (T_{cj} - T_{jk}) Y_i)} \right)$$

or equivalently;

$$B_{jik} = \arctan \left(\frac{((T_{cj} - T_{jk}) \cos \theta_i - (T_{cj} - T_{ji}) \cos \theta_k)}{((T_{cj} - T_{ji}) \sin \theta_k - (T_{cj} - T_{jk}) \sin \theta_i)} \right)$$

Where the values of T_{ji} and T_{jk} are returned values from the matched filter routine.

Now;

$$B_j = (1/(N)^2) \sum_{i=1}^N \sum_{k=1}^N B_{jik}$$

Where B_j is the direction in radians to the J-th class target. For the multiple direction routine as implemented, 'j' has values from one to five.

2. Least Mean Square Polynomial Direction Finding

The least mean square direction finding algorithm was developed in response to problems encountered with the phase difference direction finding algorithm. This new method is based on a least mean squares polynomial curve fit of the sensor data. This approach was selected since the least mean squares polynomial provides for best fit or a maximum likelihood curve fit for noisy data.

The least mean squares direction finding algorithm, as with the phase difference algorithm, assumes the seismic wave to be planar. Figure 5.2 illustrates the parameters for this model. Once the assumption of a planar seismic wave is made, the expected relation between the arrival

angle, relative delay times and sensor position in the circular array, can be made. Figure 5.3 illustrates these relations for a nine sensor circular array with a seismic wave arriving at zero degrees. Notice that the relative delay times have been scaled to be from zero to one.

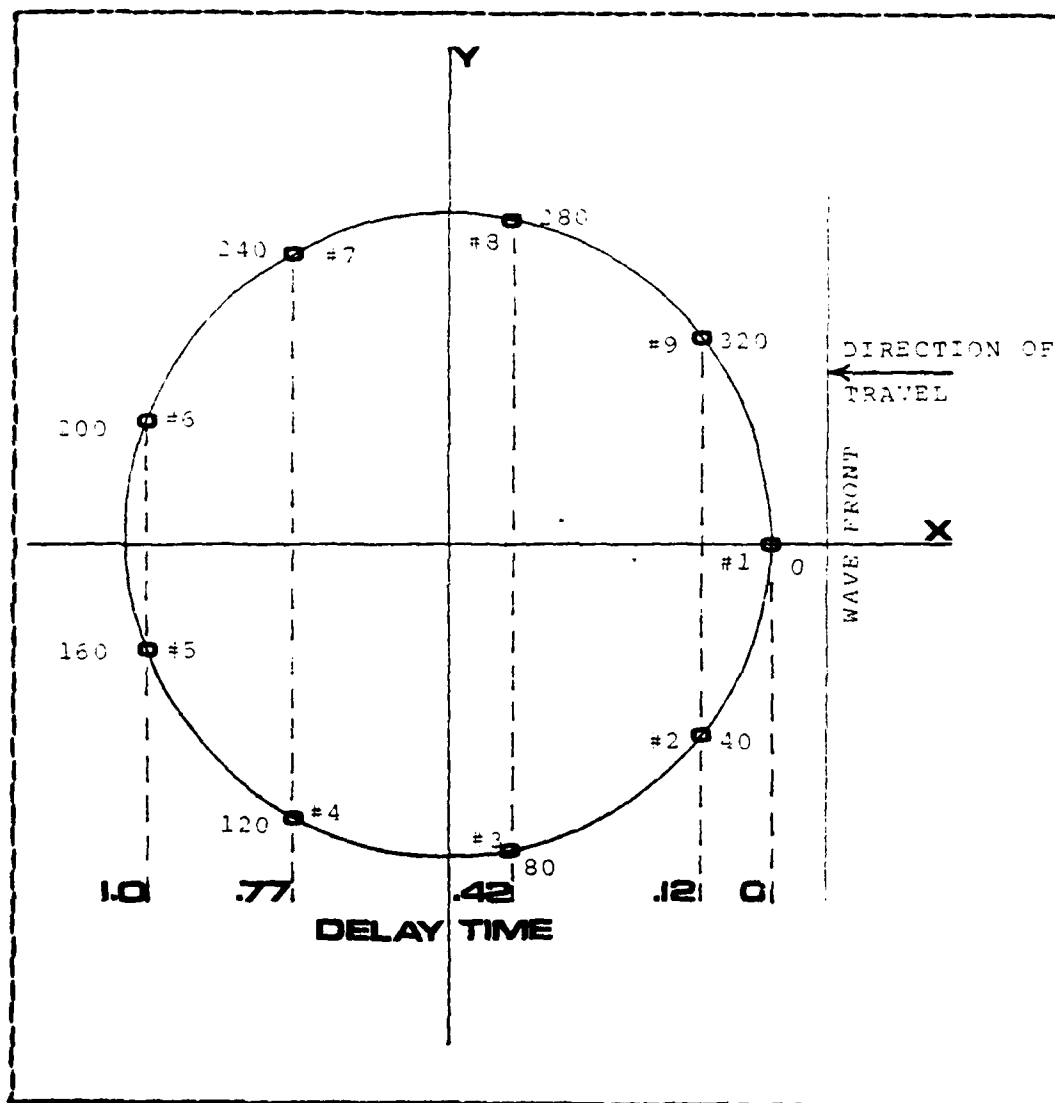


Figure 5.3 Relative Delay Times in a Nine Sensor Ring

Since the sensors in a nine sensor ring are at increments of forty degrees relative to the x-axis, a correlation can be seen between the relative time delays at each sensor and the arrival angle of the seismic wave. A plot can now be made to illustrate the relationship of sensor angles versus relative time delays. Figure 5.3 is a plot for a nine sensor ring with a planar seismic wave arriving at zero degrees. Figure 5.4 shows that for a wave at an arrival angle of 160 degrees, the delay time at sensor number five will be zero. It can be seen that the fitting of these ideal data points with a least mean squares polynomial will produce an equation for a curve whose minimum value is also at the arrival angle of the seismic wave. The minimum degree of the polynomial to fit this ideal data is four. This results from noting that the curve in figure 5.4 has three curve inflections. For experimental data, this minimum curve point corresponds to the predicted arrival angle.

Polynomials of degrees higher than four may be expected to enhance arrival angle errors since the polynomial would distort to fit noisy data. Least means squares polynomials of degree two and three however, may be useful in reducing curve sensitivity to one or two malfunctioning sensors or excessively noisy data.

3. Least Mean Squares Polynomial Algorithm Derivation

Let:

N - number of data point pairs

Y_i - the observed or experimental data position values

X_i - independent degree values with a range of 0 to 360 degrees

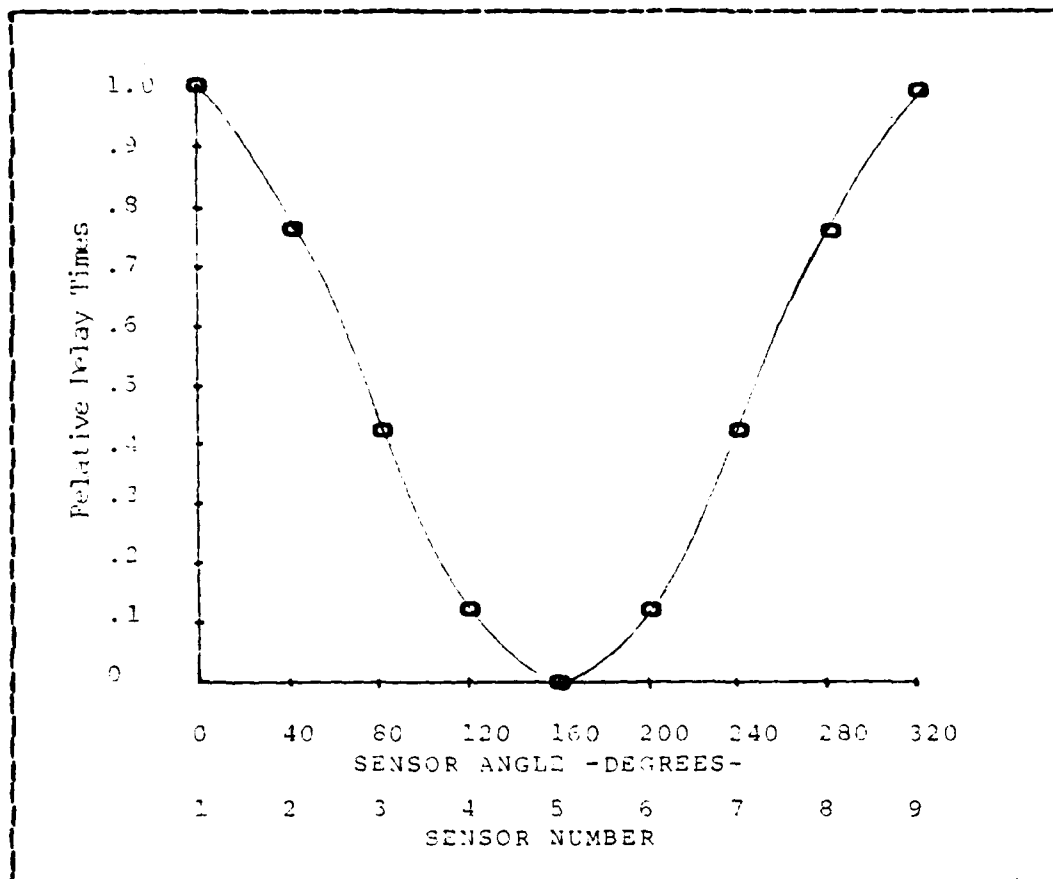


Figure 5.4 Relative Time Delay versus Sensor Angle

P_i - dependent predicted delay time values found by the least mean squares polynomial

A_i, B_i - coefficient values for the system of simultaneous equations

E_i - error between the experimental data values and predicted values of time delays

S - sum of the square errors between each data and predicted points

The derivation of the least mean squares polynomial follows:

$$S = E_1^2 + E_2^2 + E_3^2 + \dots + E_N^2$$

$$P_i = A_0 + A_1 X_i + A_2 X_i^2 + A_3 X_i^3 + \dots + A_n X_i^n \quad (5.2)$$

$$E_i = Y_i - P_i \quad (5.3)$$

$$S = \sum_{i=1}^N E_i^2 \quad (5.4)$$

Combining equations 5.2 and 5.3 yields

$$E_i = Y_i - A_0 - A_1 X_i - A_2 X_i^2 - \dots - A_n X_i^n$$

where n is the degree of the polynomial such that $N > n + 1$ and $1 < i < N$

Equation 5.4 now becomes, after substituting for E , equation 5.5

$$S = \sum_{i=1}^N (Y_i - A_0 - A_1 X_i - A_2 X_i^2 - \dots - A_n X_i^n)^2 \quad (5.5)$$

To find the minimum of the sum of the squares expressed by equation 5.5, the partial derivatives of S with respect to all of the coefficients are taken. At the minimum, these partial derivatives all vanish.

$$\begin{aligned} \partial S / \partial A_0 &= 0 = \sum_{i=1}^N 2(Y_i - A_0 - A_1 X_i - \dots - A_n X_i^n) (-1) \\ \partial S / \partial A_1 &= 0 = \sum_{i=1}^N 2(Y_i - A_0 - A_1 X_i - \dots - A_n X_i^n) (-X_i) \\ &\quad \cdot \\ &\quad \cdot \\ \partial S / \partial A_n &= 0 = \sum_{i=1}^N 2(Y_i - A_0 - A_1 X_i - \dots - A_n X_i^n) (-X_i^n) \end{aligned}$$

Dividing by two and rearranging gives $n + 1$ normal simultaneous equations. Expressed in matrix notation these equations become

$$\begin{bmatrix} N & \sum X_i & \sum X_i^2 & \dots & \sum X_i \\ \sum X_i & \sum X_i^2 & \sum X_i^3 & \dots & \sum X_i \\ \sum X_i^2 & \sum X_i^3 & \sum X_i^4 & \dots & \sum X_i \\ \vdots & \vdots & \vdots & \ddots & \vdots \\ \sum X_i^n & \sum X_i^{n+1} & \sum X_i^{n+2} & \dots & \sum X_i^{2n} \end{bmatrix} \begin{bmatrix} A_0 \\ A_1 \\ A_2 \\ \vdots \\ A_n \end{bmatrix} = \begin{bmatrix} \sum Y_i \\ \sum X_i Y_i \\ \sum X_i^2 Y_i \\ \vdots \\ \sum X_i^n Y_i \end{bmatrix}$$

The coefficient matrix in the above system of equations can be solved for. The minimum value for P_i can then be found. The X corresponding to this minimum value is declared to be the predicted arrival angle of the seismic wave. [Ref. 10]

4. Adaptive Target Direction Finding

An adaptive method is used to improve the directions found for all single target cases. To enhance the resolution accuracy of the peak filter output position, the single target seismic data is copied into the filter data section for its class of target. Subsequently, when the matched filter routine is called for each sensor, filtering is performed with sensor data from that time period.

5. Software features of the Multiple Target Direction Routine

As first addressed in the previous chapter, the multiple direction routine allows for the selection of the number of data points to be used from the 1024 size buffer of sensor data. This was included to investigate the algorithm's performance for various data/filter window sizes and to allow an option for reduced CPU time utilization for interactive program runs. Tabular output from this routine includes the event number, the time period for the data, identification and directions for up to five target classes, and up to four simulated target specifications. Also, notice that the routine adapts to the number of sensors

specified for the ring. This was necessary since the experimental data included sensor rings of six and nine vertical sensors.

C. MULTIPLE TARGETS OF THE SAME TARGET CLASS

A limitation on the system, as presented up to this point, is its inability to engage multiple targets of the same class. Discrete targets may appear as separate entities since the seismic signals generated by the two same class targets are not likely to be incident at the same time. This situation is greatly complicated for continuous time signals, such as tracked or wheeled vehicles. For such continuous time targets, mutual distortion would be the likely result.

An algorithm is needed to identify these multiple peaks without erroneously declaring elements in the same peak as targets. By using only the positive half of the matched filter output and performing data smoothing on the remaining points, a curve with its number of peaks equalling the number of targets present could be generated. Numerical methods for curve fitting or interpolation are available [Ref. 11.]. Polynomial curve fitting, Sterling's method and variations of Newton's method are only a few of the possible approaches applicable for equally spaced data. By differentiating the smoothed data, the peaks and valleys of the filter output can be found. The height of the curve corresponding to the points where the derivative is zero can now be compared to the selected matched filter threshold. This excludes the valley points and leaves the points remaining which correspond to the relative times of the same class targets.

This method is constrained by resolution of close proximity, time limited targets and near phase synchronization

of continuous targets. The variables to be optimized through experimentation may be the degree of the smoothing of the curve data and the exclusion of erroneous valleys associated with the same target's data.

VI. ANALYSIS OF SEISMIC DATA

Analysis of the simulated and experimental seismic data will be conducted as detailed in Table II. Table III lists the matched filter contents for the simulated data. Table IV is the test plan for the experimental data. Table V lists the matched filter signals used for the experimental data analysis.

Table VI summarizes the results of the simulated and experimental data runs for direction finding. The window size, used for all multiple target direction finding results, was 300. Table VII lists the events in which targets were missed or incorrectly identified.

The time domain phase difference directions found, are not presented for the reasons noted earlier. Errors of up to eighty degrees were not uncommon with this method.

Each event run will be accompanied by the following graphic output:

1. Least Mean Square Initial Direction
2. Matched Filter Response
3. Amplitude Response
4. Amplitude Response of any Malfunctioning Sensor
5. Frequency Response
6. Least Mean Squares Polynomial Curve Fitting (LMSP)
Using Matched Filter Outputs
7. Multiple Target Direction Summary Resulting from
Least Mean Squares Curve Fitting

TABLE II
Test Plan for Simulated Data

<u>Event</u>	<u>#Sen</u>	<u>#Tots</u>	<u>Frequency</u>	<u>Amplitude</u>	<u>Direction</u>
001	9	1	10	3000	0
001	9	1	10	3000	40
001	9	1	10	3000	120
001	9	1	10	3000	240

TABLE III
Matched Filter for Simulated Targets

<u>Filter</u>	<u>Frequency</u>	<u>Amplitude</u>	<u>Direction</u>
Tracked Veh	30	2000	0
Wheeled Veh	10	2000	0
Shell blast data from event #383			0
Helicopter	15	2000	0
Personnel	20	2000	0

TABLE IV
Test Plan for Experimental Data

<u>Event</u>	<u>#Sen</u>	<u>#Tgts</u>	<u>Dir</u>	<u>Target</u>	<u>Distance</u>
383	9	1	0	Shot	5KM
382	9	1	0	Shot	5Km
372	6	1	315	Helicopter	5 - 15KM
375	6	1	0	Tank	5 - 0Km
374	6	1	315	Helicopter	15 - 5KM
302	6	1	0	Mortar	1KM
314	6	1	315	LVT	4 - 5KM
354	6	5	0	105mm How	5Km
			225	175mm Gun	4KM
			315	LVT	4 - 5KM
			0	M-60 Tank	4 - 5KM

TABLE V
Matched Filter for Experimental Data

<u>Target</u>	<u>Event Used as Filter</u>
Tracked Vehicle	375
Wheeled Vehicle	none (background noise)
Blast/Recoil	383
Helicopter	372
Personnel	none (background noise)

TABLE VI
Summary of Direction Finding Results

<u>Event</u>	<u>#Sen</u>	<u>#Tqts</u>	<u>Distance</u>	<u>Dir</u>	<u>Initial</u> <u>Dir</u>	<u>Error</u>	<u>LMSP</u> <u>Dir</u>	<u>Error</u>
001	9	1	N/A	0	N/A		0	0
001	9	1	N/A	40	N/A		40	0
001	9	1	N/A	120	N/A		120	0
001	9	1	N/A	240	N/A		240	0
383	9	1	5KM	0	4(3)	1.1	28(4) -5	7.8 1.4
382	9	1	5KM	0	-14(4)	3.9	-6	1.67
372	6	1	5 - 15KM	315	-32	3.6	-59	3.9
375	6	1	5 - 0KM	0	312	13.3	FAILED	
374	6	1	15 - 5KM	315	-59	4.0	FAILED	
302	6	1	1KM	0	4	1.1	6	1.7
354	6	5	5KM	0	-3	.8	3	.8
			4 KM	225	291	18.0	FAILED	
			4 - 5KM	315	FAILED			
			4 - 5KM	0	FAILED			

*Note: The matched filter threshold was set at .9 for all single targets and .6 for all multiple target events. Brackets indicate the use of other than a second degree polynomial.

TABLE VII
Missed or Incorrectly Identified Targets

<u>Event</u>	<u>#Targets</u>	<u>Target</u>	<u>Nature of Errors</u>
375	1	Tank	1. Matched filter was not based on a high S/N sample signal
374	1	Helo	2. Small seismic signal amplitudes
354	1	LVT	3. Malfunctioning sensor(s)
	1	Tank	
	1	175mm Gun	Sever clipping distortion of input signal

Note: The numbered error sources apply to all events listed.

LEAST MEAN SQUARES POLYNOMIAL

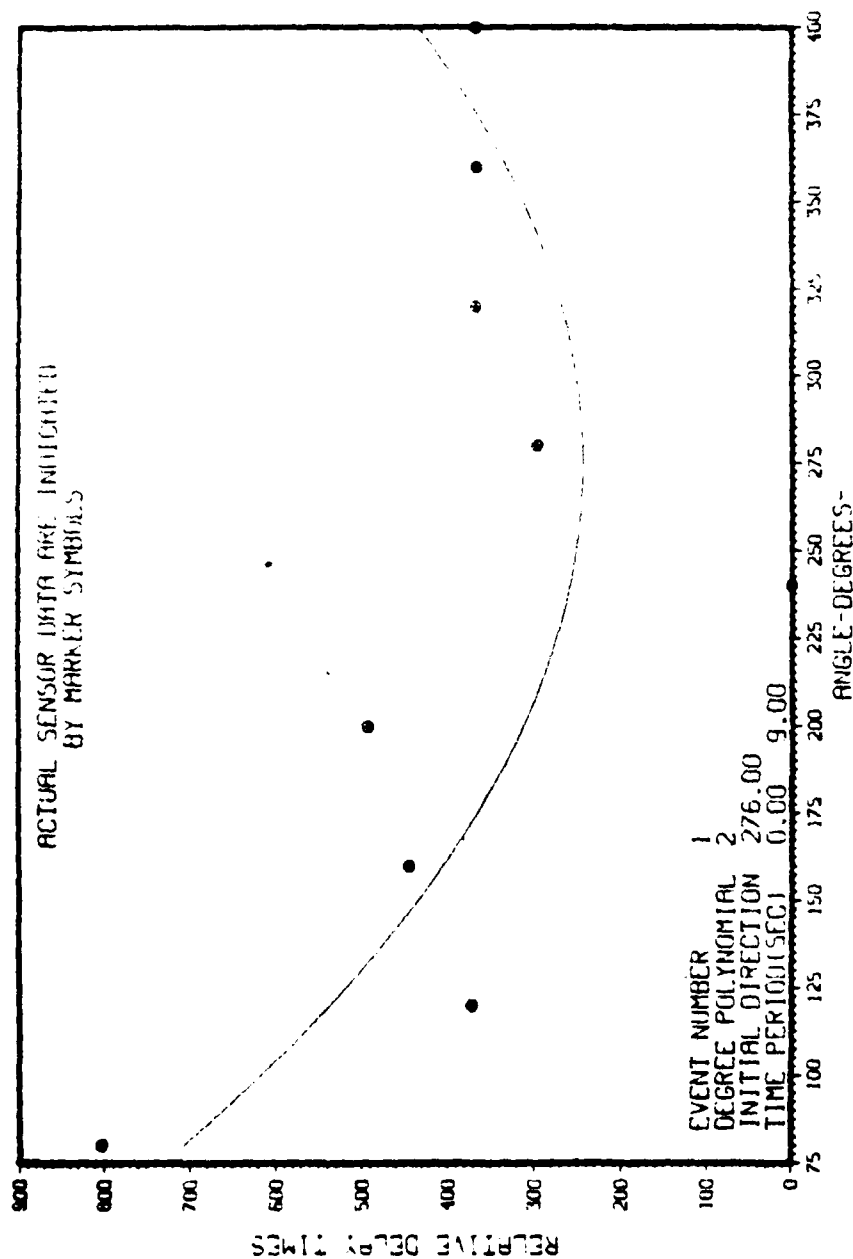


Figure 6.1 Sample Least Mean Squares Initial Direction for Event 001

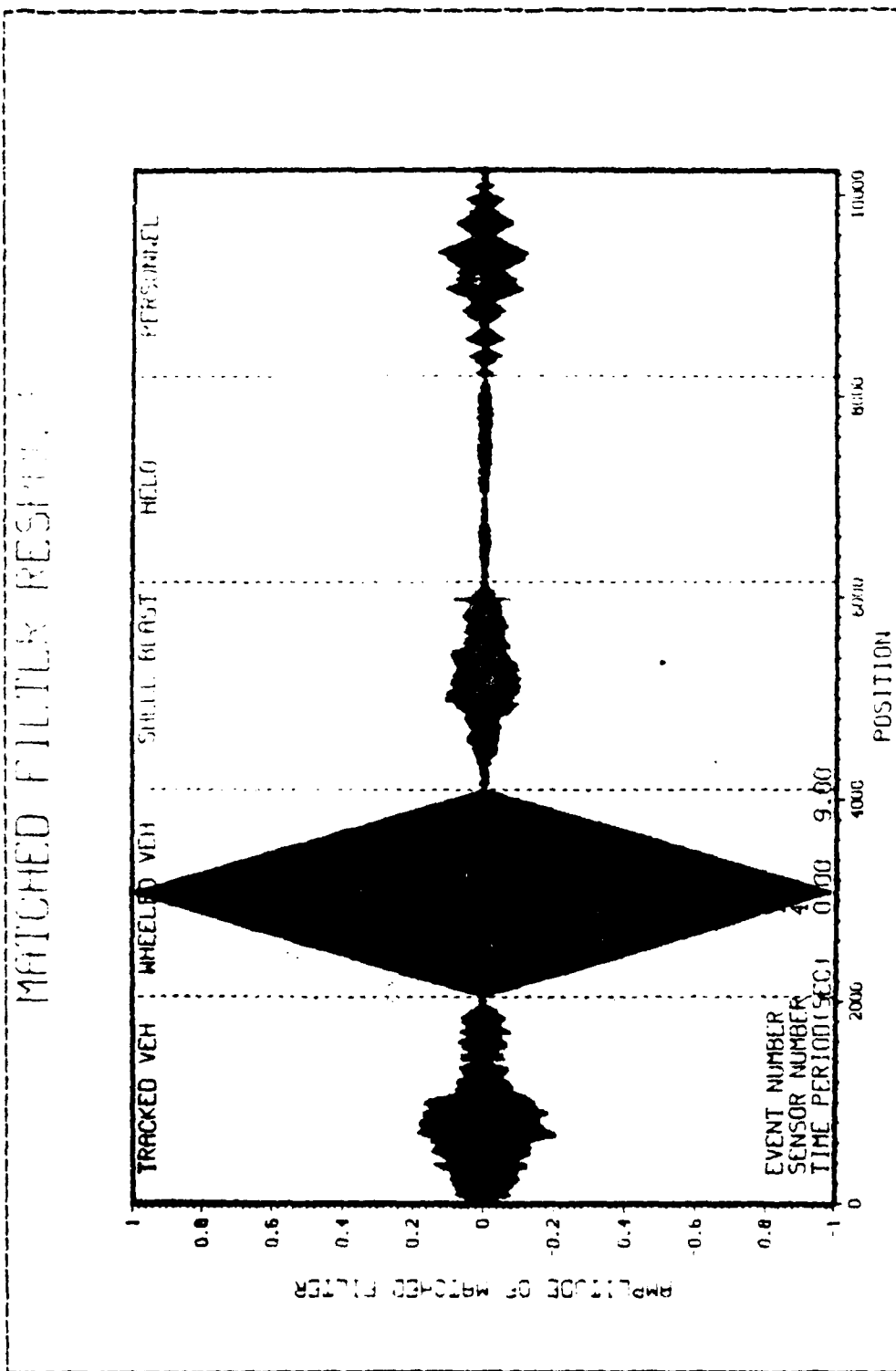


Figure 6.2 Sample Matched Filter Response for Event 001

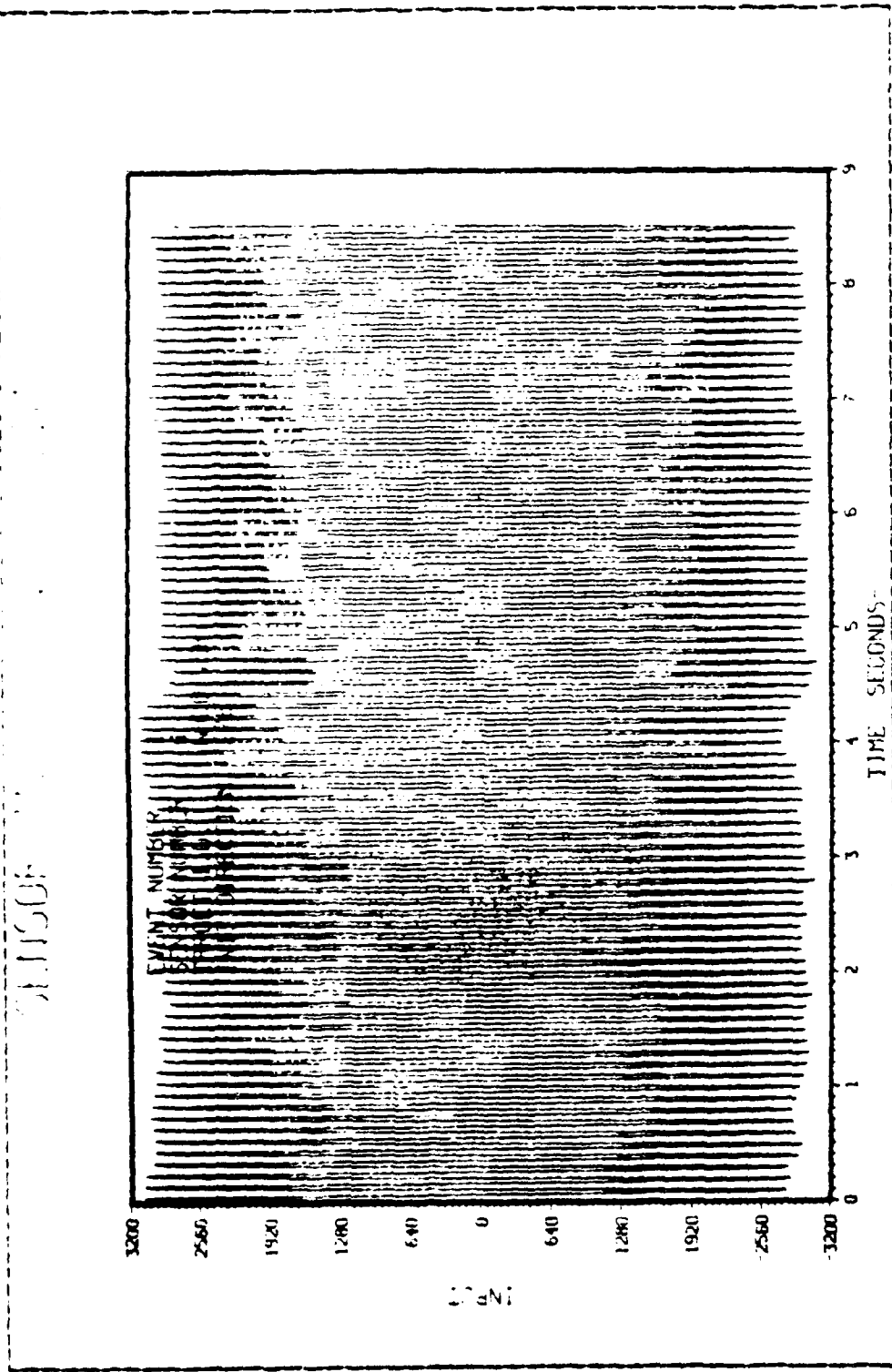


Figure 6.3 Sample Amplitude Response for Event 001

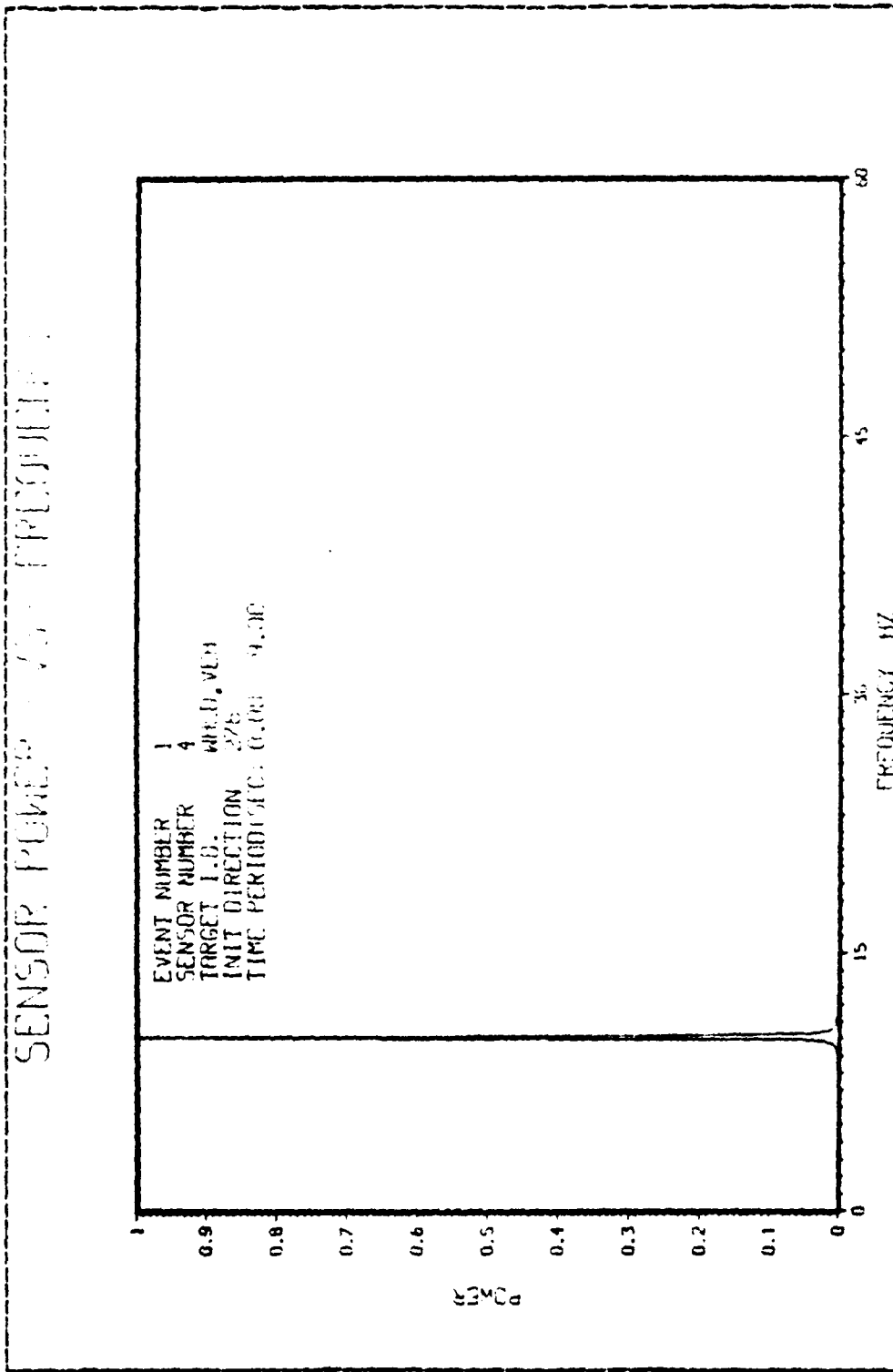


Figure 6.4 Sample Frequency Response for Event 001

LEAST MEAN SQUARES POLYNOMIAL

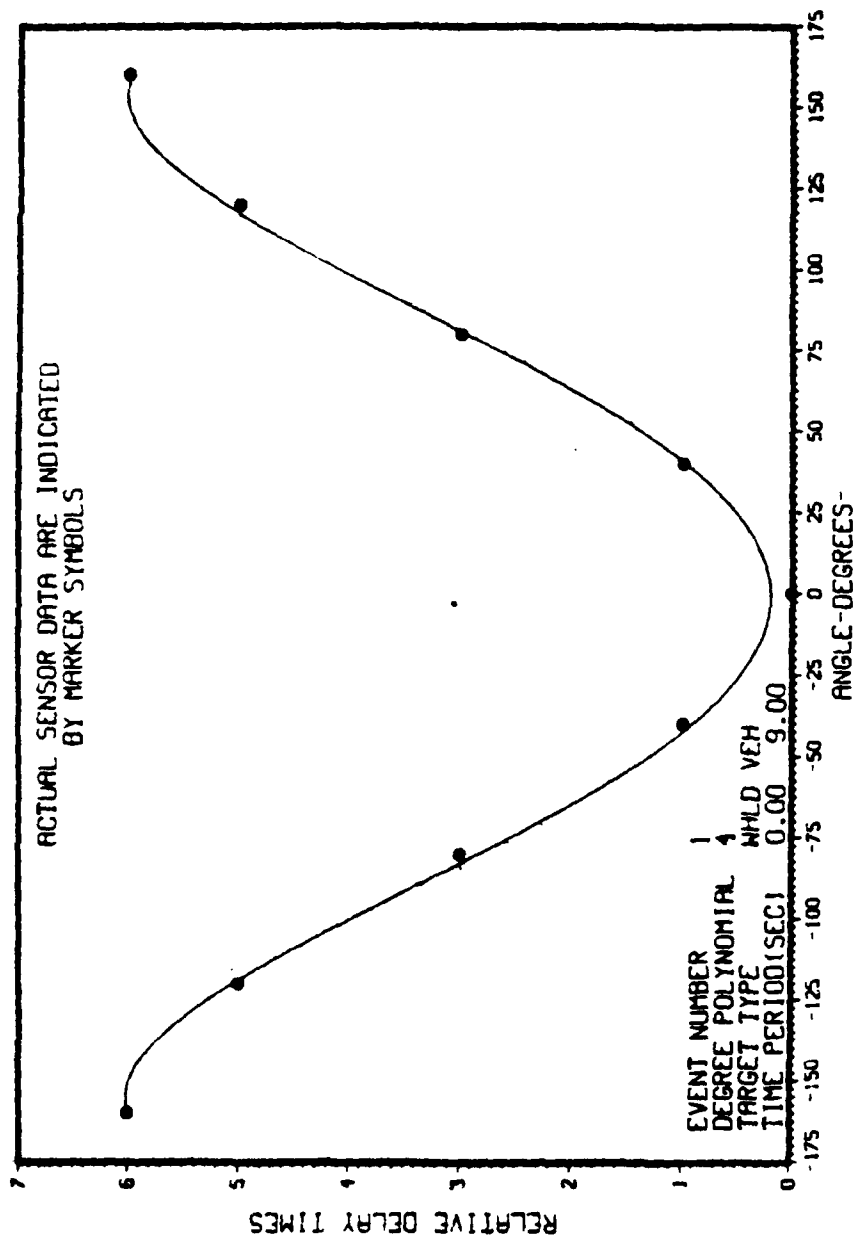


Figure 6.5 LMSP Matched Filter Direction for Event 001

MULTIPLE TARGET - MATCHED FILTER OUTPUT

EVENT NUMBER 1

TIME PERIOD(SEC) 0.00 9.00

WHEELED VEHICLE DIRECTION - 0.00

SIMULATED TRKD VEHICLE	TARGET FREQUENCY	0.00
AMPLITUDE	0.0000	
DIRECTION	0.0000	
SIMULATED WILD VEHICLE	TARGET FREQUENCY	10.00
AMPLITUDE	3000.0000	
DIRECTION	0.0000	
SIMULATED HELICOPTER	TARGET FREQUENCY	0.00
AMPLITUDE	0.0000	
DIRECTION	0.0000	
SIMULATED PERSONNEL	TARGET FREQUENCY	0.00
AMPLITUDE	0.0000	
DIRECTION	0.0000	

Figure 6.6 LMSP Multiple Target Direction Summary for Event 001

LEAST MEAN SQUARES POLYNOMIAL

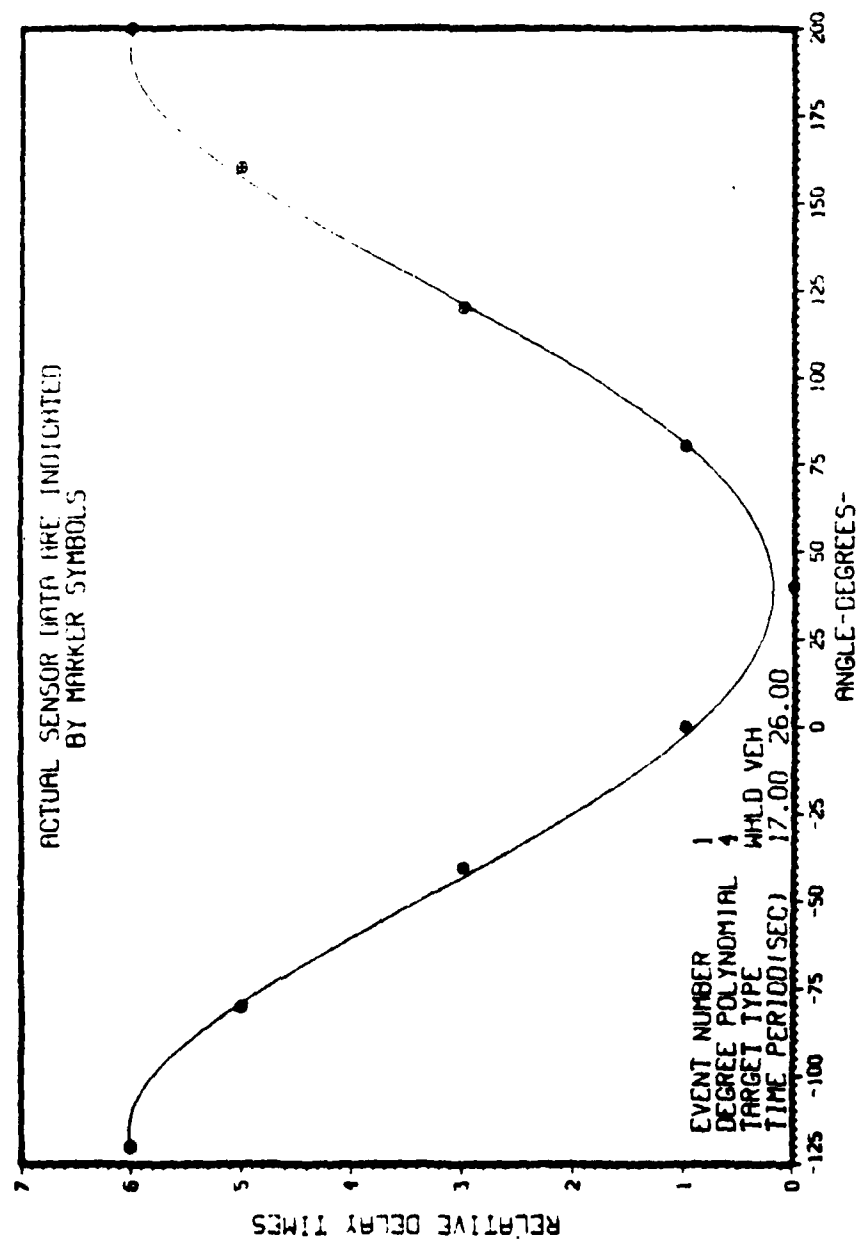


Figure 6.7 LMSP Matched Filter Direction for Event 001

MULTIPLE TARGET - MATCHED - LMSPP OUTPUT

EVENT NUMBER	1	
TIME PERIOD(SEC)	17.00 26.00	
WHEELED VEHICLE	DIRECTION - 40.000	
SIMULATED TRKD VEHICLE	TARGET FREQUENCY	0.00
AMPLITUDE	0.0000	
DIRECTION	0.0000	
SIMULATED WILD VEHICLE	TARGET FREQUENCY	10.00
AMPLITUDE	3000.0000	
DIRECTION	40.0000	
SIMULATED HELICOPTER	TARGET FREQUENCY	0.00
AMPLITUDE	0.0000	
DIRECTION	40.0000	
SIMULATED PERSONNEL	TARGET FREQUENCY	0.00
AMPLITUDE	0.0000	
DIRECTION	0.0000	

Figure 6.8 LMSPP Multiple Target Direction Summary for Event 001

LEAST MEAN SQUARES POLYNOMIAL

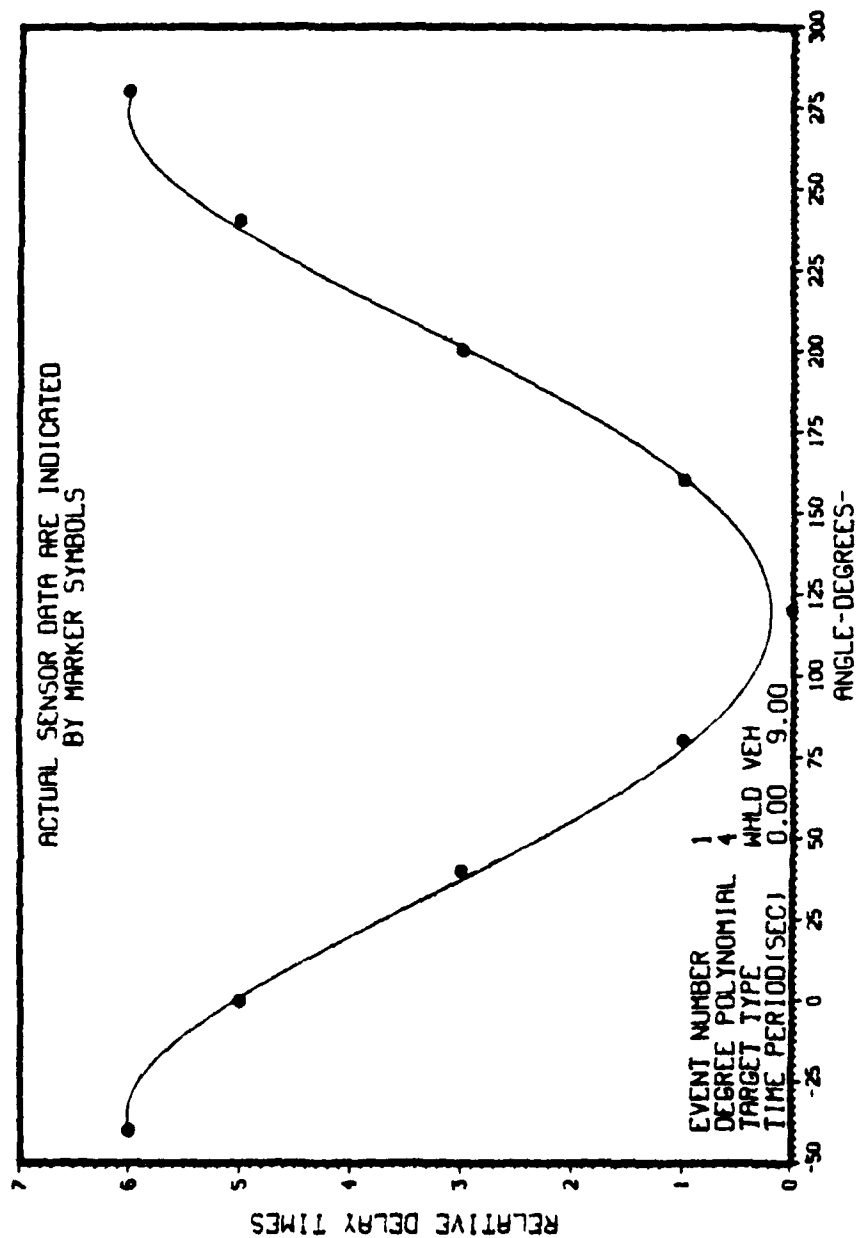


Figure 6.9 LMSP Matched Filter Direction for Event 001

MULTIPLE TARGET - MATCHED FILTER OUTPUT

EVENT NUMBER	1
TIME PERIOD(SEC)	0.00 9.00
WHEELED VEHICLE	DIRECTION - 120.00
SIMULATED TRKD VEHICLE	TARGET FREQUENCY 10.00
AMPLITUDE	3000.0000
DIRECTION	120.0000
SIMULATED WMLD VEHICLE	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED HELICOPTER	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED PERSONNEL	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000

Figure 6.10 LESP Multiple Target Direction Summary for Event 001

LEAST MEAN SQUARES POLYNOMIAL

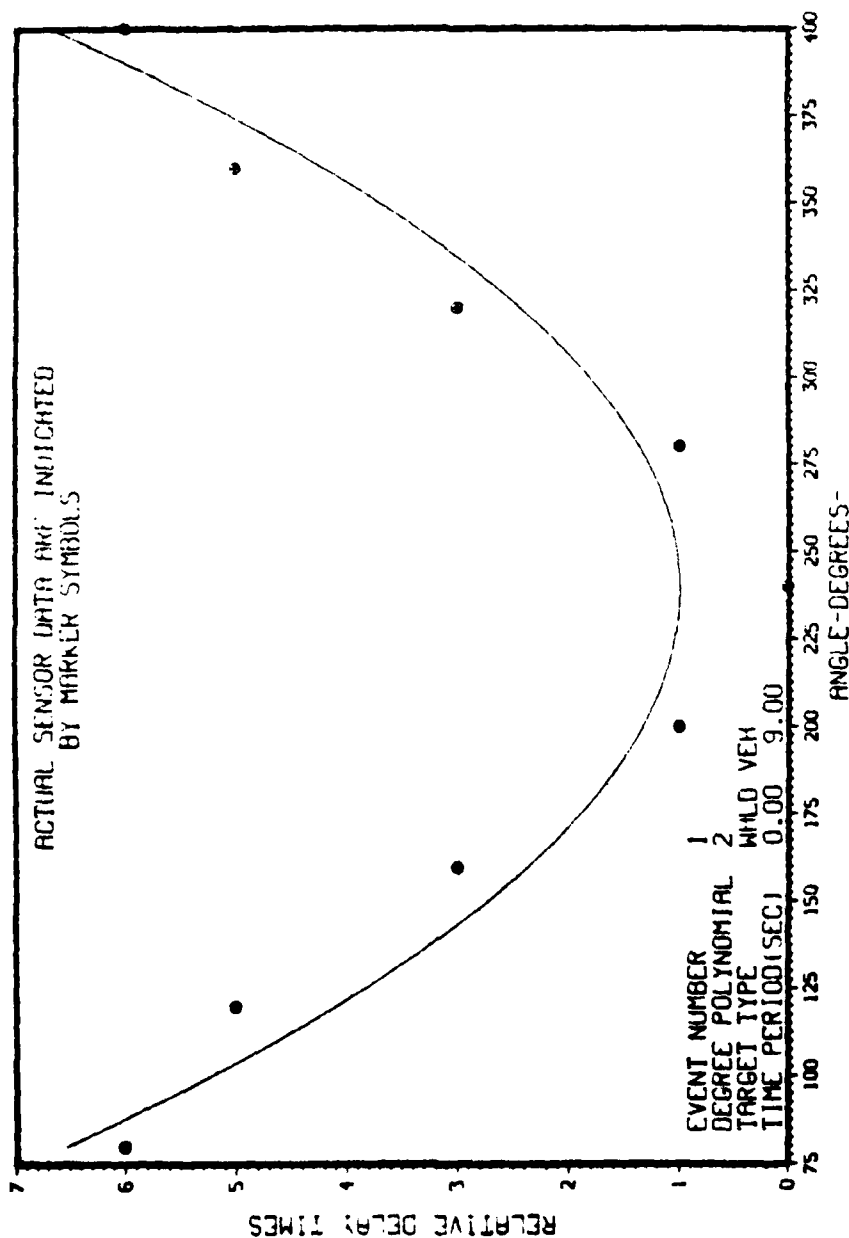


Figure 6.11 LMSP Matched Filter Direction for Event 001

MULTIPLE TARGET - MATCHED FILTER OUTPUT

```

EVENT NUMBER      1
TIME PERIOD(SEC) 0.00  9.00

WHEELED VEHICLE   DIRECTION - 240.00

SIMULATED TRKD VEHICLE TARGET FREQUENCY  0.00
AMPLITUDE 0.0000
DIRECTION 0.0000
SIMULATED WMD VEHICLE TARGET FREQUENCY  10.00
AMPLITUDE 3000.0000
DIRECTION 240.0000
SIMULATED HELICOPTER TARGET FREQUENCY  0.00
AMPLITUDE 6.0000
DIRECTION 6.0000
SIMULATED PERSONNEL TARGET FREQUENCY  0.00
AMPLITUDE 0.0000
DIRECTION 0.0000
    
```

Figure 6.12 LMSF Multiple Target Direction Summary for Event 001

LEAST MEAN SQUARES POLYNOMIAL

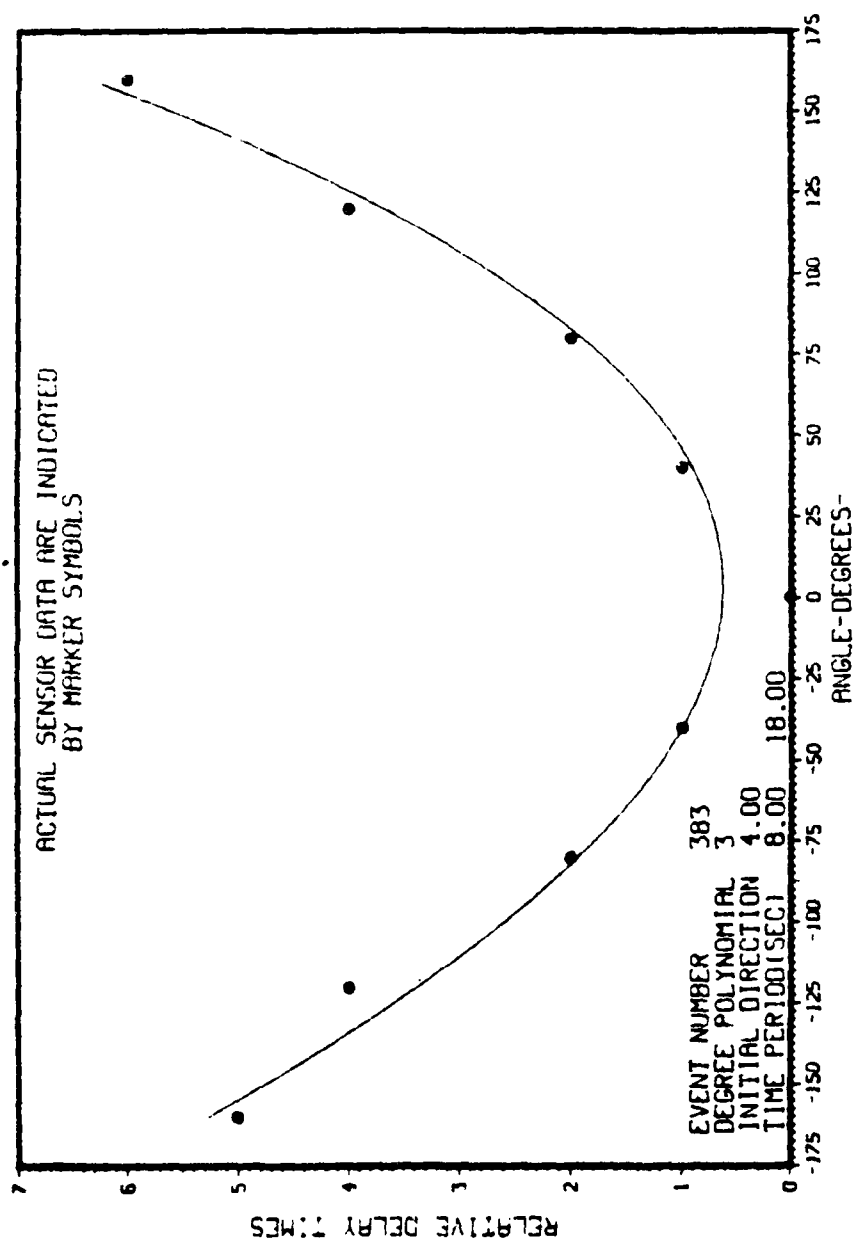


Figure 6.13 LMSP Initial Direction for Event 383

MATCHED FILTER RESPONSE

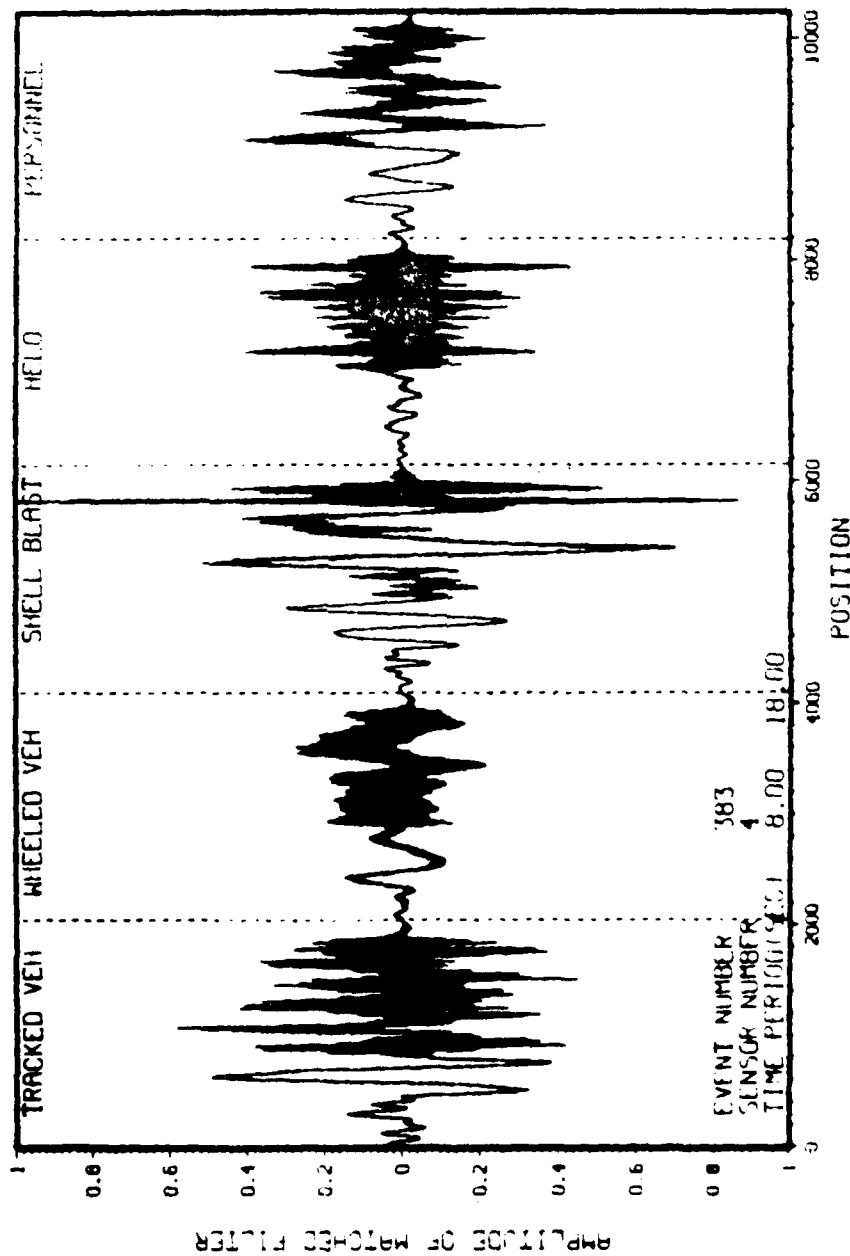


Figure 6.14 Matched Filter Response for Event 383

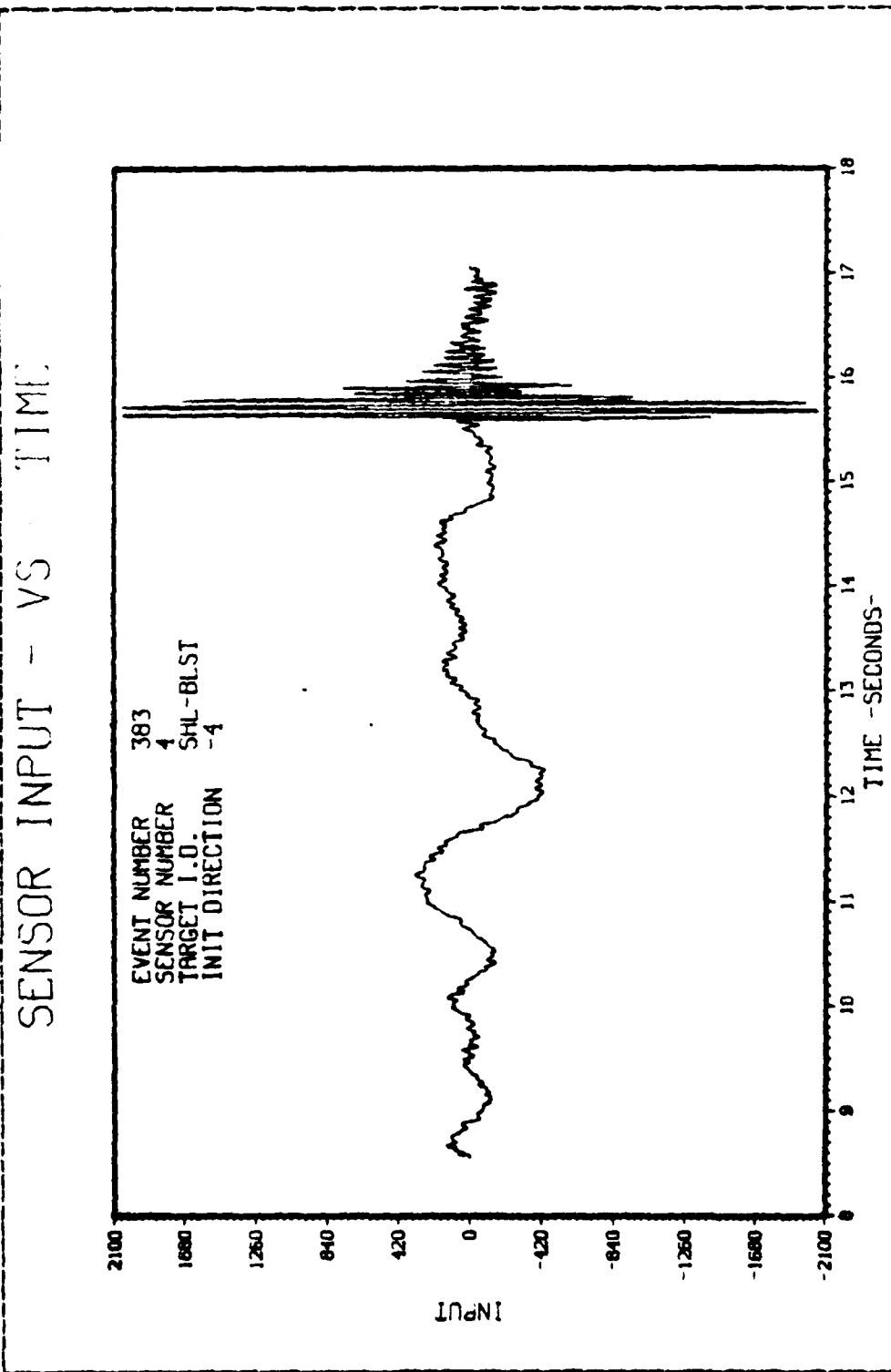


Figure 6.15 Amplitude Response for Event 383

SENSOR POWER -VS- FREQUENCY

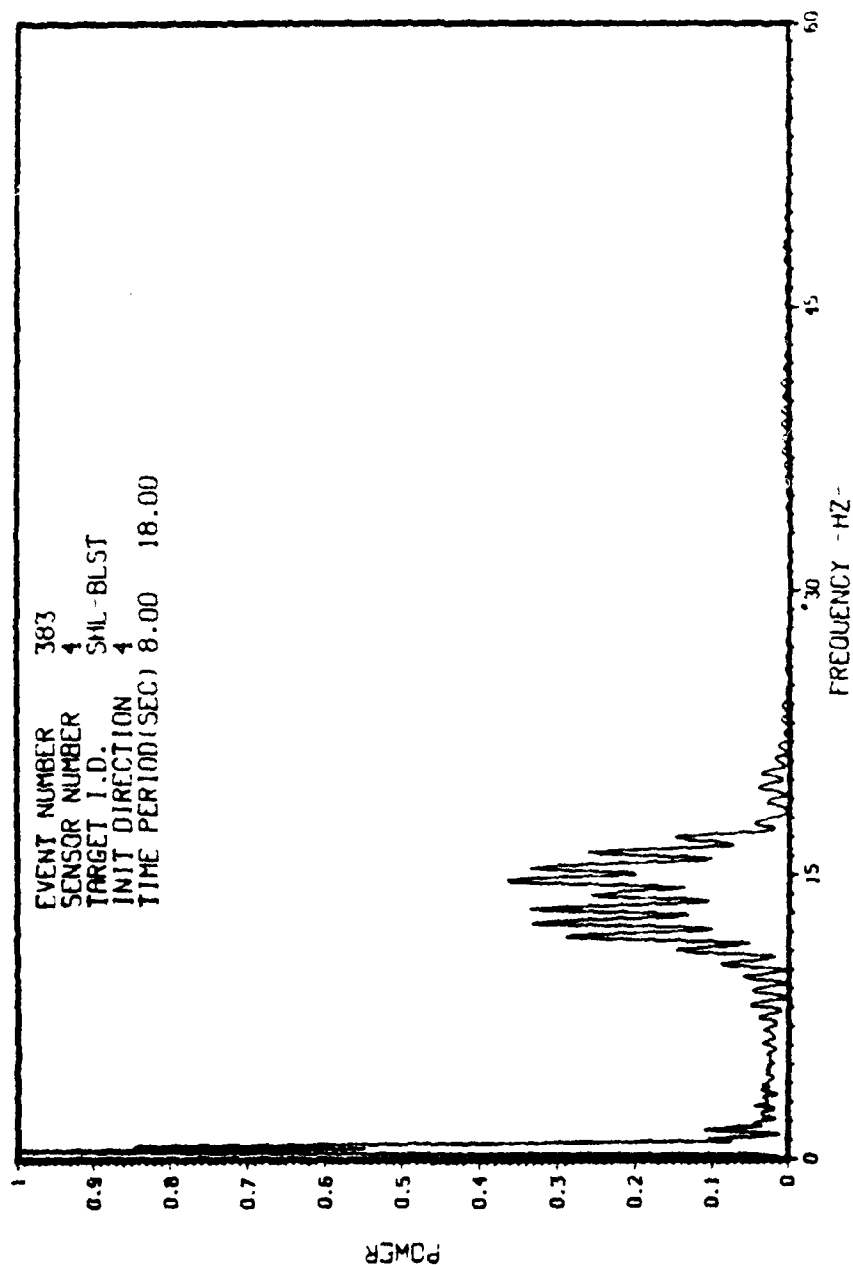


Figure 6.16 Frequency Response for Event 383

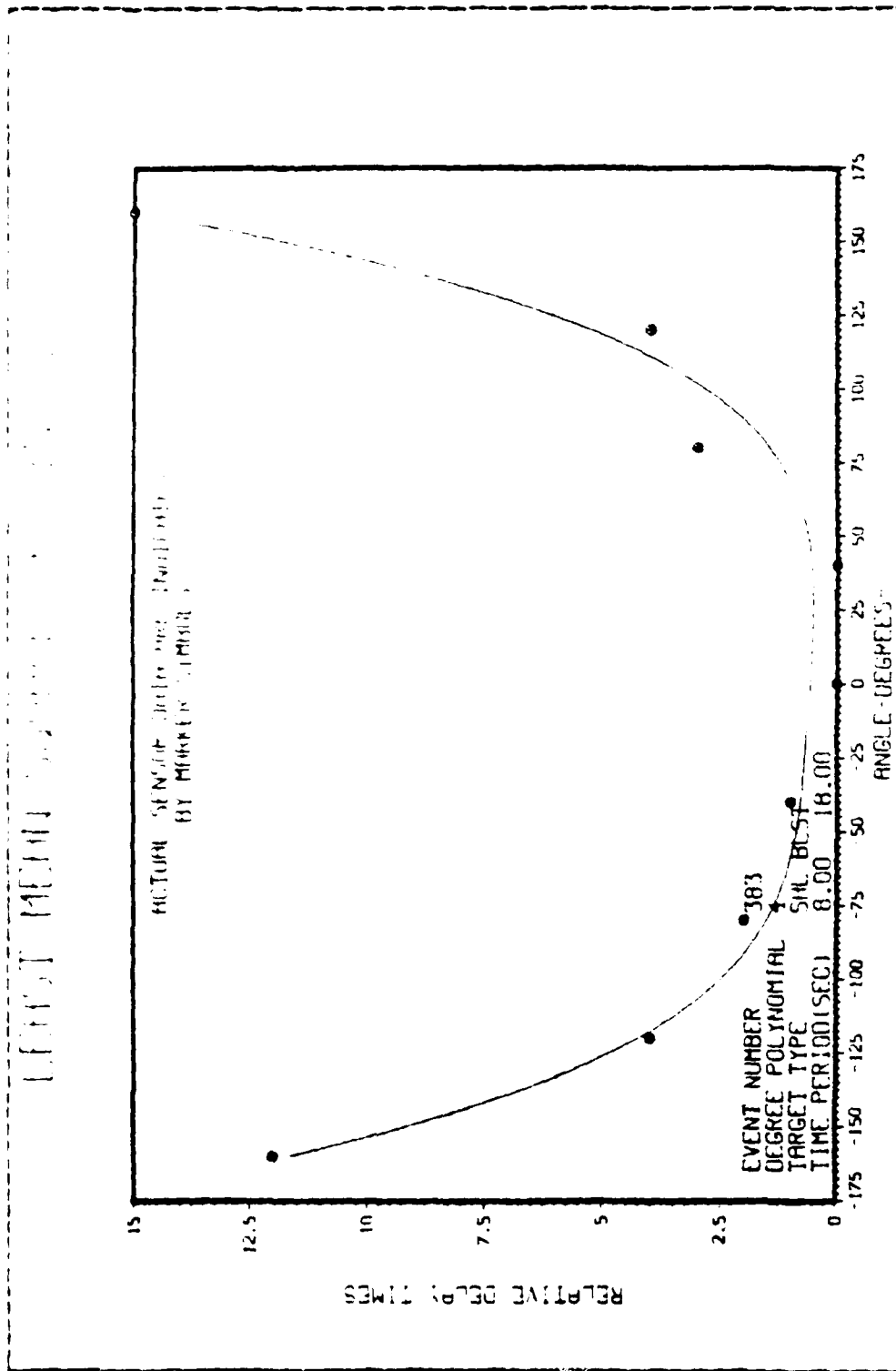


Figure 6.17 Fourth Degree LMSP Matched Filter Direction for Event 383

LEAST MEAN SQUARES POLYNOMIAL

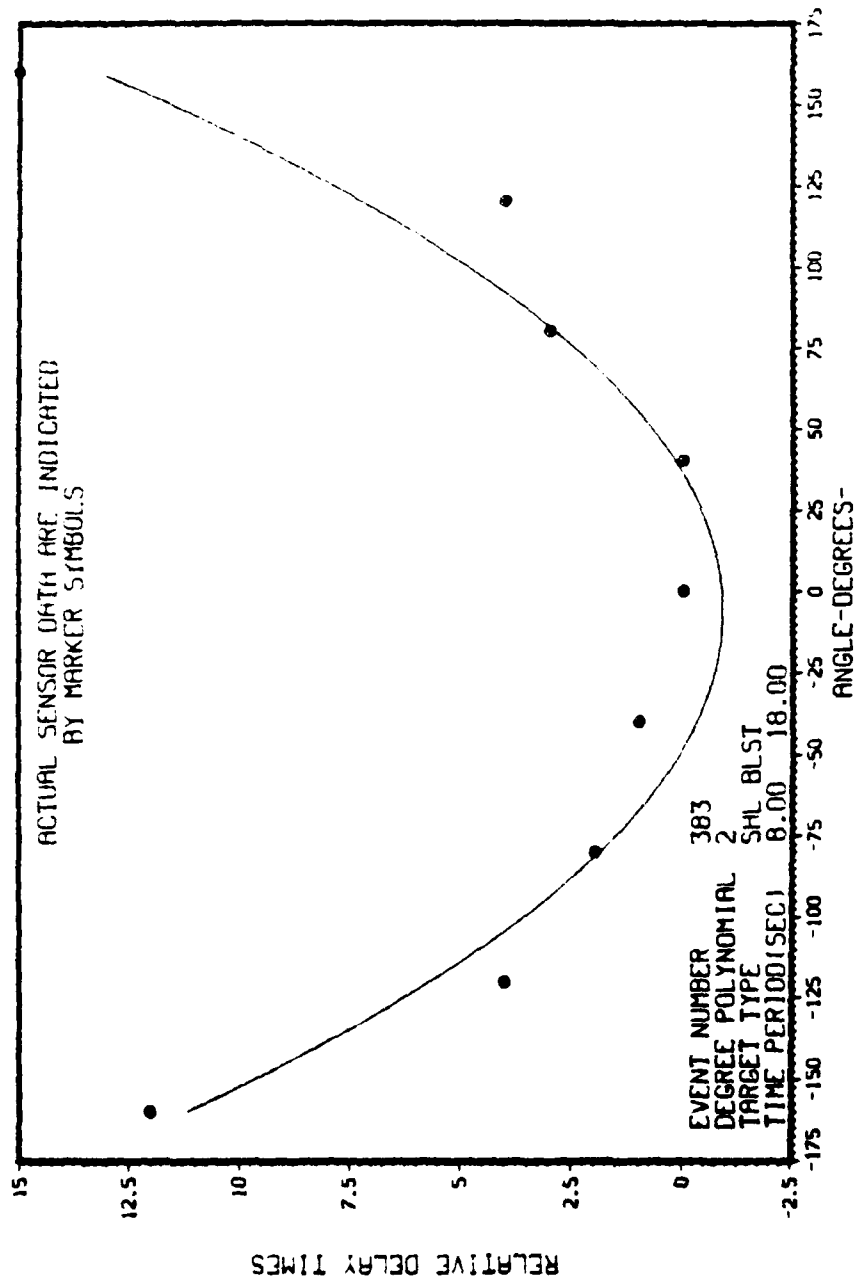


Figure 6.18 Second Degree LMSP Matched Filter Direction for Event 383

MULTIPLE TARGET - MATCHED FILTER OUTPUT

EVENT NUMBER 383

TIME PERIOD(SEC) 8.00 18.00

SHELL BLAST DIRECTION - 136.00

SIMULATED TRKO VEHICLE	TARGET FREQUENCY	0.00
AMPLITUDE	0.0000	
DIRECTION	0.0000	
SIMULATED WILD VEHICLE	TARGET FREQUENCY	0.00
AMPLITUDE	0.0000	
DIRECTION	0.0000	
SIMULATED HELICOPTER	TARGET FREQUENCY	0.00
AMPLITUDE	0.0000	
DIRECTION	0.0000	
SIMULATED PERSONNEL	TARGET FREQUENCY	0.00
AMPLITUDE	0.0000	
DIRECTION	0.0000	

Figure 6.19 LMSP Multiple Target Direction Summary for Event 383

LEAST MEAN SQUARES POLYNOMIAL

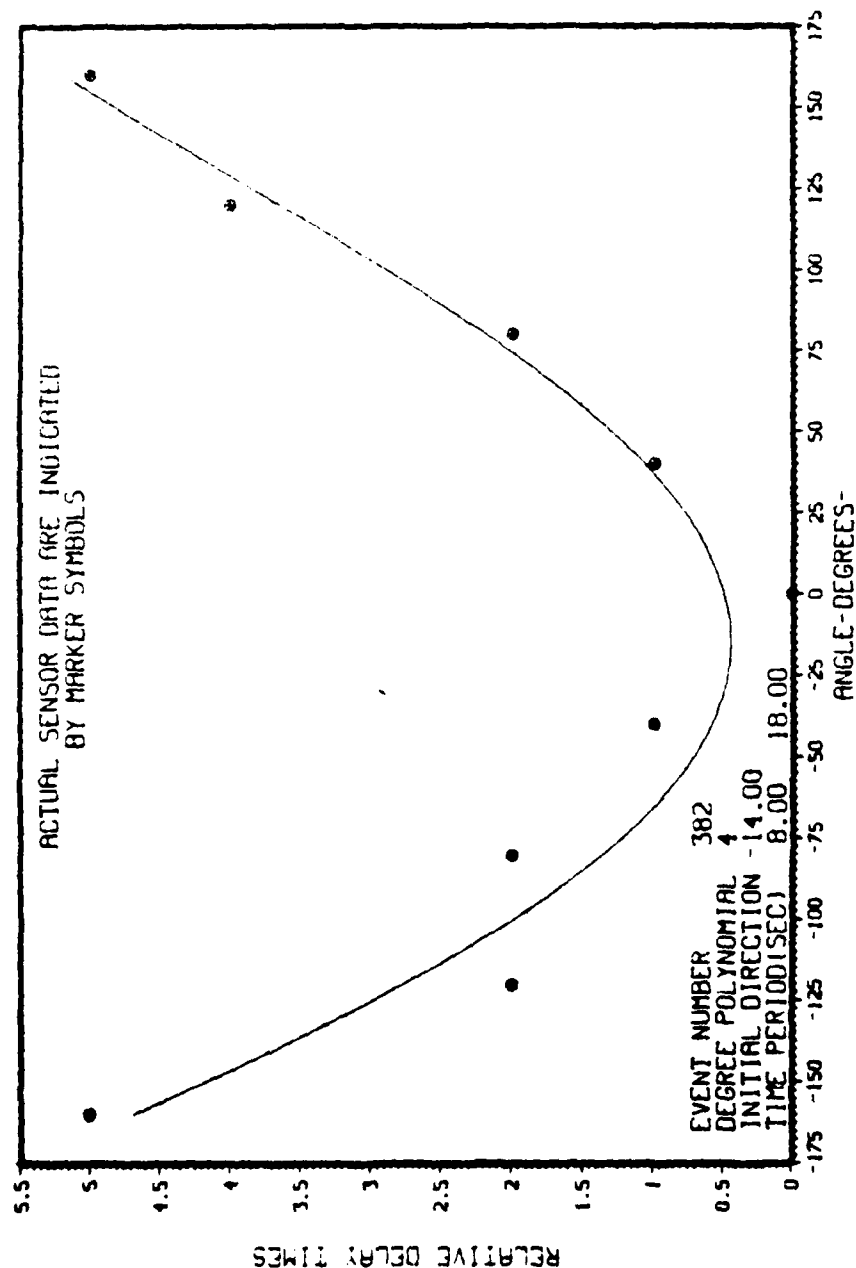


Figure 6.20 LMSP Initial Direction for Event 382

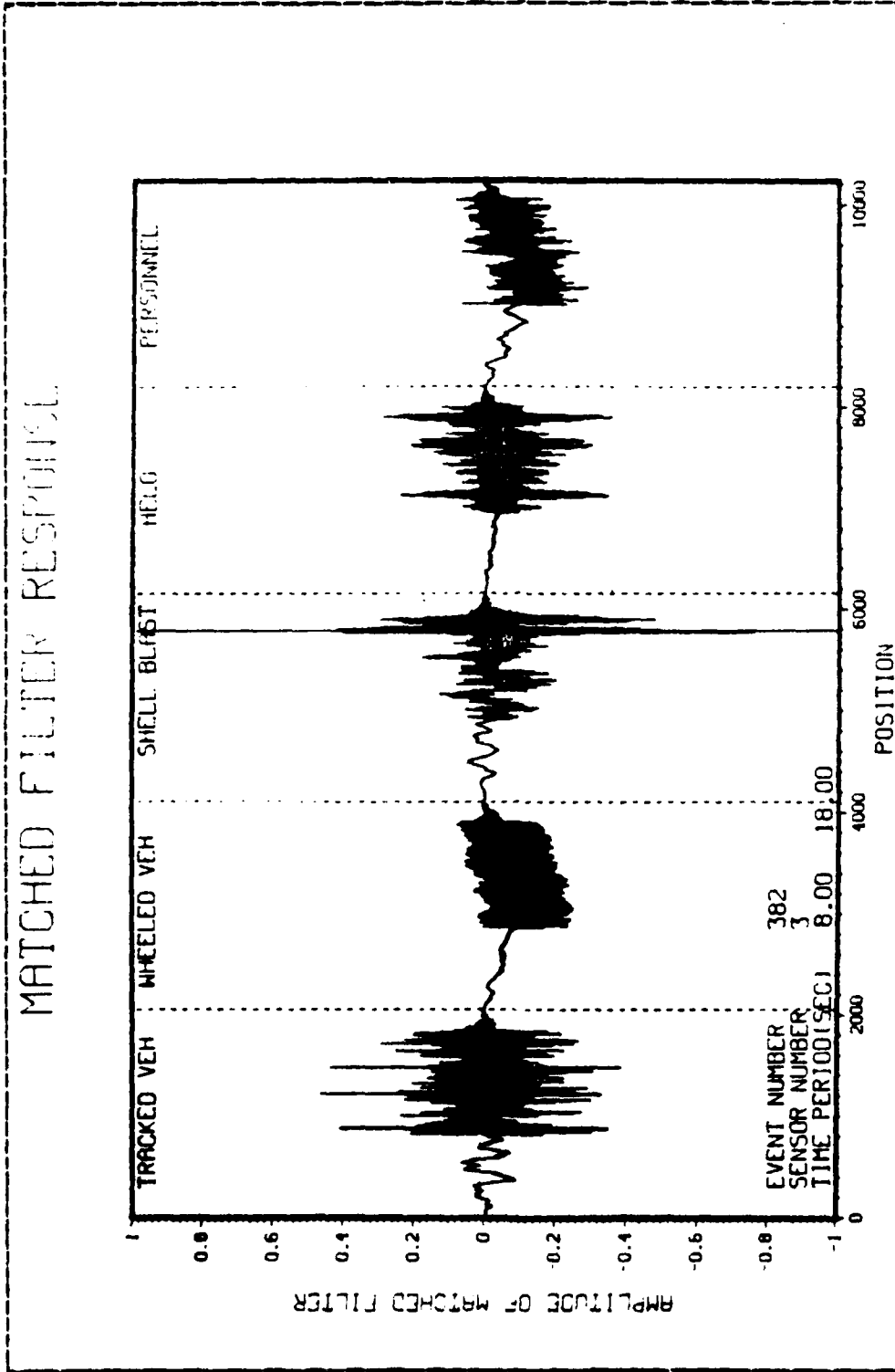


Figure 6.21 Matched Filter Response for Event 382

SENSOR INPUT - WAVEFORM

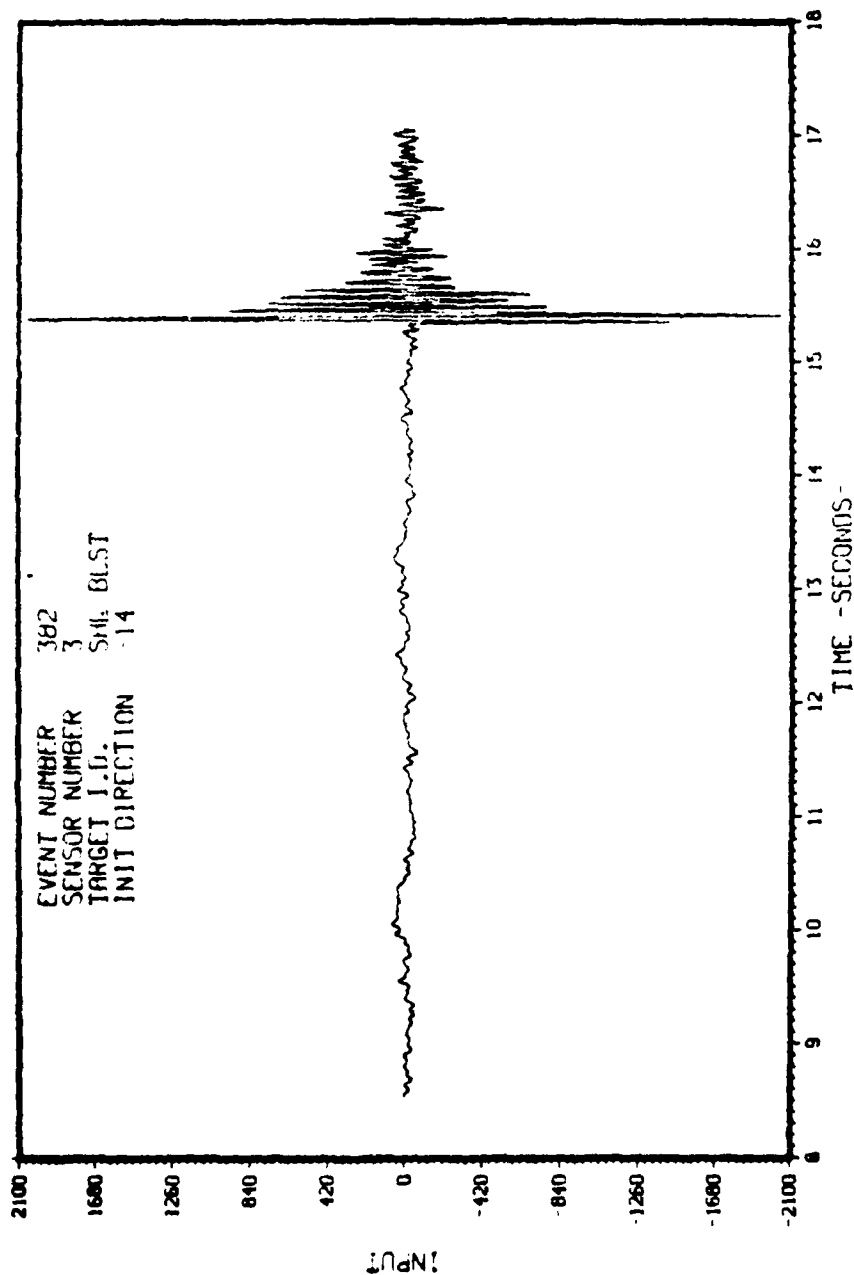


Figure 6.22 Amplitude Response for Event 382

SENSOR POWER - VS- FREQUENCY

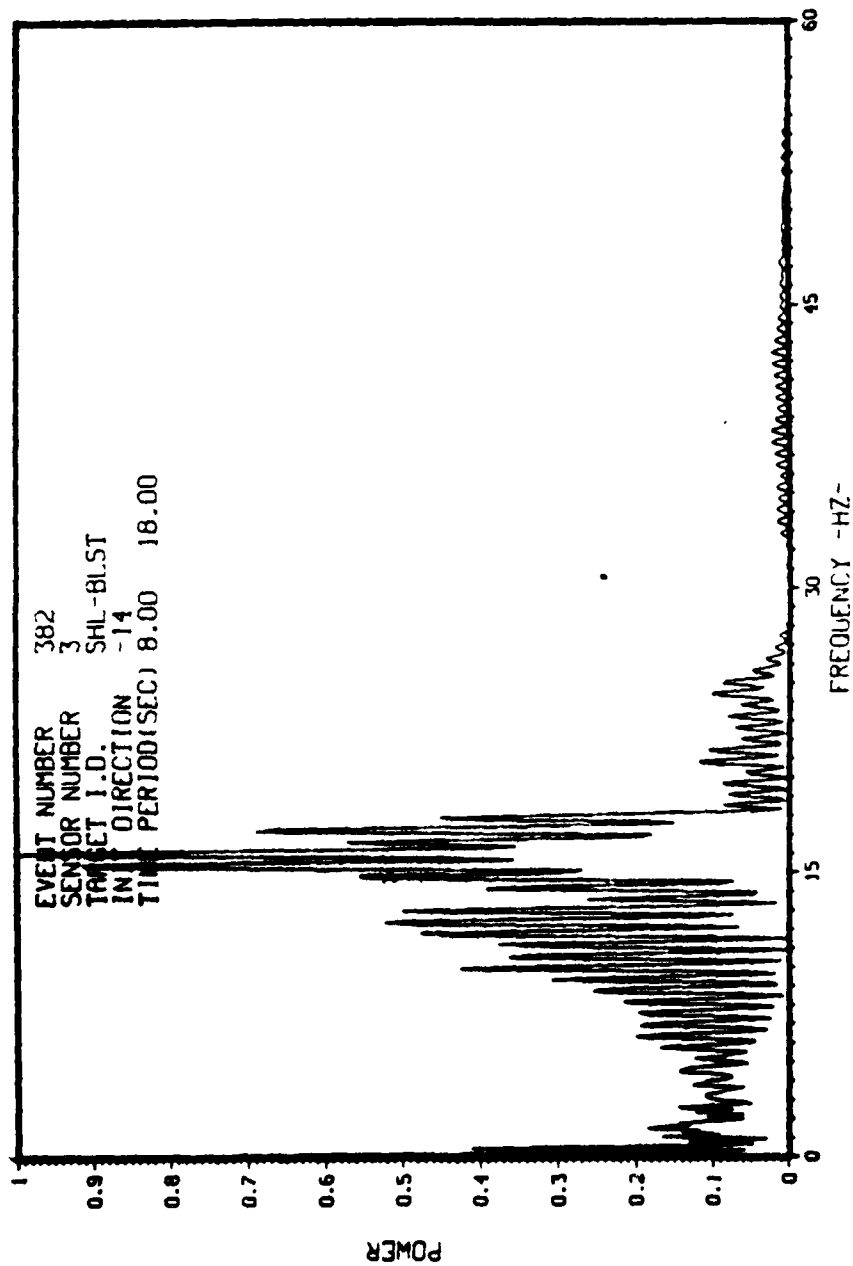


Figure 6.23 Frequency Response for Event 382

LEAST MEAN SQUARES CURVE FIT

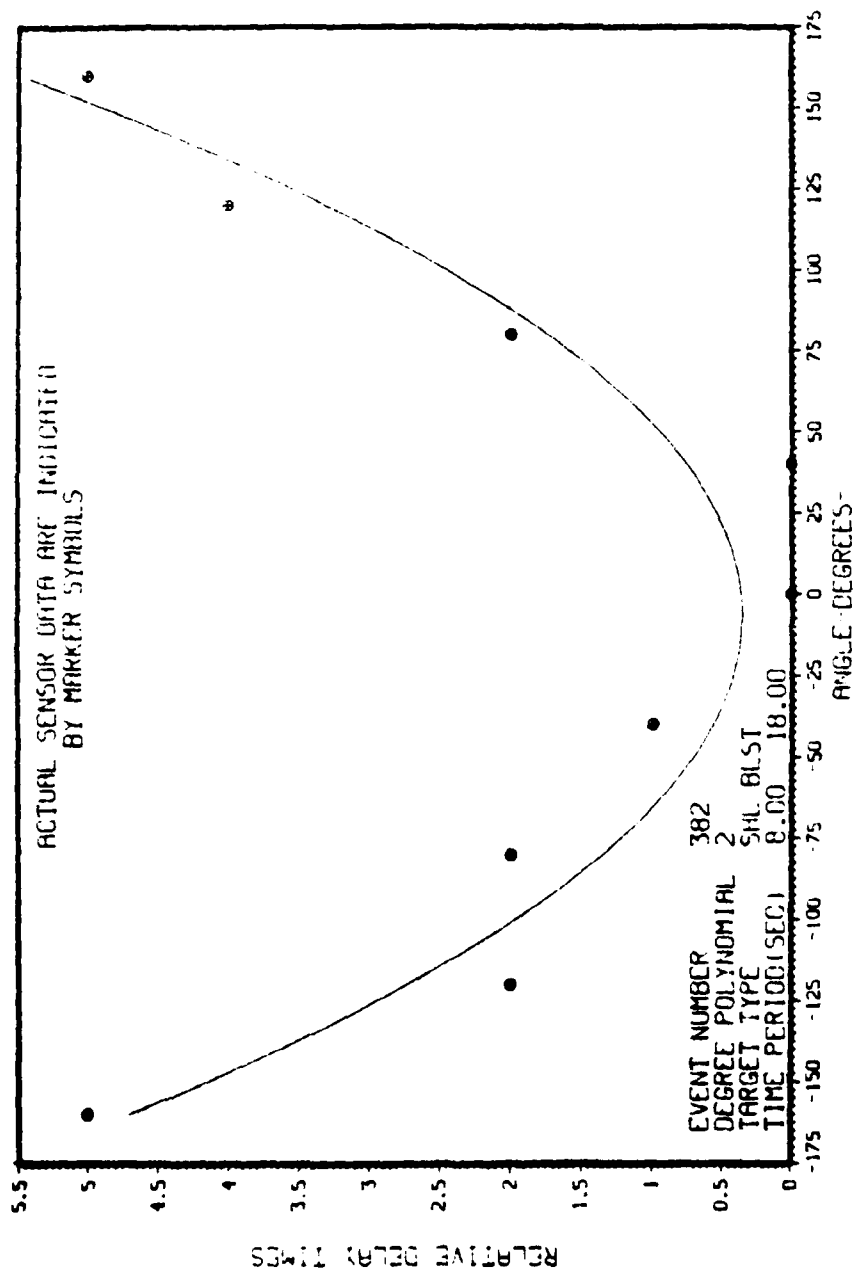


Figure 6.24 LNMP Matched Filter Direction for Event 382

MULTIPLE TARGET - MATCHED FILTER OUTPUT

EVENT NUMBER 382

TIME PERIOD(SEC) 8.00 18.00

SHELL BURST DIRECTION - 6.00

SIMULATED TRK VEHICLE TARGET FREQUENCY	0.00
AMPLITUDE 0.0000	
DIRECTION 0.0000	
SIMULATED WLD VEHICLE TARGET FREQUENCY	0.00
AMPLITUDE 0.0000	
DIRECTION 0.0000	
SIMULATED HELICOPTER TARGET FREQUENCY	0.00
AMPLITUDE 0.0000	
DIRECTION 0.0000	
SIMULATED PERSONNEL TARGET FREQUENCY	0.00
AMPLITUDE 0.0000	
DIRECTION 0.0000	

Figure 6.25 LMSP Multiple Target Direction Summary for Event 382

LEAST MEAN SQUARES POLYNOMIAL

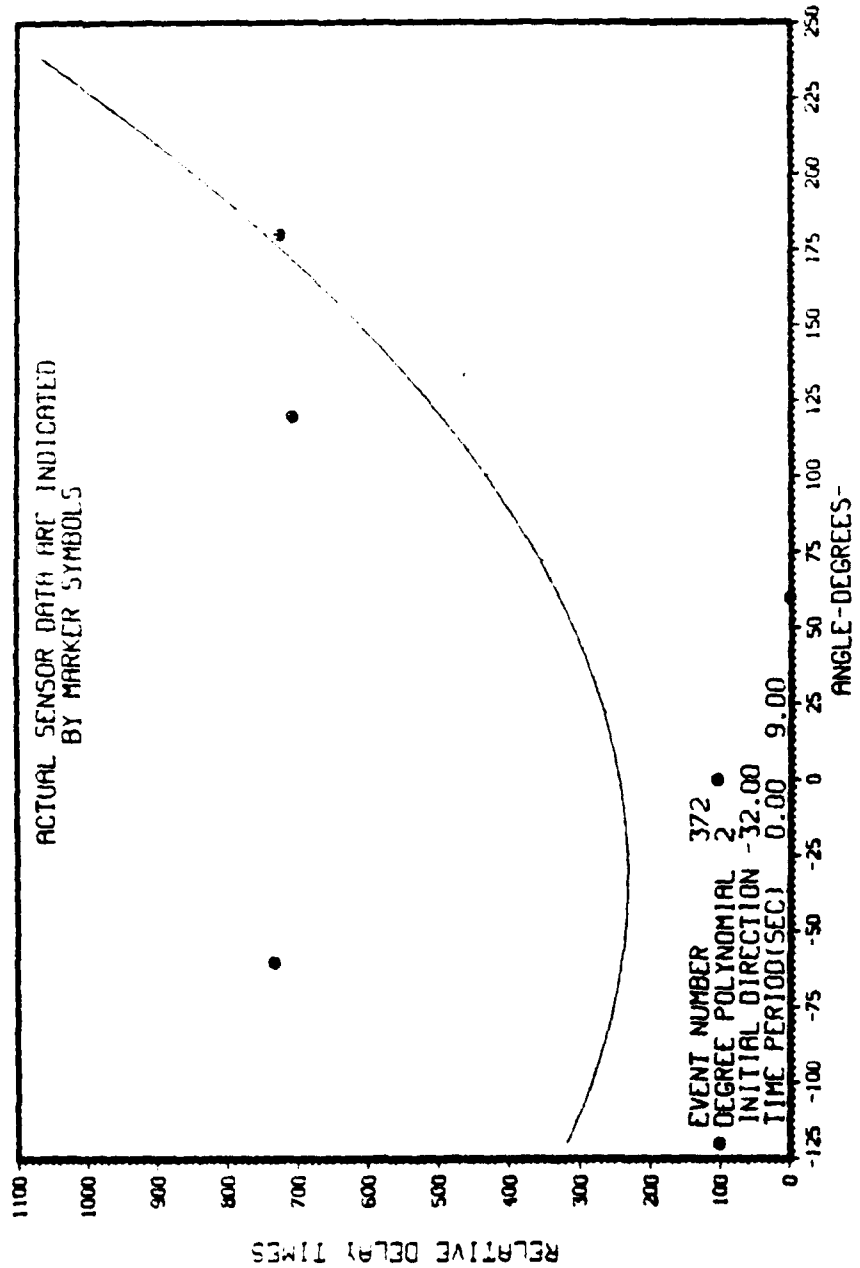


Figure 6.26 LMSP Initial Direction for Event 372

MATCHED FILTER RESPONSE

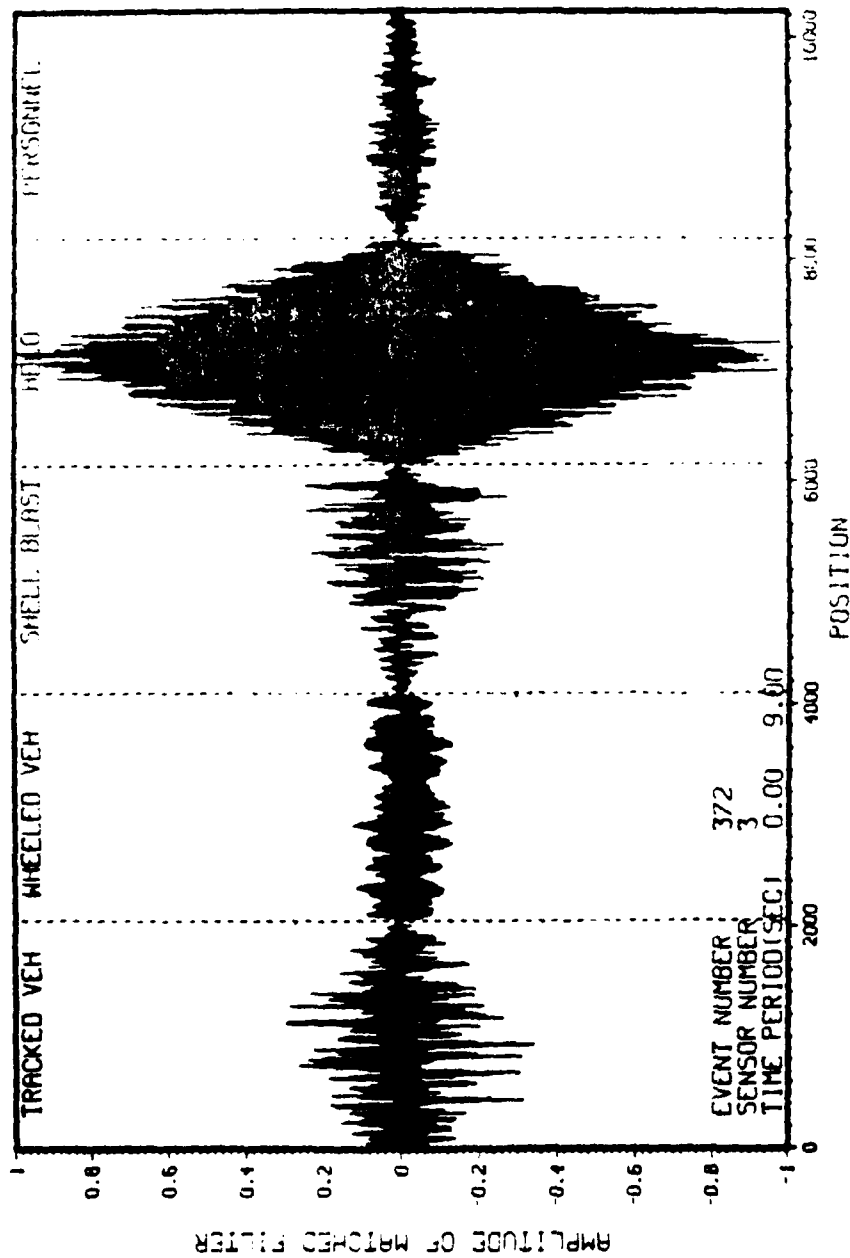


Figure 6.27 Matched Filter Response for Event 372

SENSOR UNIT

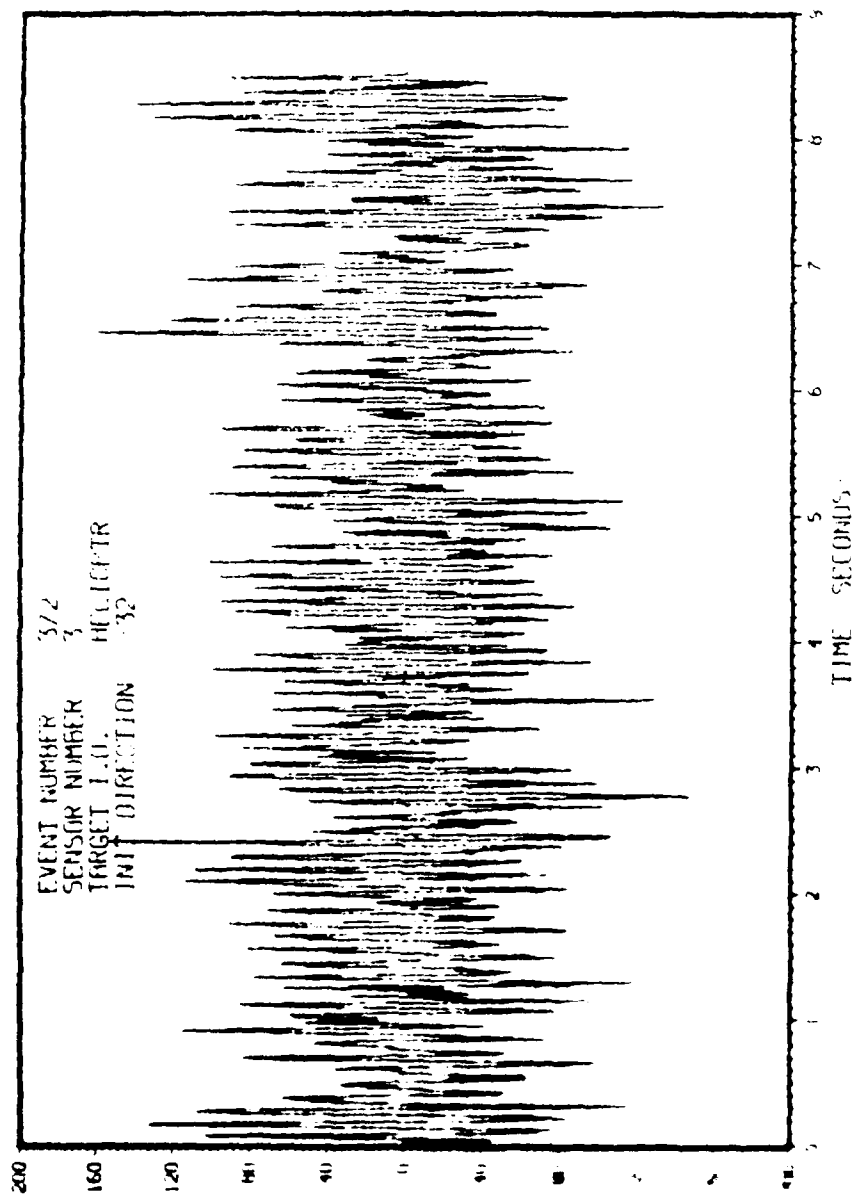


Figure 6.28 Amplitude Response for Event 372

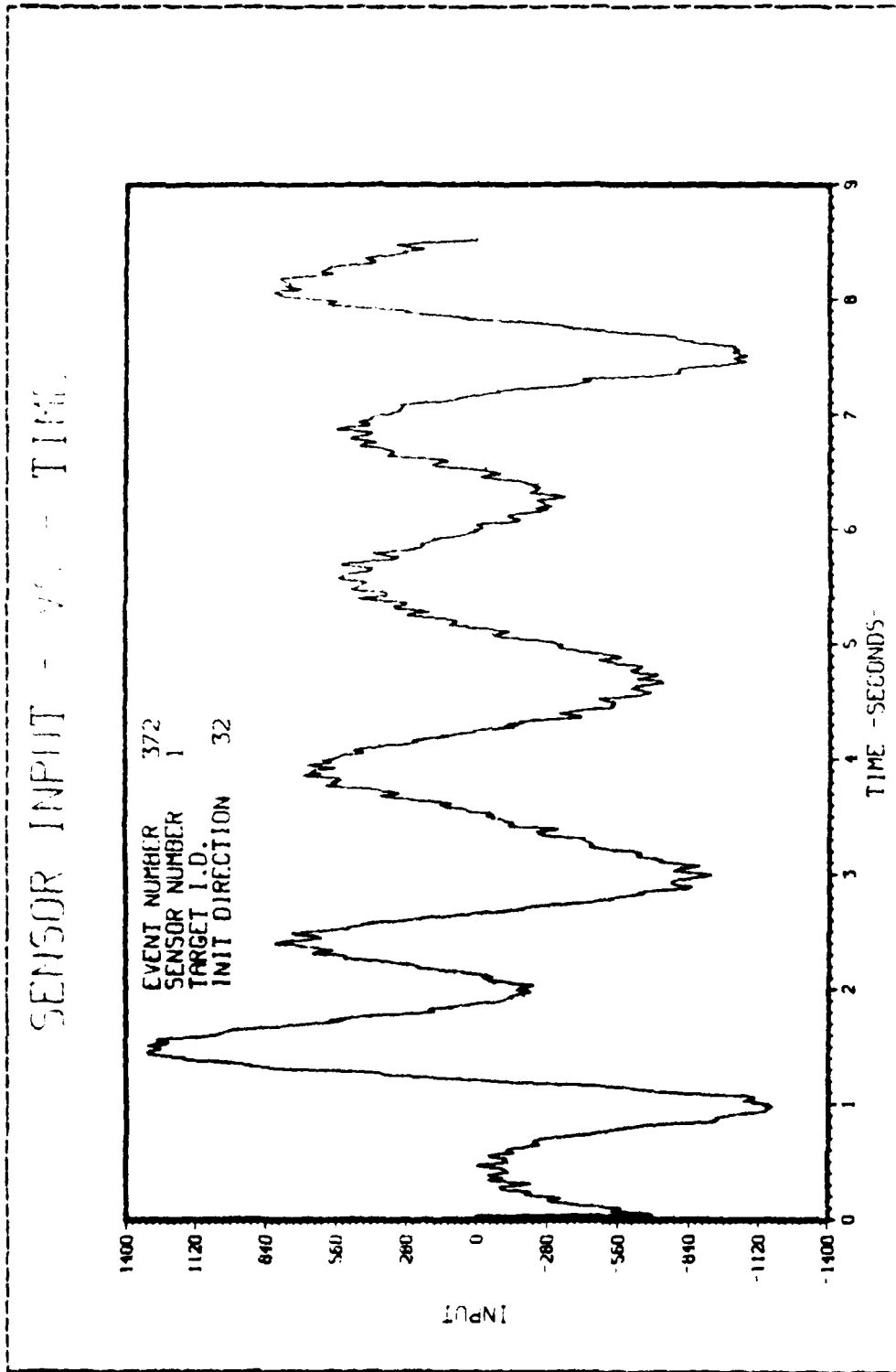


Figure 6.29 Amplitude Response of Malfunctioning Sensor for Event 372

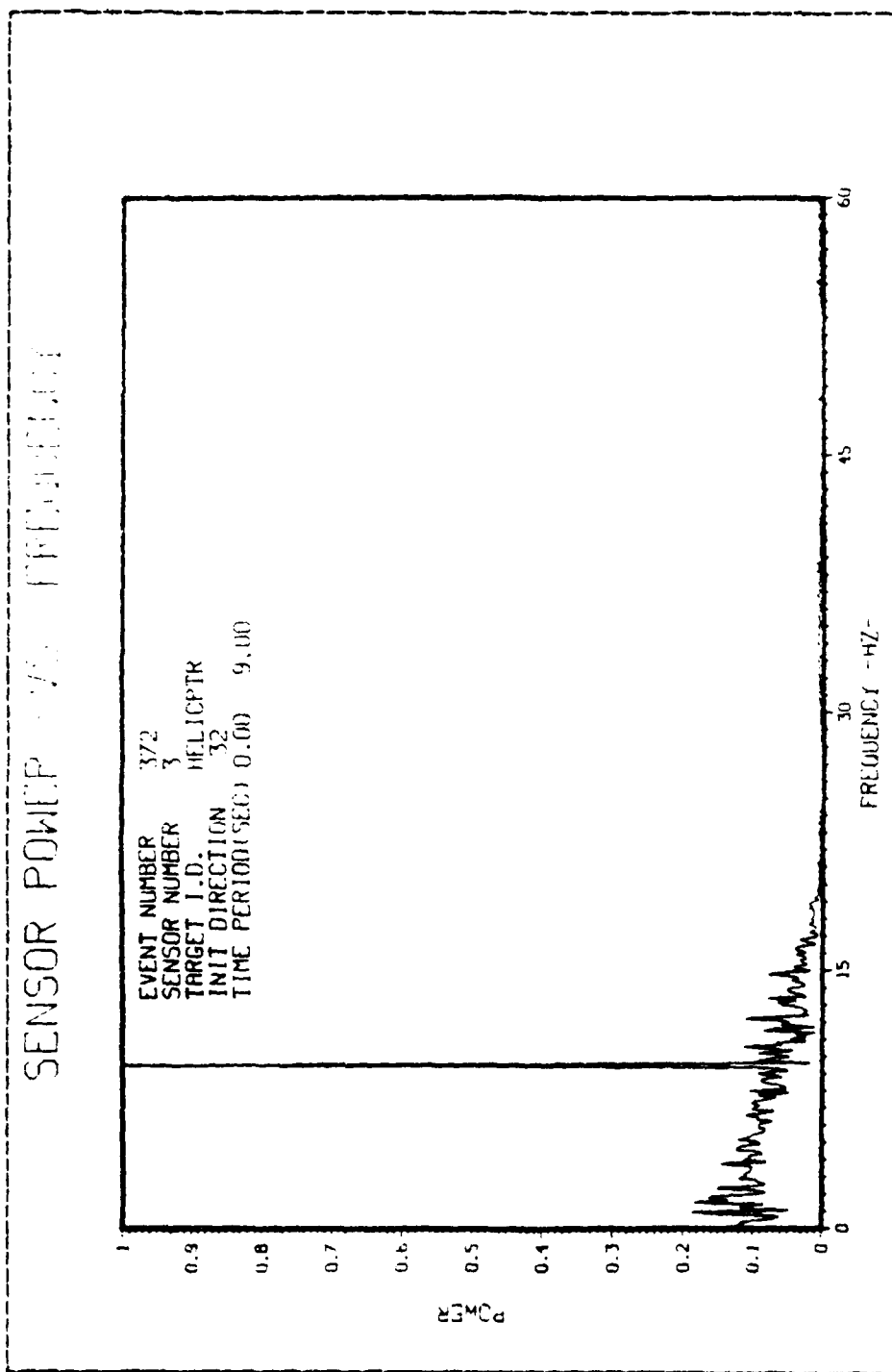


Figure 6.30 Frequency Response for Event 372

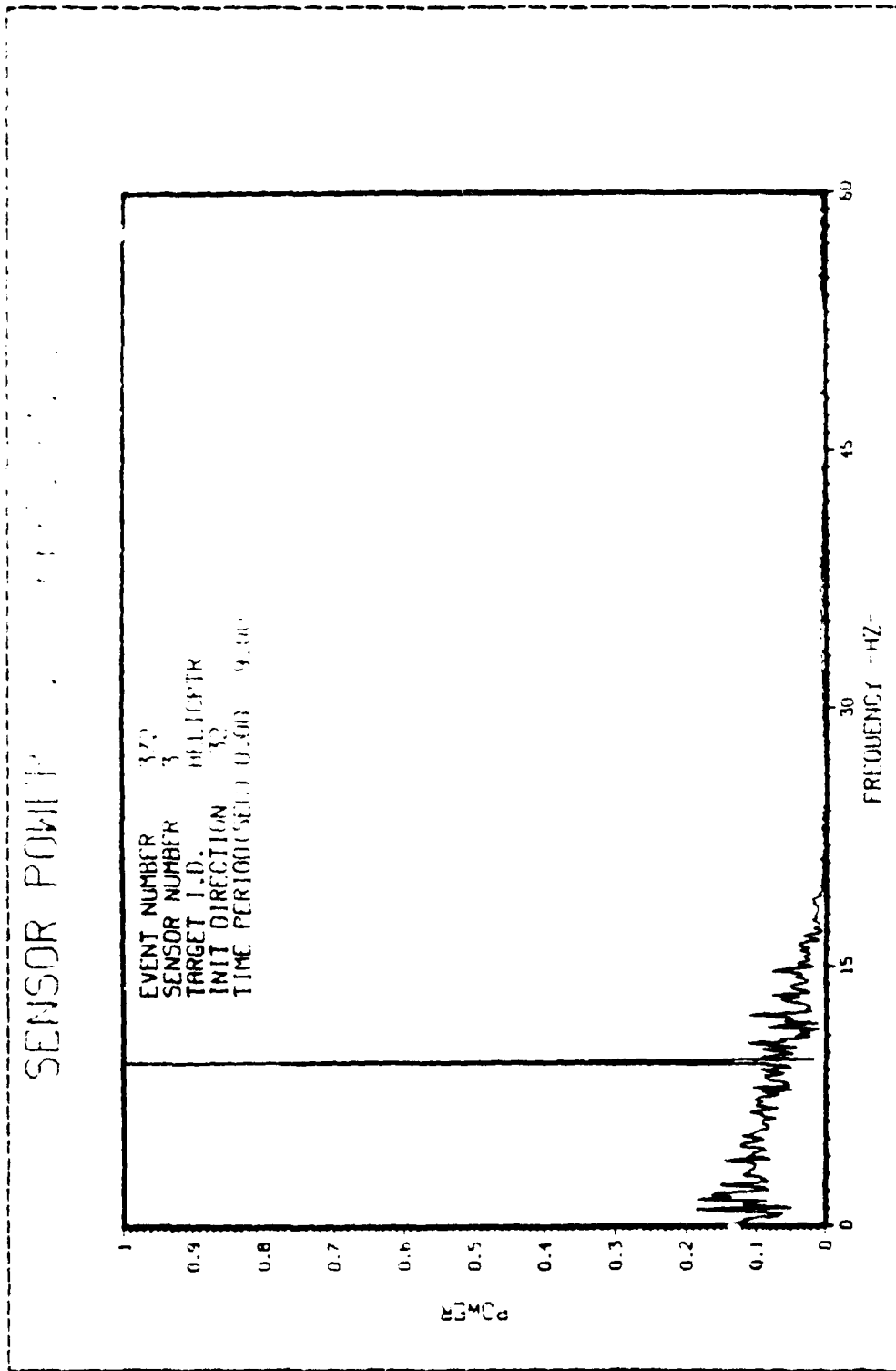


Figure 6.30 Frequency Response for Event 372

LEAST MEAN SQUARES POLYNOMIAL

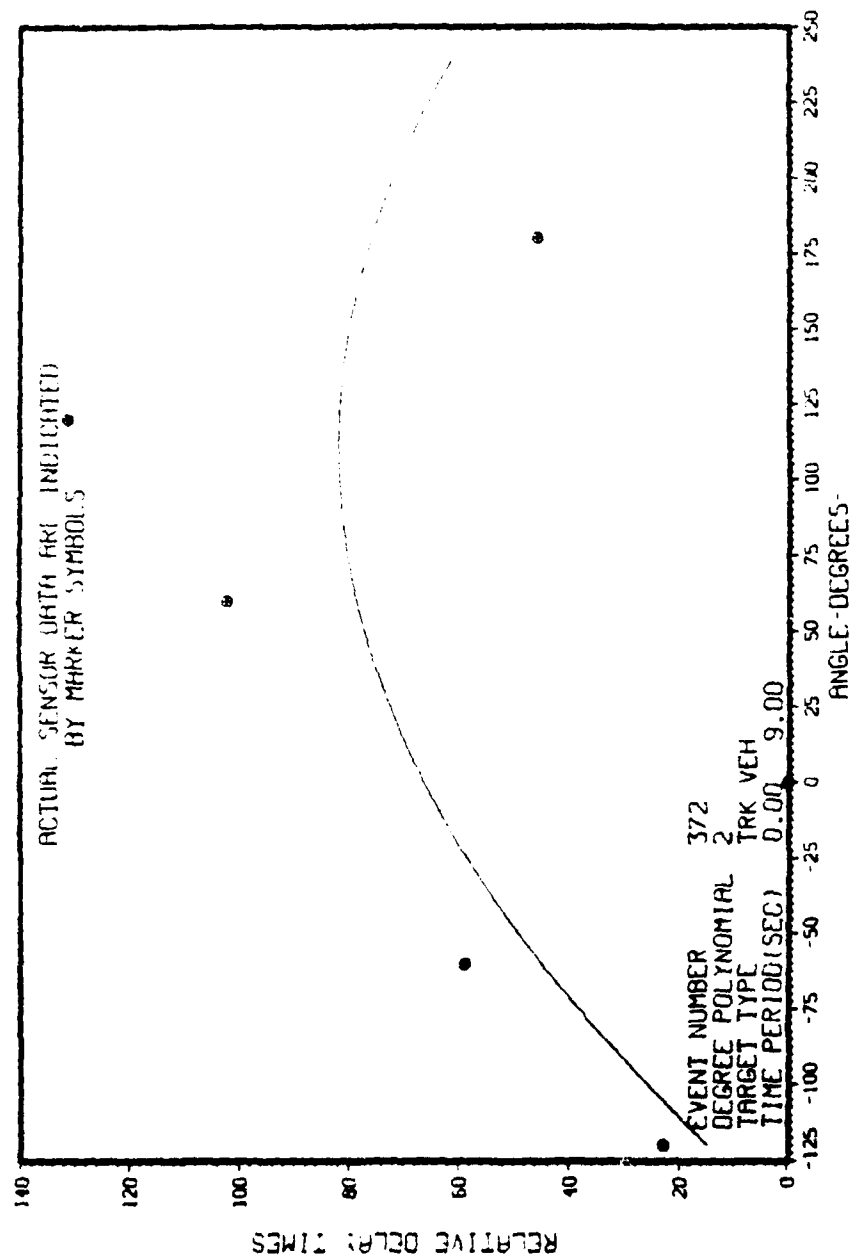


Figure 6.31 LMSP Matched Filter Direction for Event 372

MULTIPLE TARGET - MATCHED FILTER OUTPUT

EVENT NUMBER	372
TIME PERIOD(SEC)	0.00 9.00
TRACKED VEHICLE	DIRECTION - 0.00
HELICOPTER	DIRECTION - 0.00
SIMULATED TRKD VEHICLE	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED WILD VEHICLE	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED HELICOPTER	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED PERSONNEL	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000

Figure 6.32 LMSP Multiple Target Direction Summary for Event 372

LEAST MEAN SQUARES POLYNOMIAL

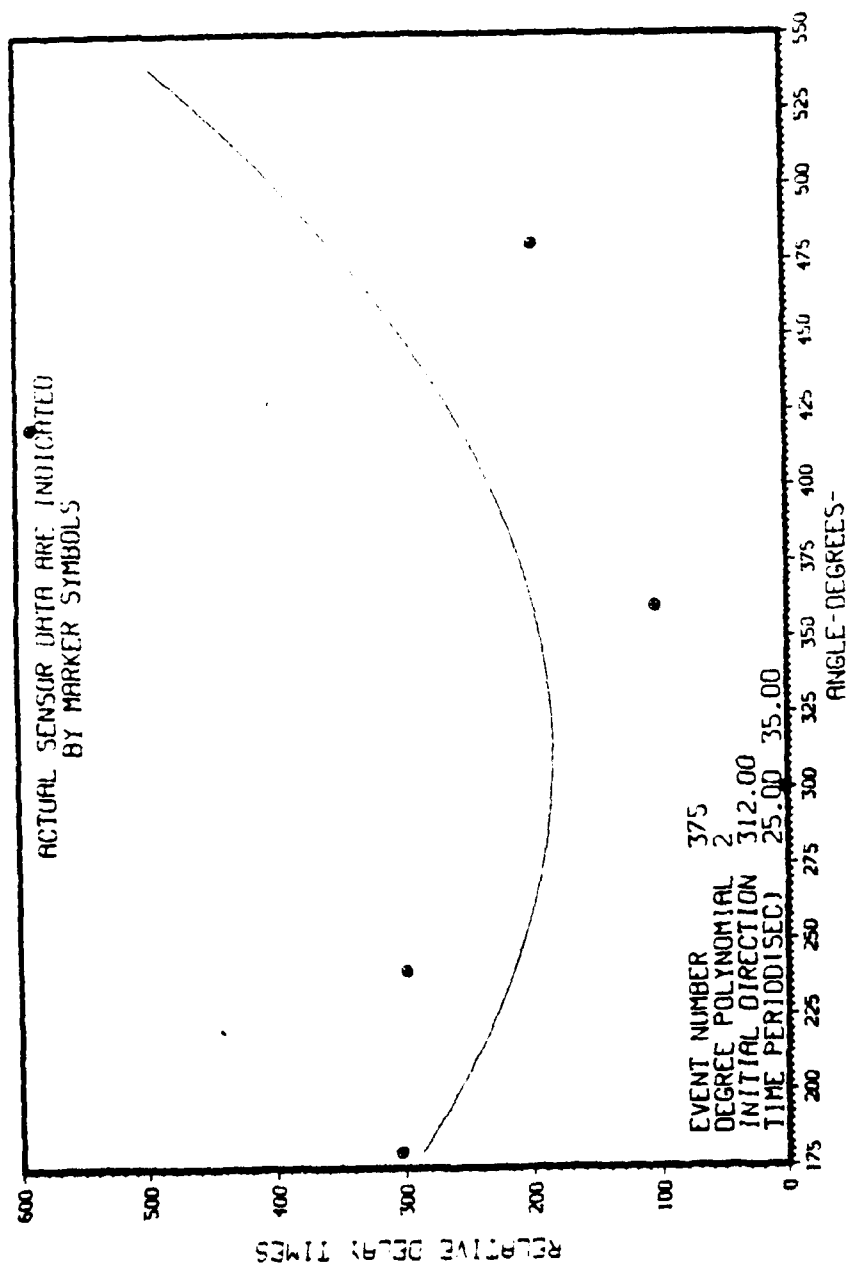


Figure 6.33 LMSP Initial Direction for Event 375

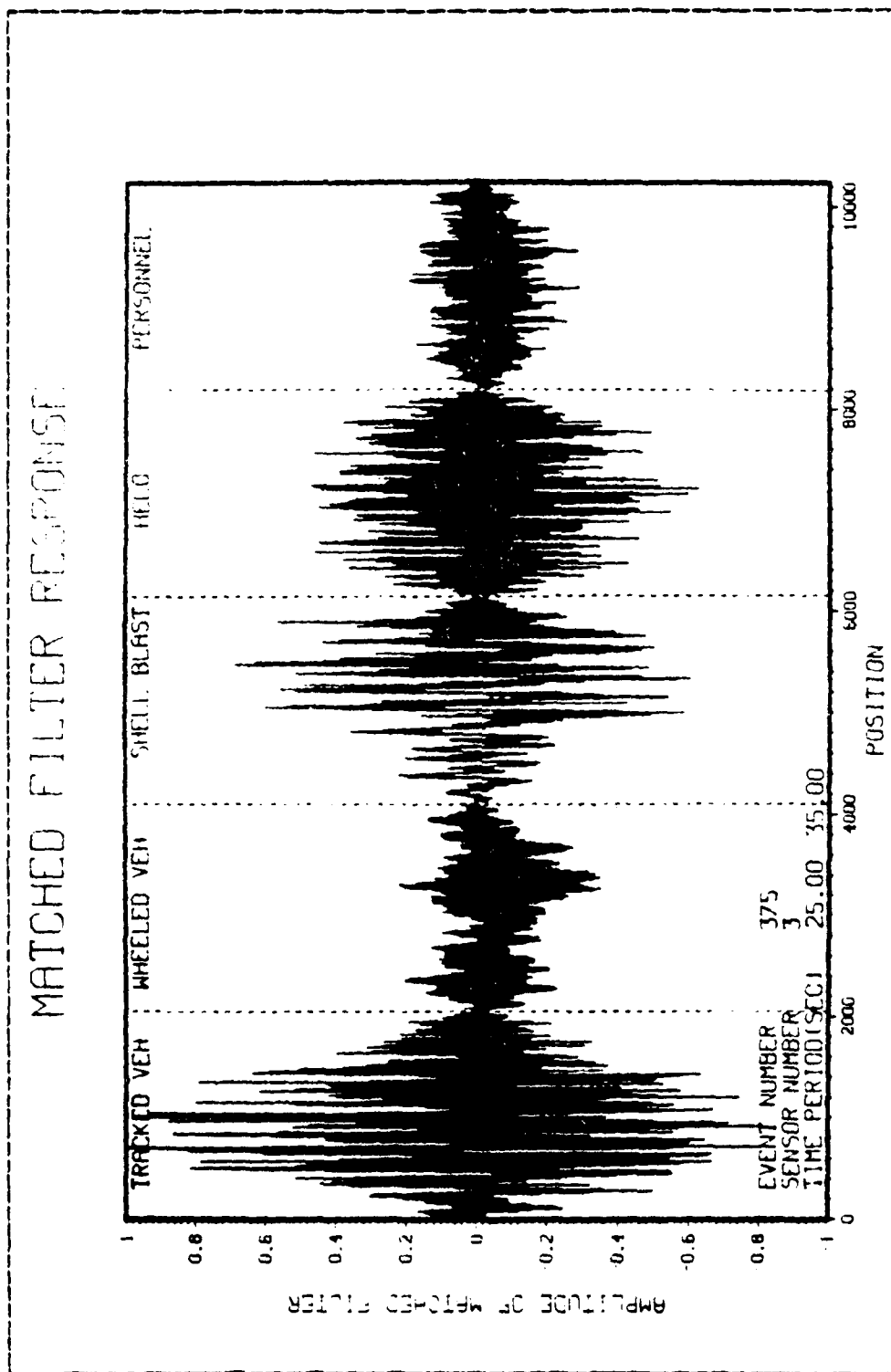


Figure 6.34 Matched Filter Response for Event 375

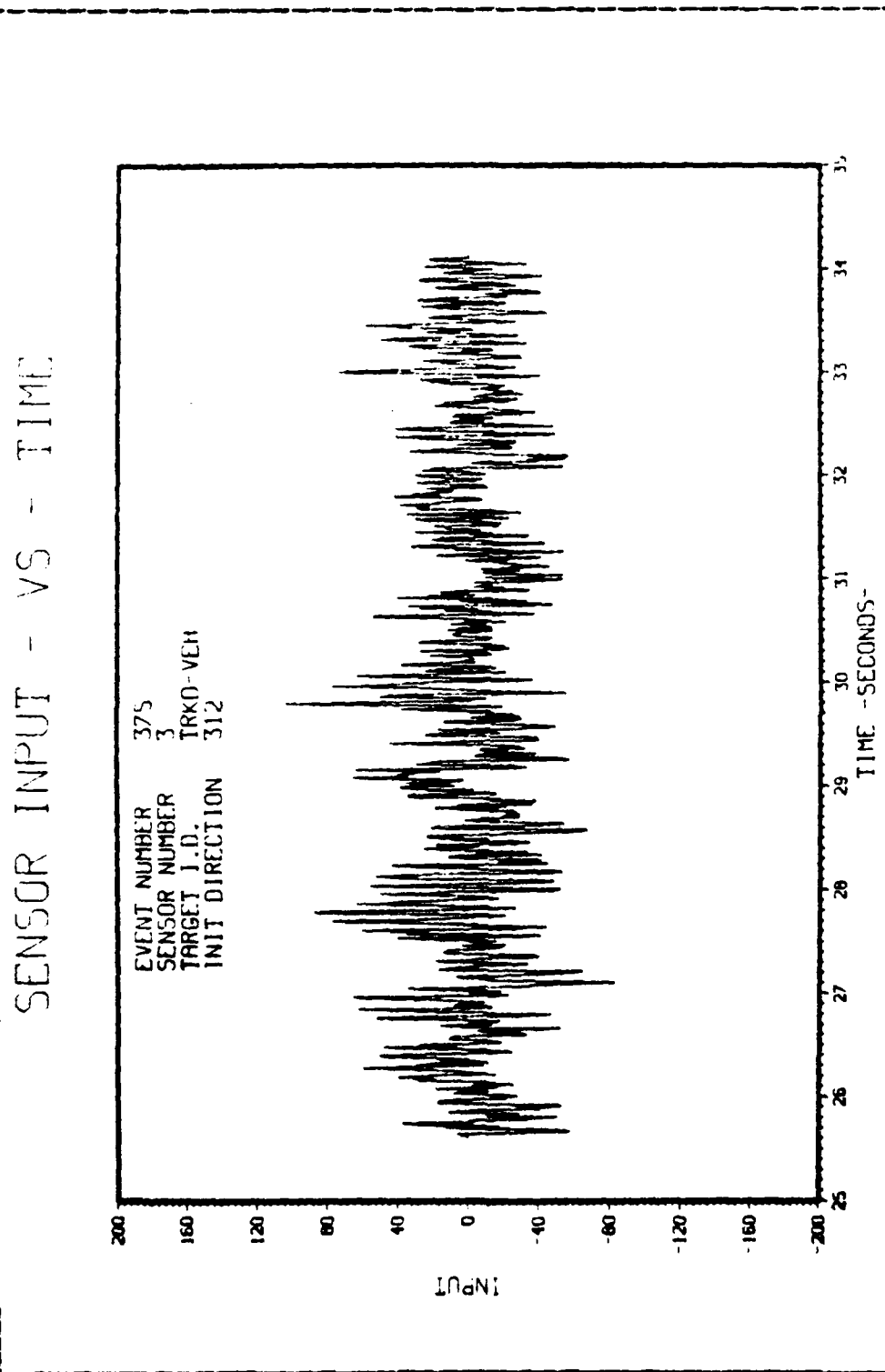


Figure 6.35 Amplitude Response for Event 375

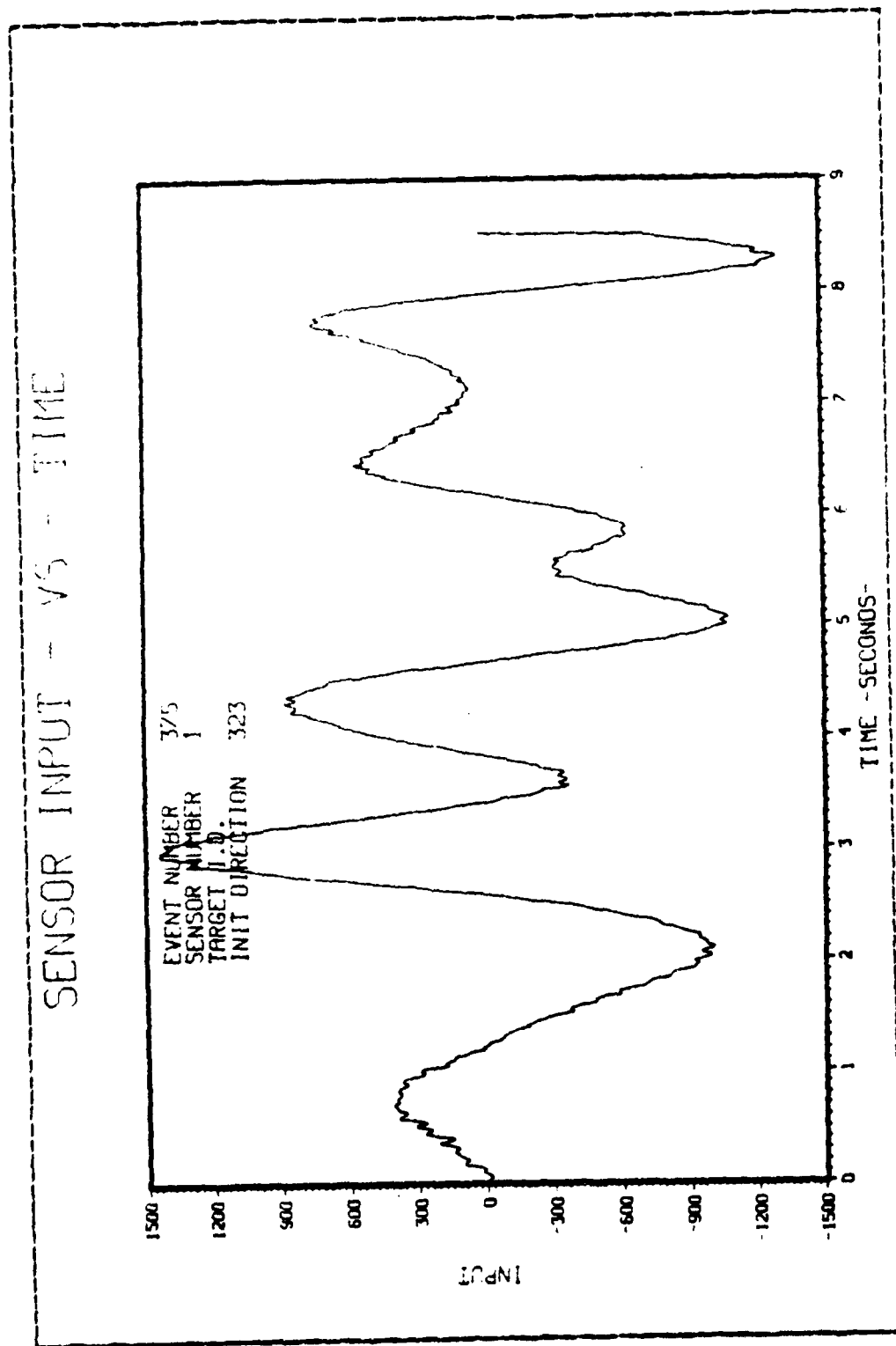


Figure 6.36 Amplitude Response of Malfunctioning Sensor for Event 375

SENSOR POWER -VS- FREQUENCY

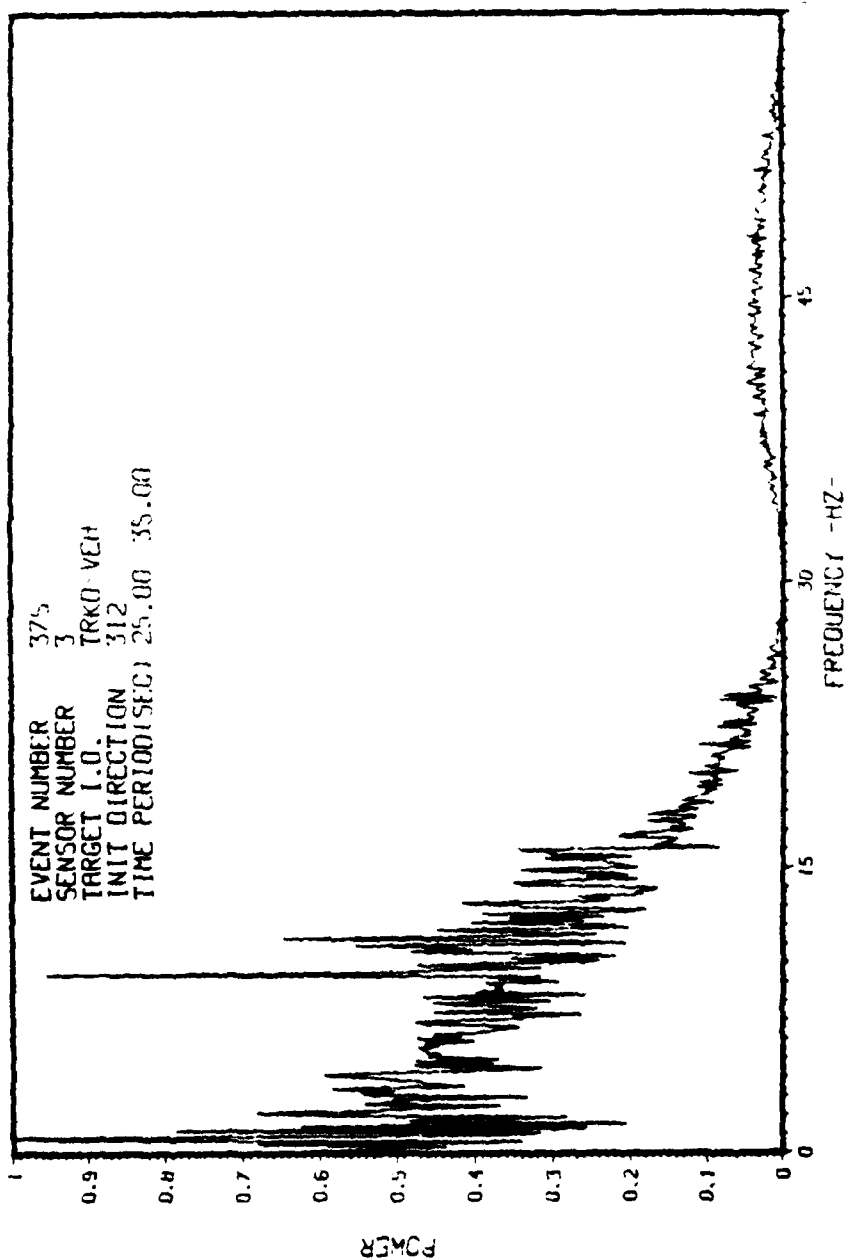


Figure 6.37 Frequency Response for Event 375

LEAST MEAN SQUARES POLYNOMIAL

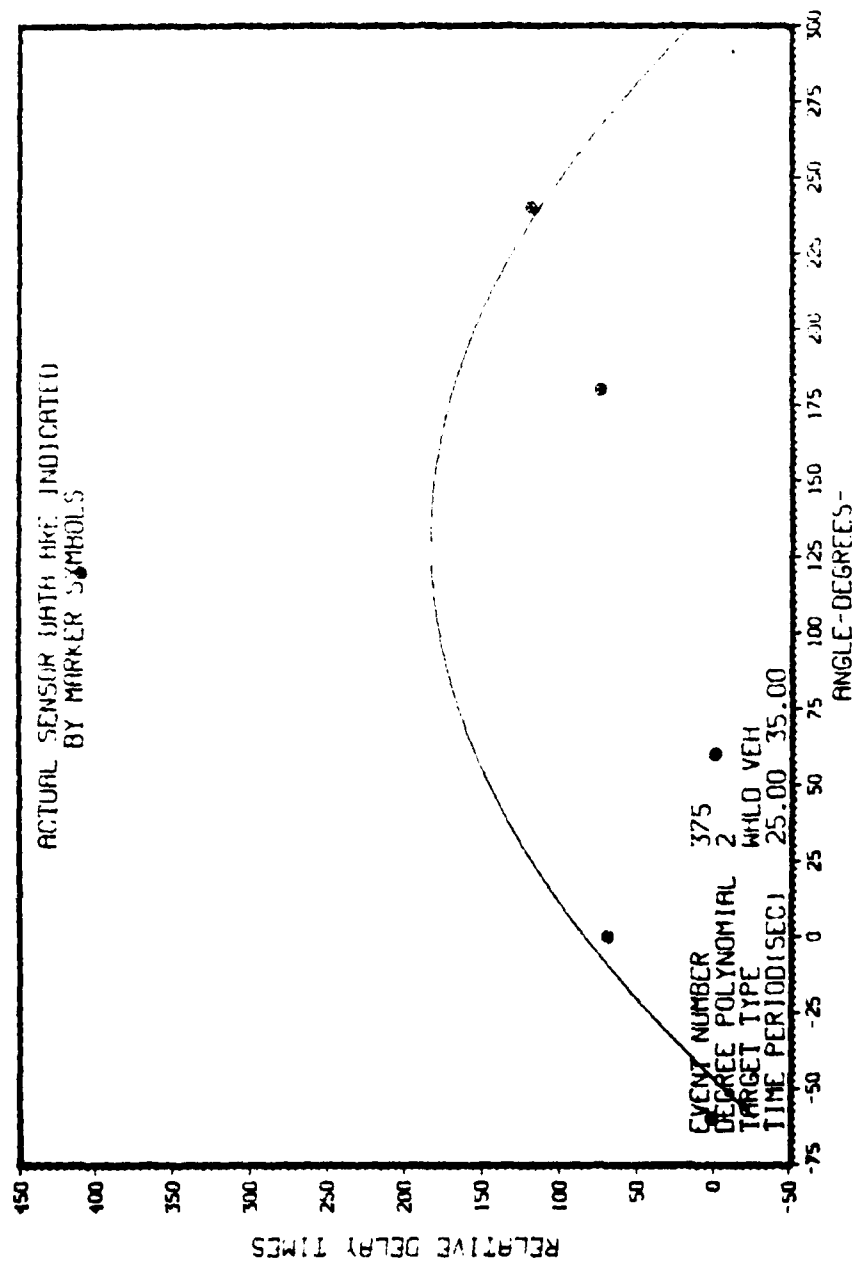


Figure 6.38 LMSP Matched Filter Direction for Event 375

MULTIPLE TARGET - MATCHED FILTER OUTPUT

EVENT NUMBER	375
TIME PERIOD(SEC)	25.00 35.00
TRACKED VEHICLE	DIRECTION - 0.00
SHELL BLAST	DIRECTION - 312.00
HELICOPTER	DIRECTION - 0.00
SIMULATED TRKD VEHICLE	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED WILD VEHICLE	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED HELICOPTER	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED PERSONNEL	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000

Figure 6.39 LAMP Multiple Target Direction Summary for Event 375

LEAST MEAN SQUARES POLYNOMIAL

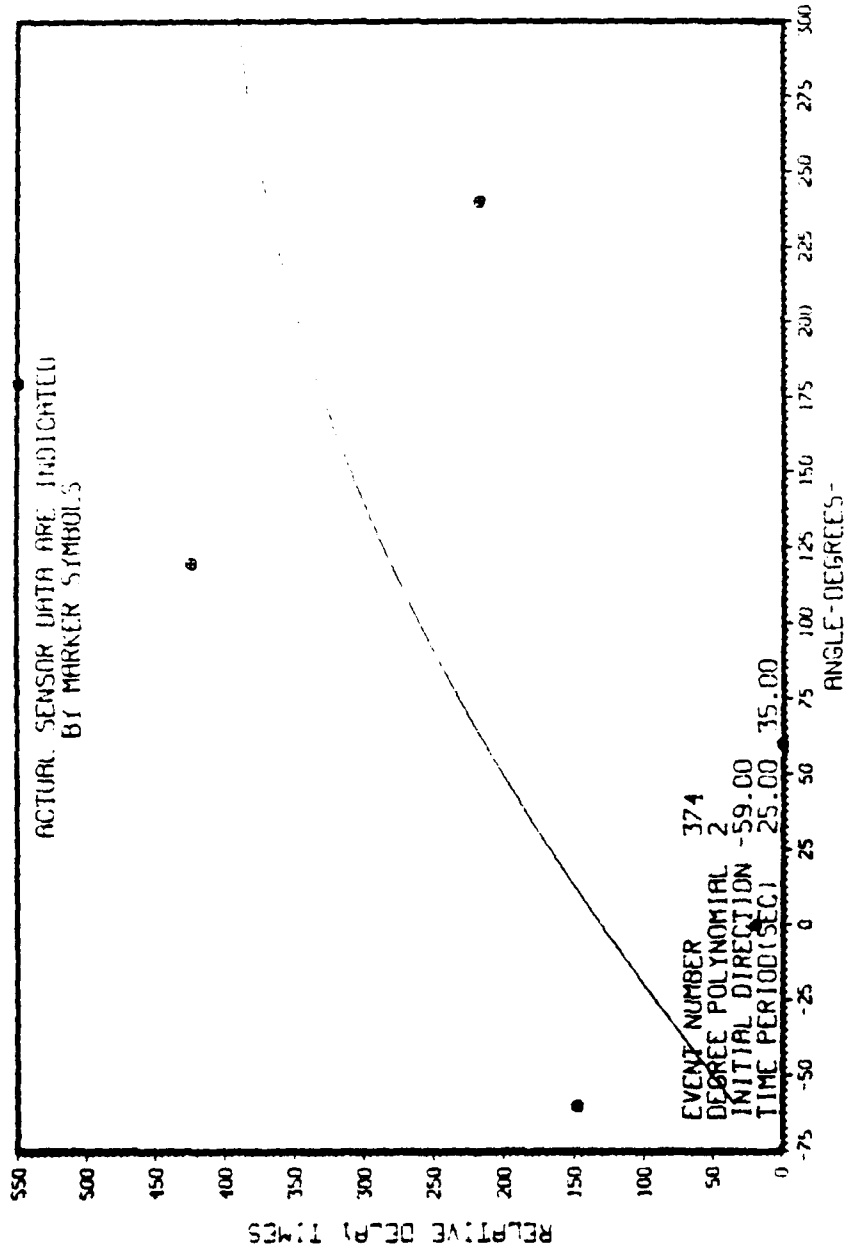


Figure 6.40 LMSP Initial Direction for Event 374

MATCHED FILTER RESPONSE

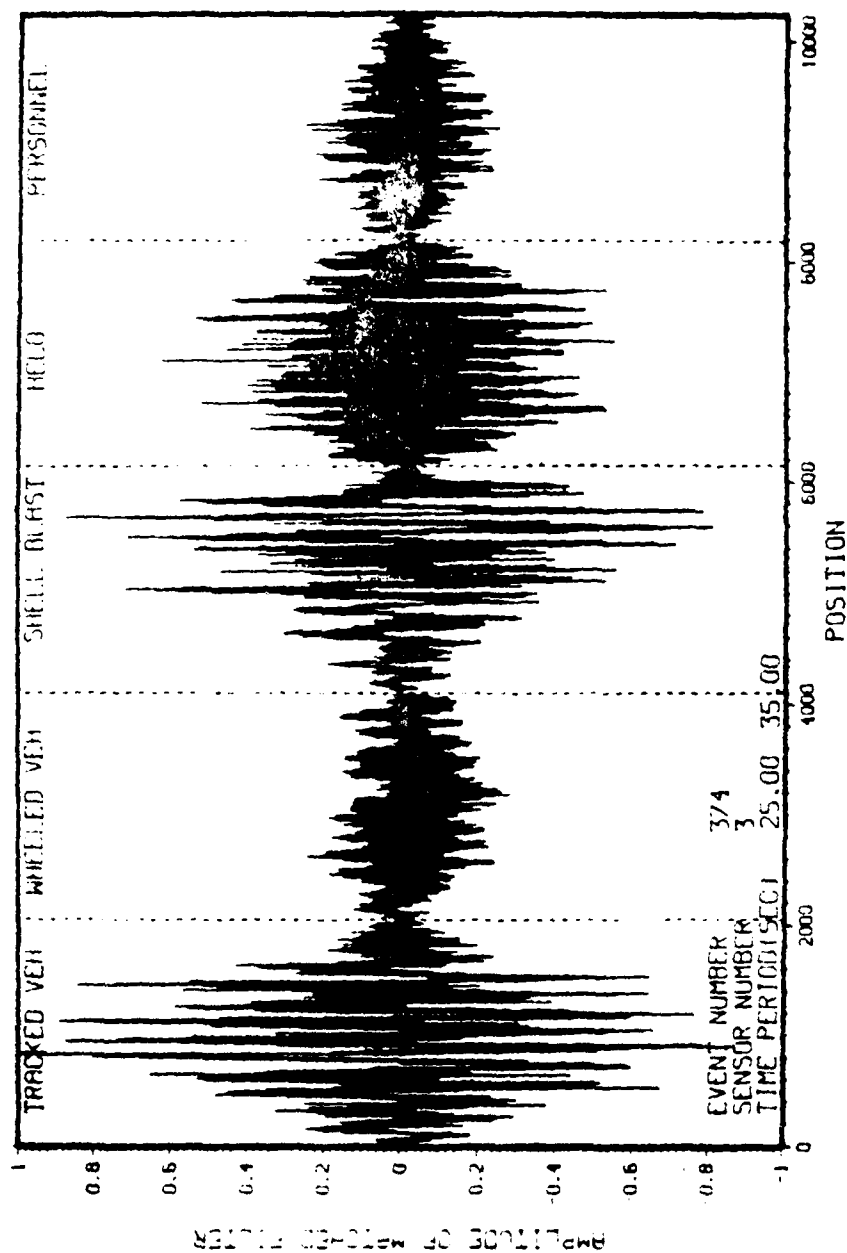


Figure 6.41 Matched Filter Response for Event 374

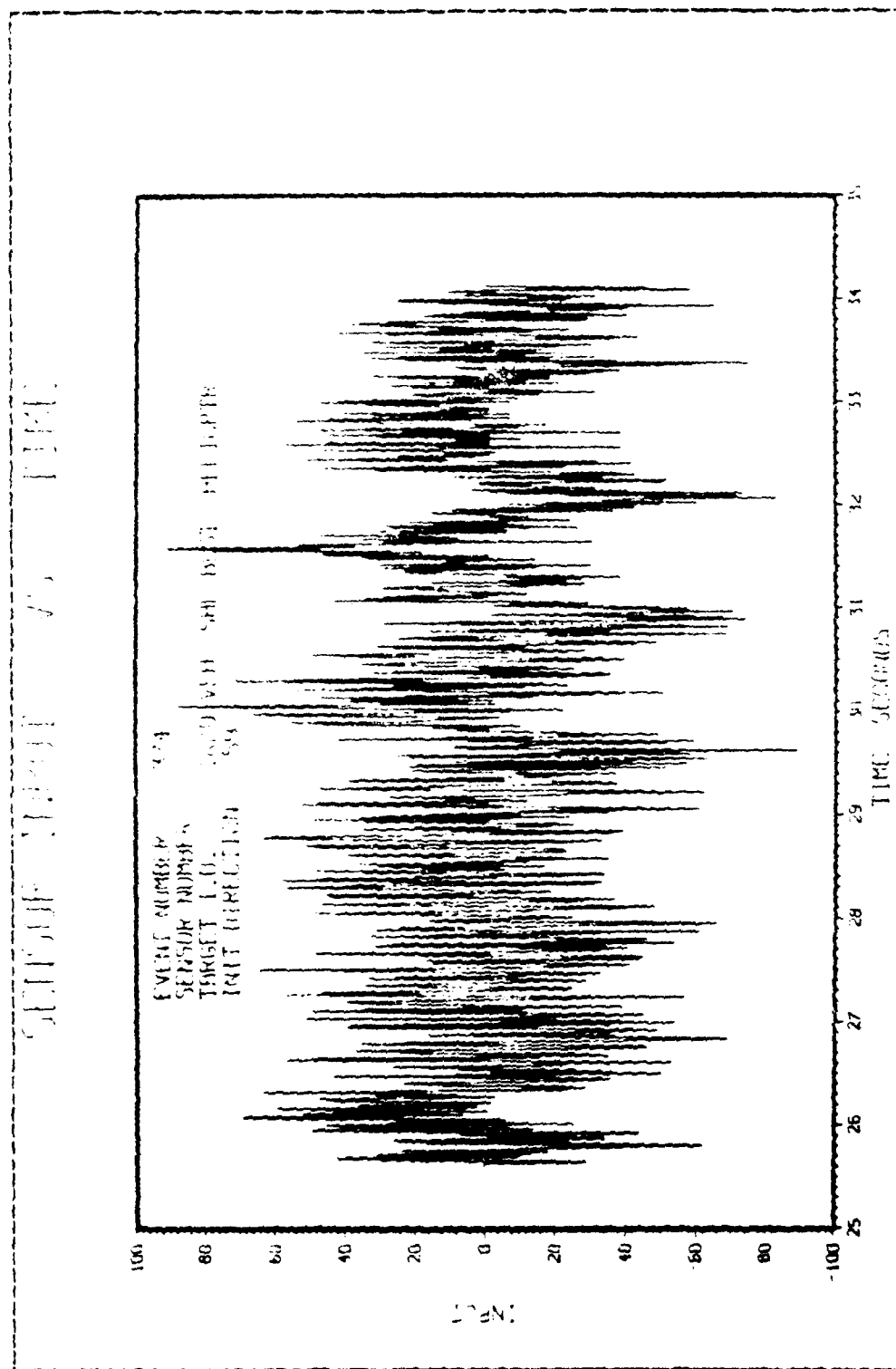


Figure 6.42 Amplitude Response for Event 374

SENSOR INPUT - VS - TIME

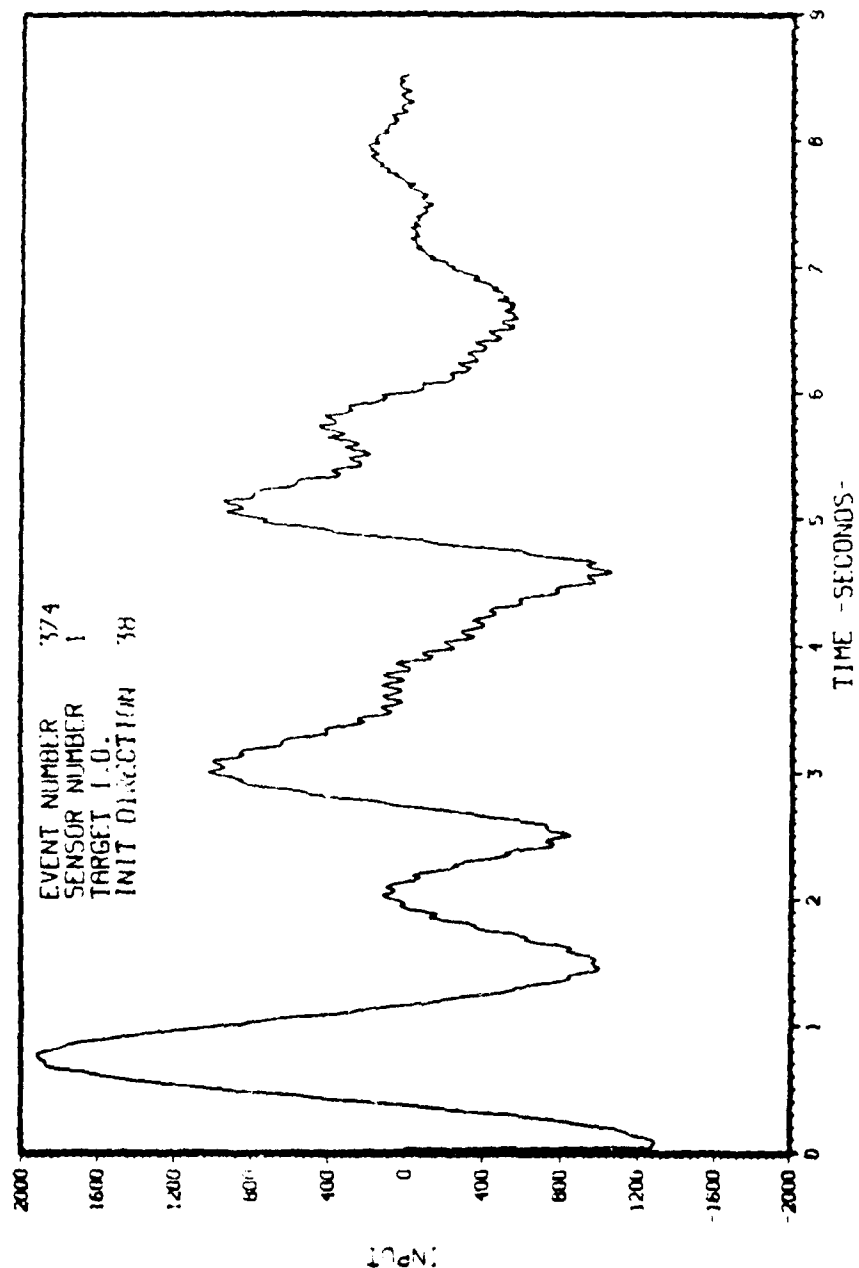


Figure 6.43 Amplitude Response of Malfunctioning Sensor for Event 374

AD-A138 914

MULTIPLE TARGET IDENTIFICATION AND DIRECTION FINDING
USING MATCHED FILTERING TECHNIQUES(U) NAVAL
POSTGRADUATE SCHOOL MONTEREY CA J L JOHNSTON DEC 83

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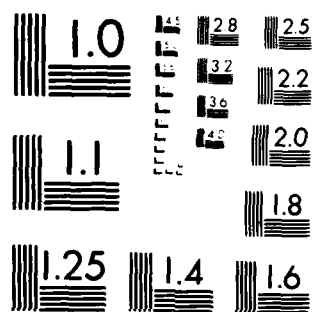
END

DATE

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4 84

DTIC



MICROCOPY RESOLUTION TEST CHART
NATIONAL BUREAU OF STANDARDS-1963-A

SENSOR POWER - VS- FREQUENCY

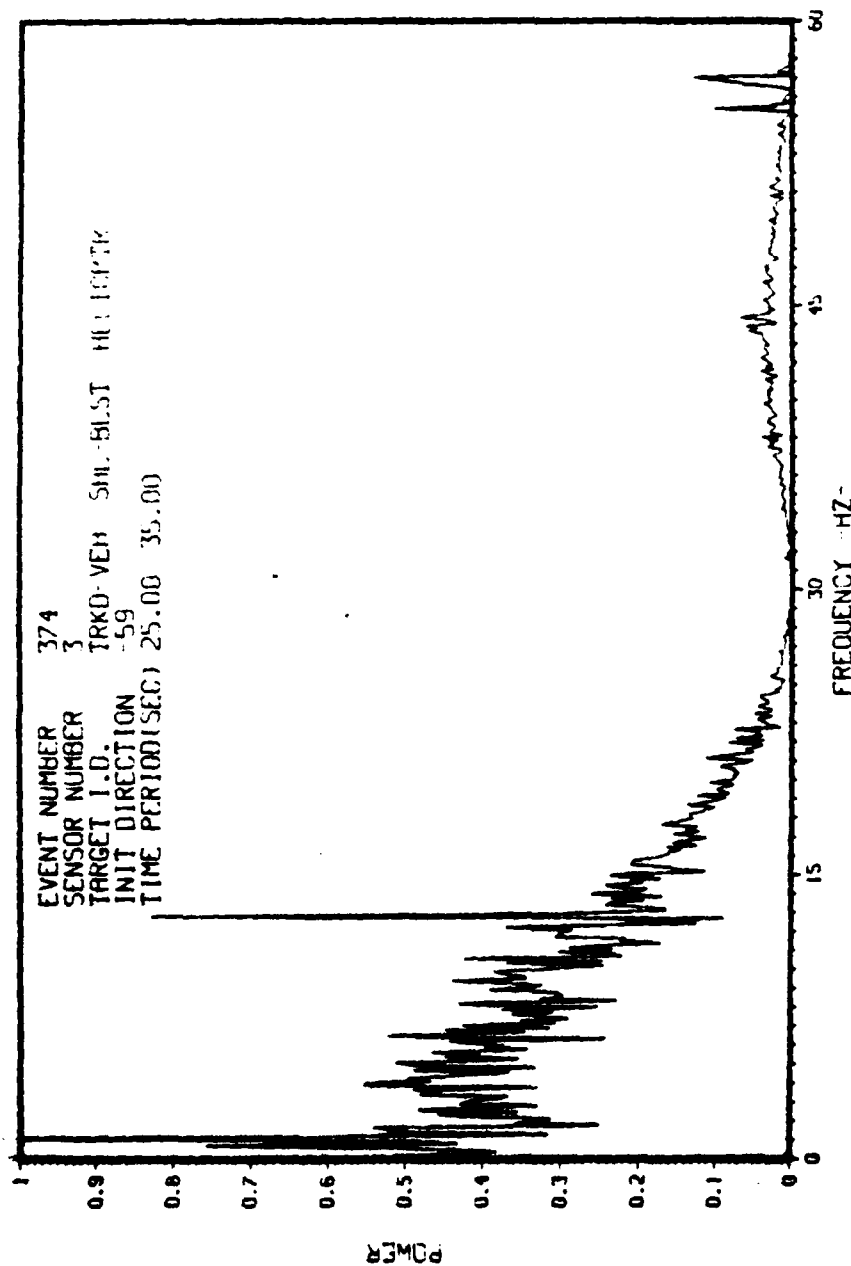


Figure 6.44 Frequency Response for Event 374

LEAST MEAN SQUARES POLYNOMIAL

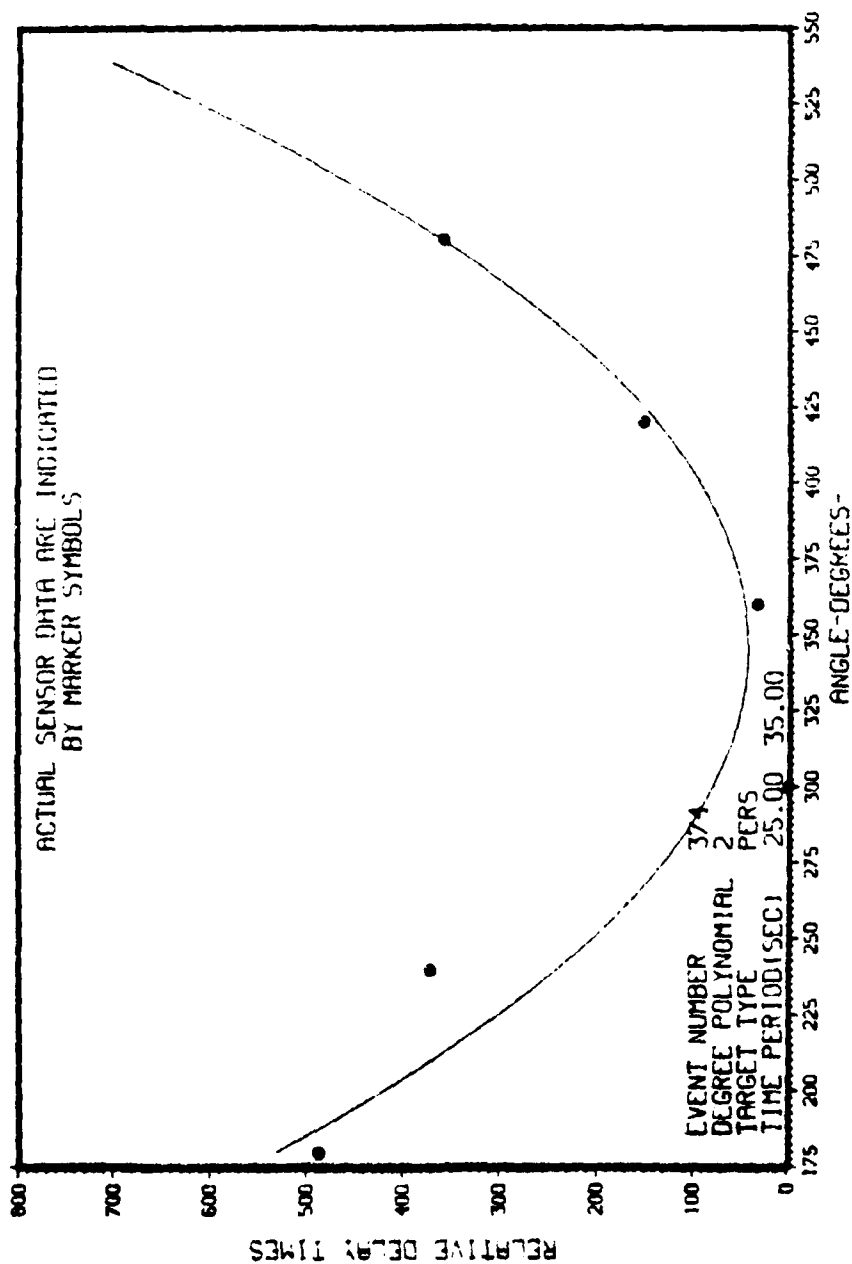


Figure 6.45 LMSP Matched Filter Direction for Event 374

MULTIPLE TARGET - MATCHED FILTER OUTPUT

EVENT NUMBER	374
TIME PERIOD(SEC)	25.00 35.00
TRACKED VEHICLE	DIRECTION - 59.00
SHELL BLAST	DIRECTION - 59.00
PERSONNEL	DIRECTION - 347.00
SIMULATED TRKD VEHICLE	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED WILD VEHICLE	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED HELICOPTER	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED PERSONNEL	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000

Figure 6.46 LMS Multiple Target Direction Summary for Event 374

LEAST MEAN SQUARES POLYNOMIAL

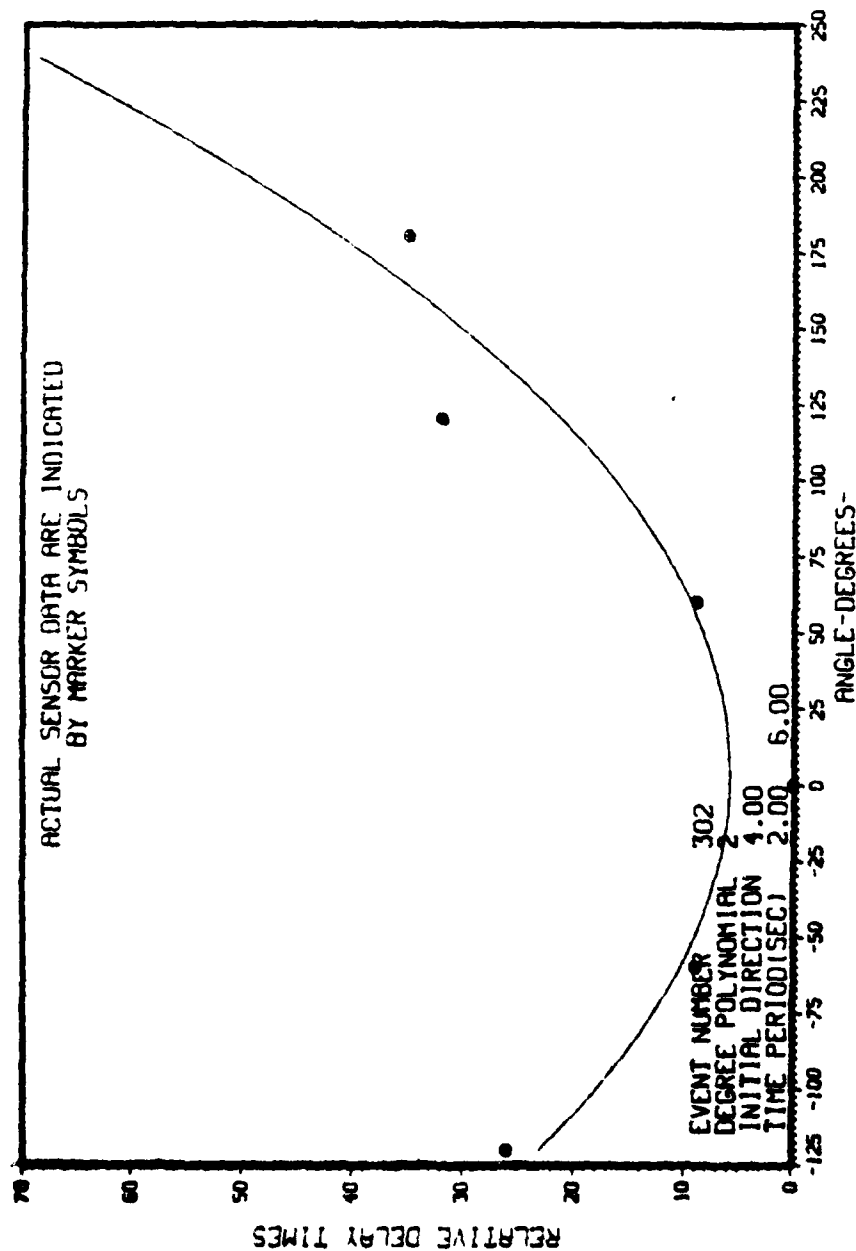


Figure 6.47 LMSP Initial Direction for Event 302

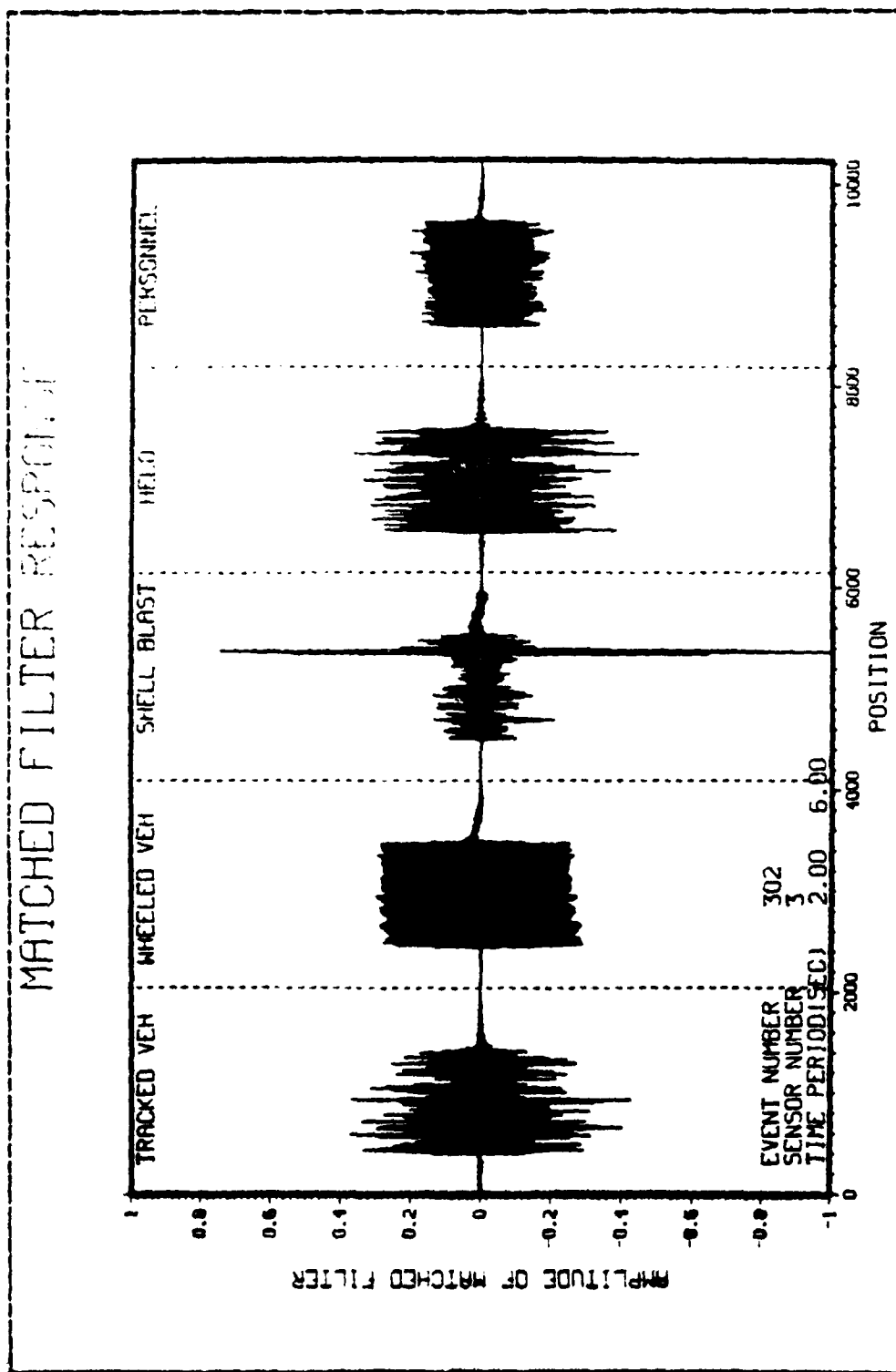


Figure 6.48 Matched Filter Response for Event 302

SENSOR INPUT - VS - TIME

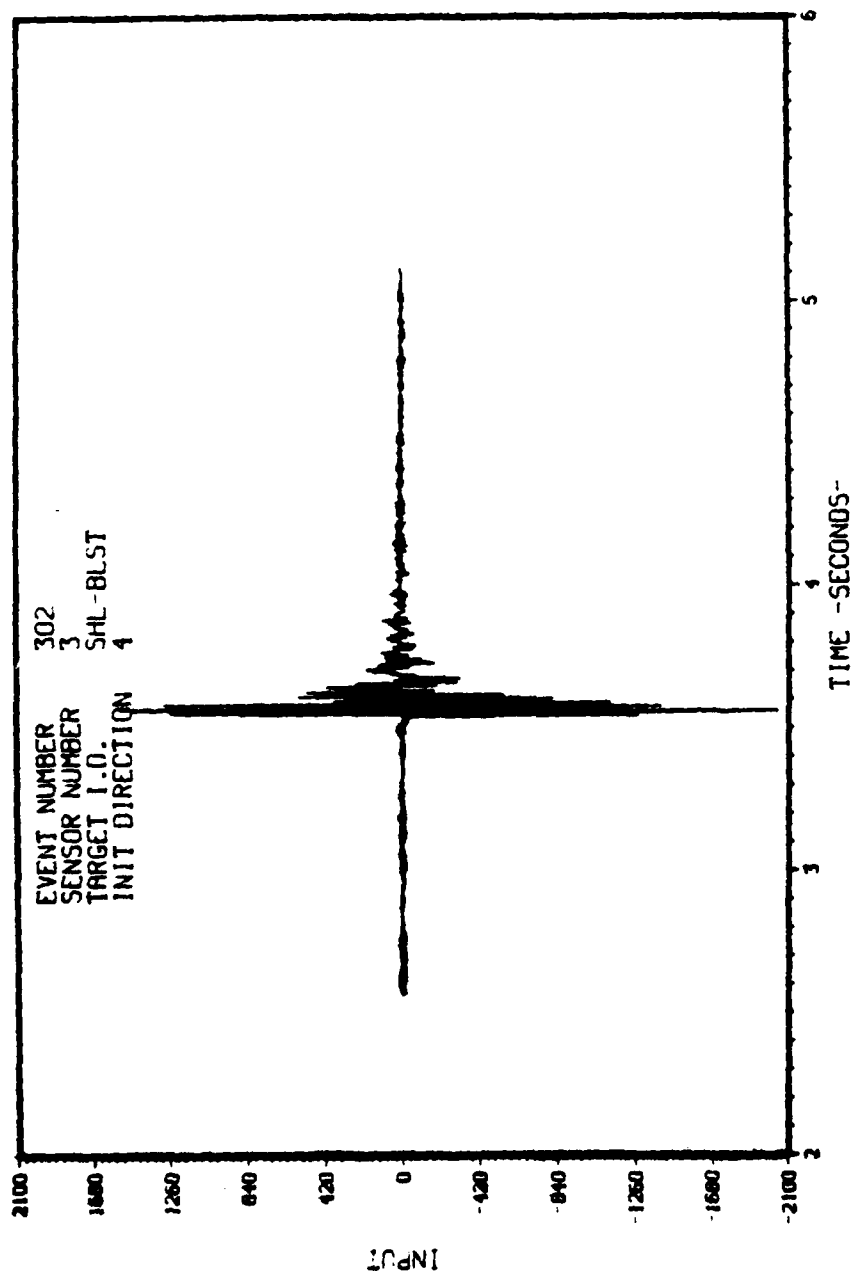


Figure 6.49 Amplitude Response for Event 302

SENSOR POWER -VS- FREQUENCY

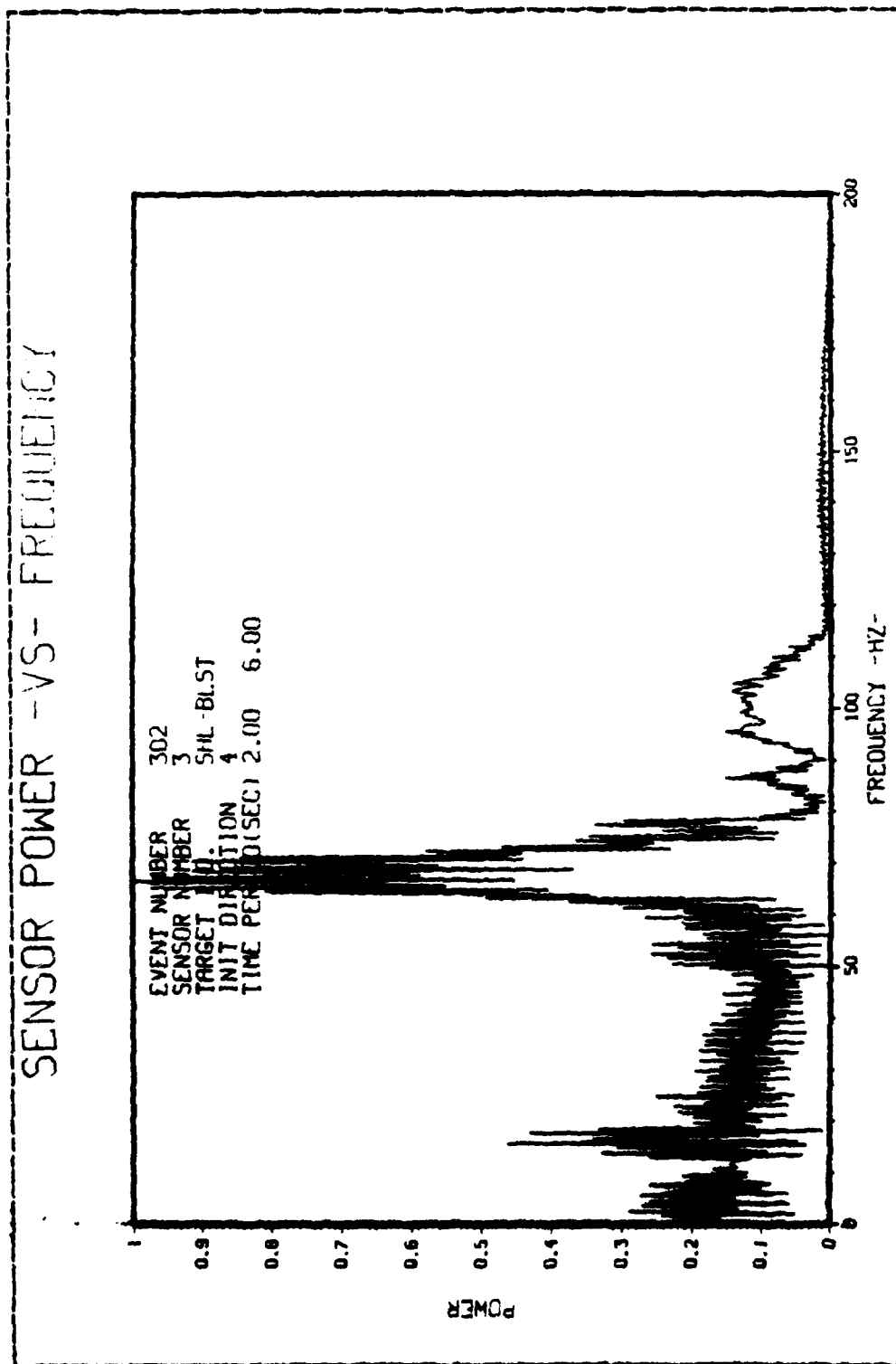


Figure 6.50 Frequency Response for Event 302

LEAST MEAN SQUARES POLYNOMIAL

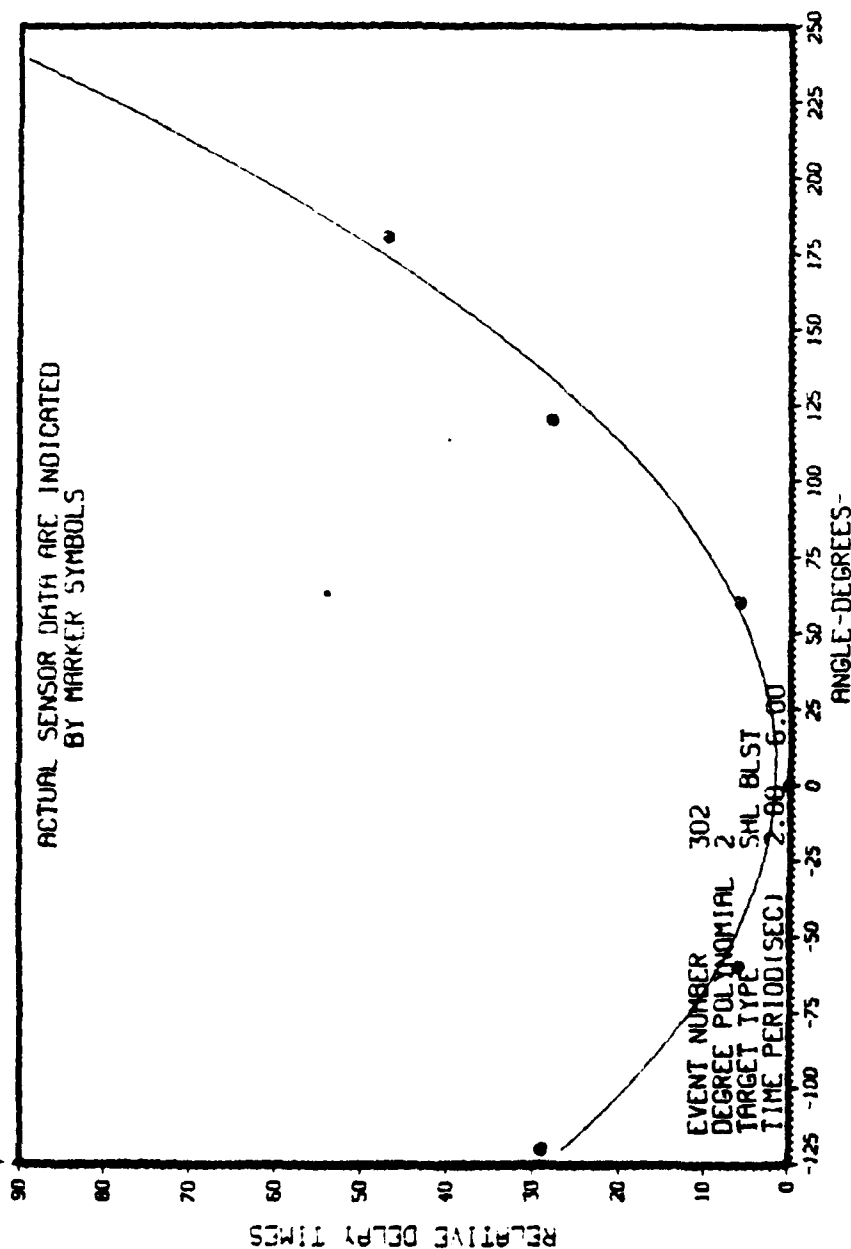


Figure 6.51 LMSP Matched Filter Direction for Event 302

MULTIPLE TARGET - MATCHED FILTER OUTPUT

EVENT NUMBER	302
TIME PERIOD(SEC)	2.00 6.00
SHELL BLAST	DIRECTION - 6.00
SIMULATED TRKO VEHICLE TARGET FREQUENCY	0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED WILD VEHICLE TARGET FREQUENCY	0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED HELICOPTER TARGET FREQUENCY	0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED PERSONNEL TARGET FREQUENCY	0.00
AMPLITUDE	0.0000
DIRECTION	0.0000

Figure 6.52 LNRP Multiple Target Direction Summary for Event 302

LEAST MEAN SQUARES POLYNOMIAL

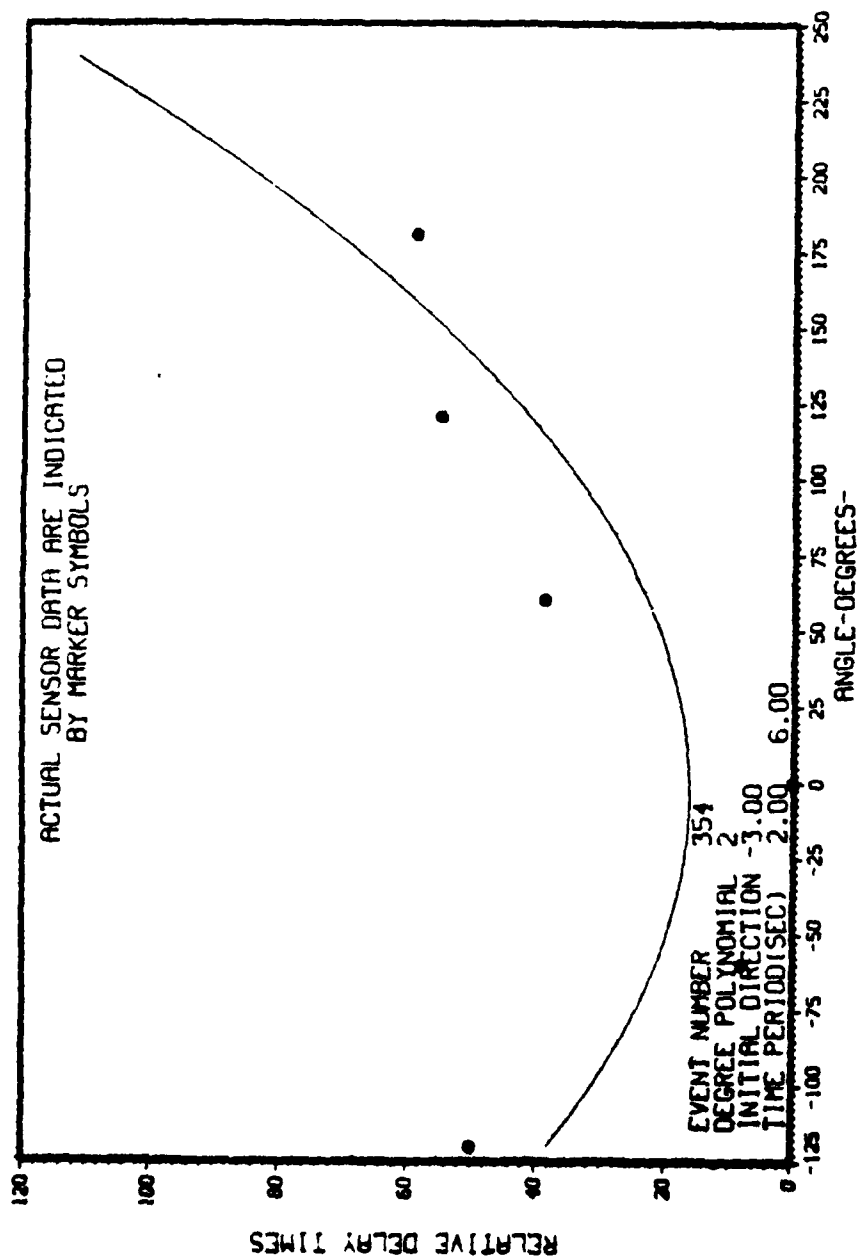


Figure 6.53 LMSP Initial Direction for Event 354 (2 - 6sec)

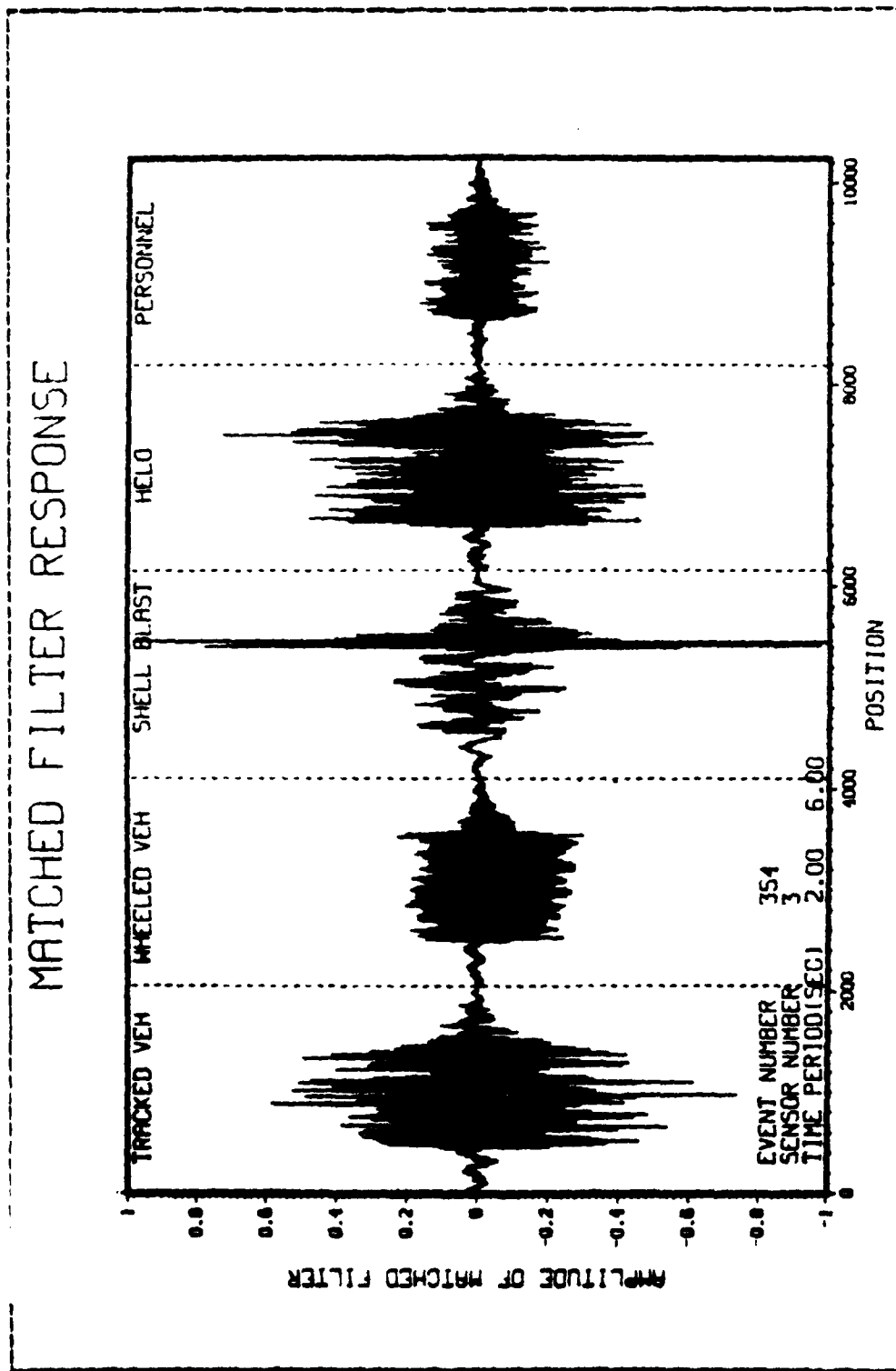


Figure 6.54 Matched Filter Res use for Event 354 (2 - 6sec)

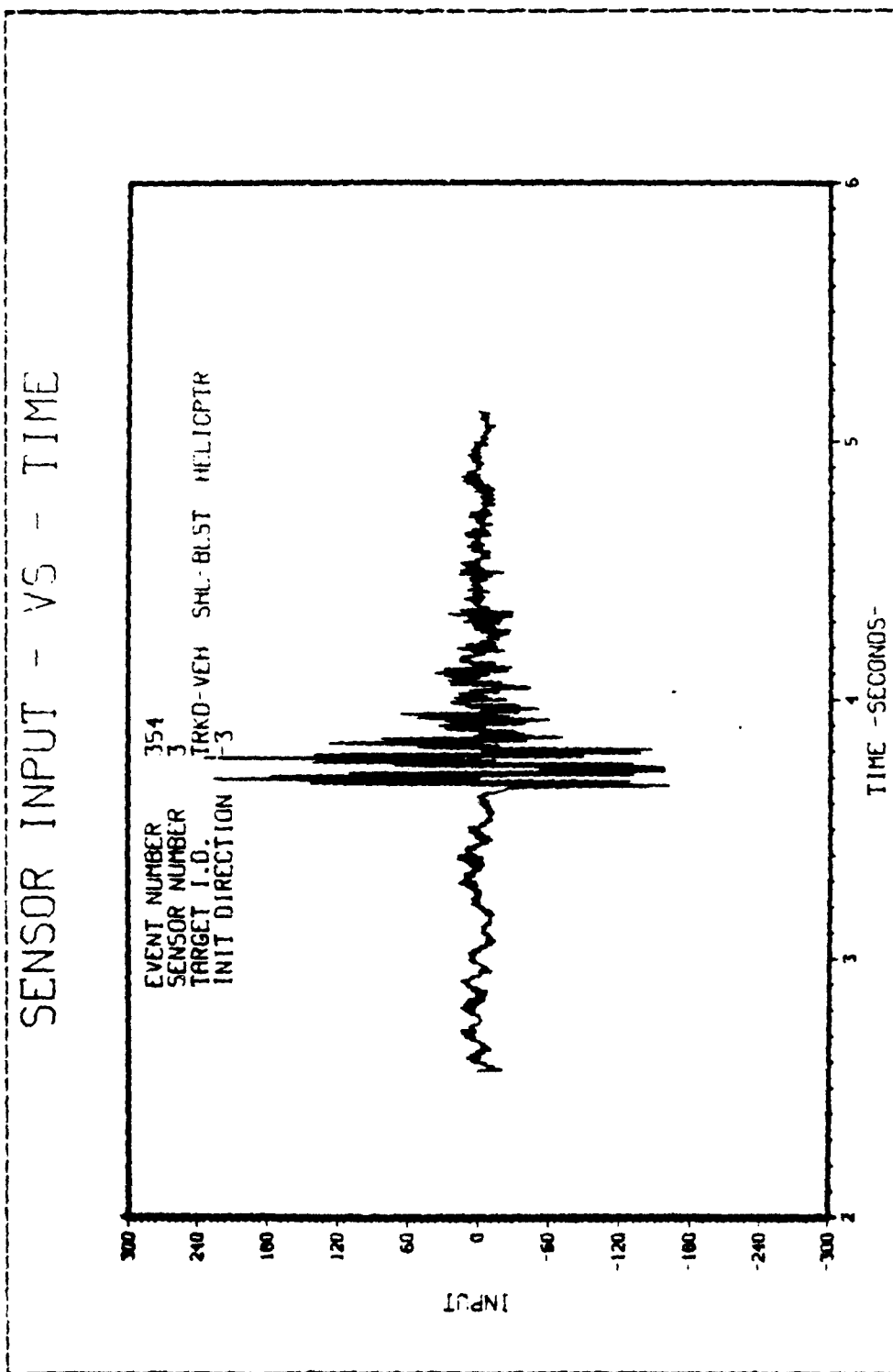


Figure 6.55 Amplitude Response for Event 354 (2 - 6sec)

SENSOR POWER -VS- FREQUENCY

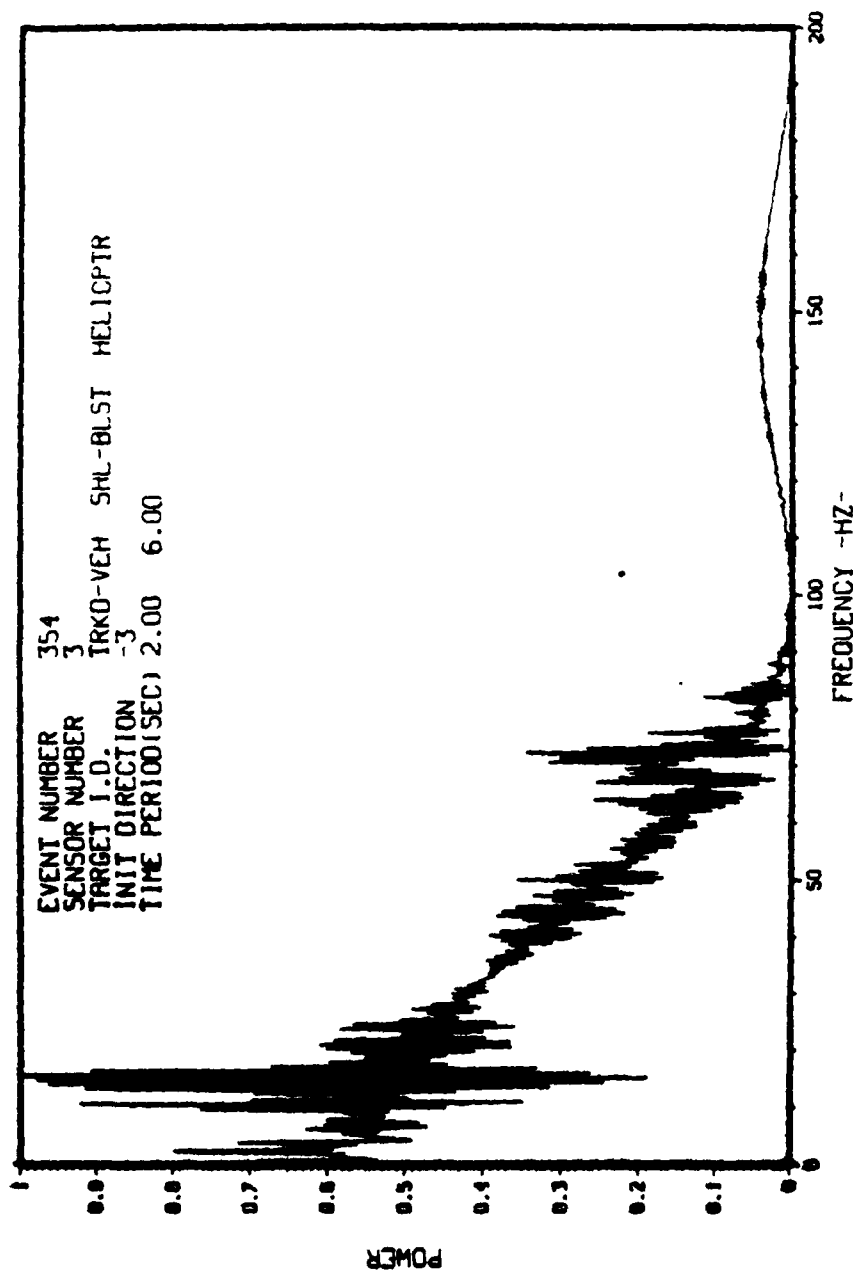


Figure 6.56 Frequency Response for Event 354 (2 - 6sec)

LEAST MEAN SQUARES POLYNOMIAL

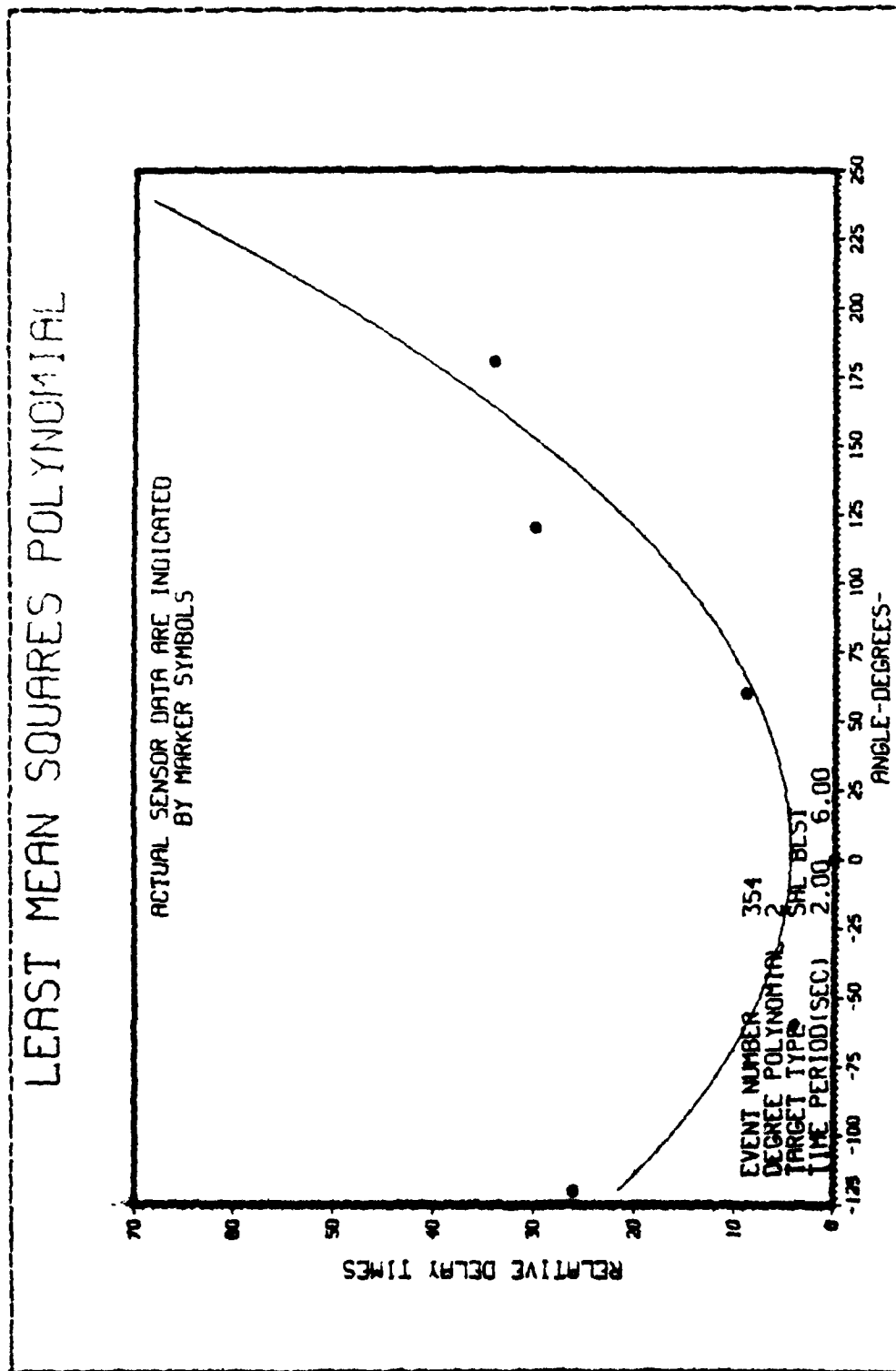


Figure 6.57 LMSP Matched Filter Direction for Event 354 (2 - 6sec)

MULTIPLE TARGET - MATCHED FILTER OUTPUT

EVENT NUMBER	354
TIME PERIOD(SEC)	2.00 6.00
TRACKED VEHICLE	DIRECTION - 332.00
SHELL BLAST	DIRECTION - 3.00
HELICOPTER	DIRECTION - 355.00
SIMULATED TRKD VEHICLE	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED WILD VEHICLE	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED HELICOPTER	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED PERSONNEL	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000

Figure 6.58 LSP Multiple Target Direction Summary Event 354 (2 - 6sec)

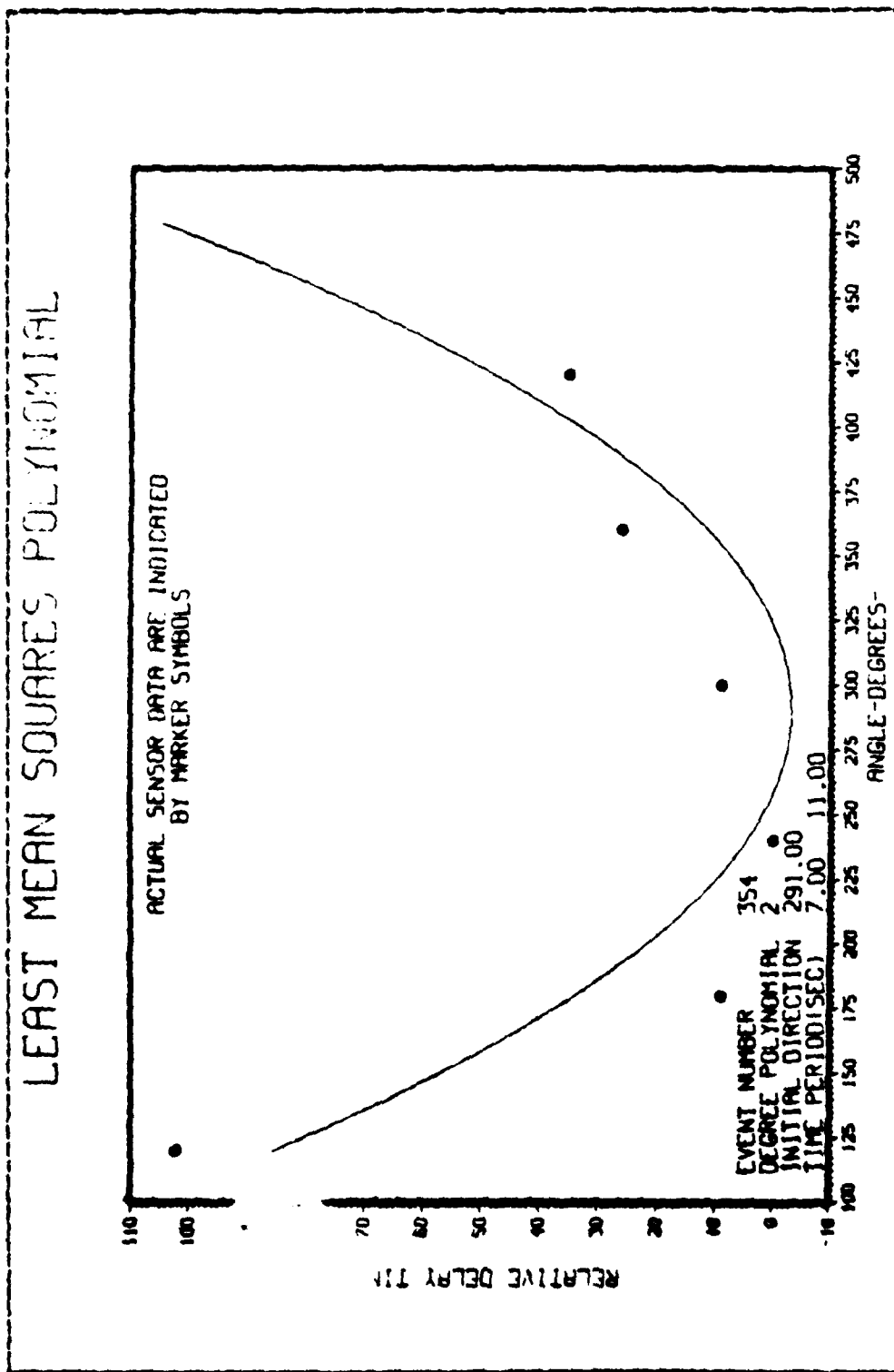


Figure 6.59 LNRP Initial Direction for Event 354 (7 - 11sec)

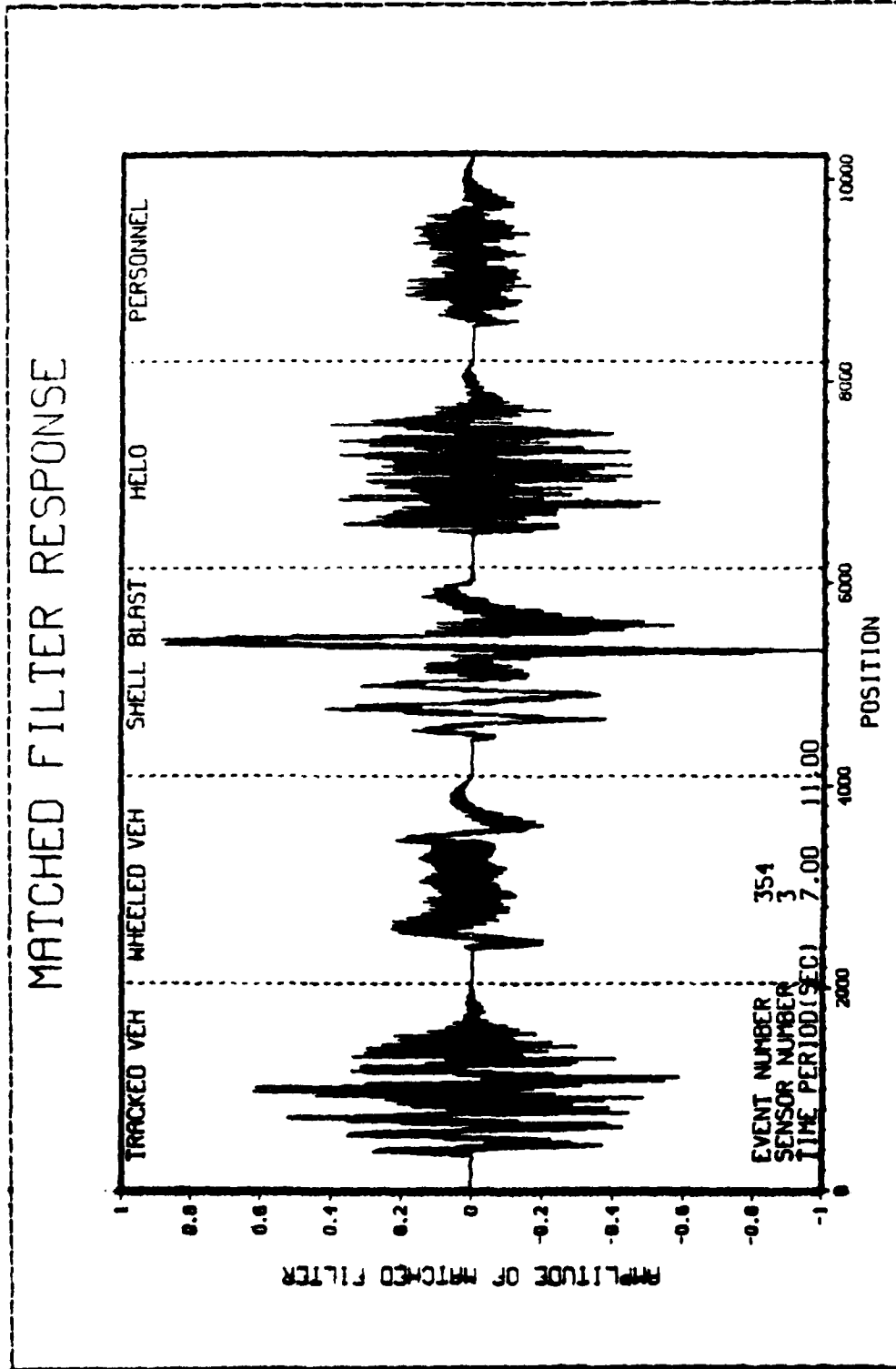


Figure 6.60 Matched Filter Response for Event 354 (7 - 11sec)

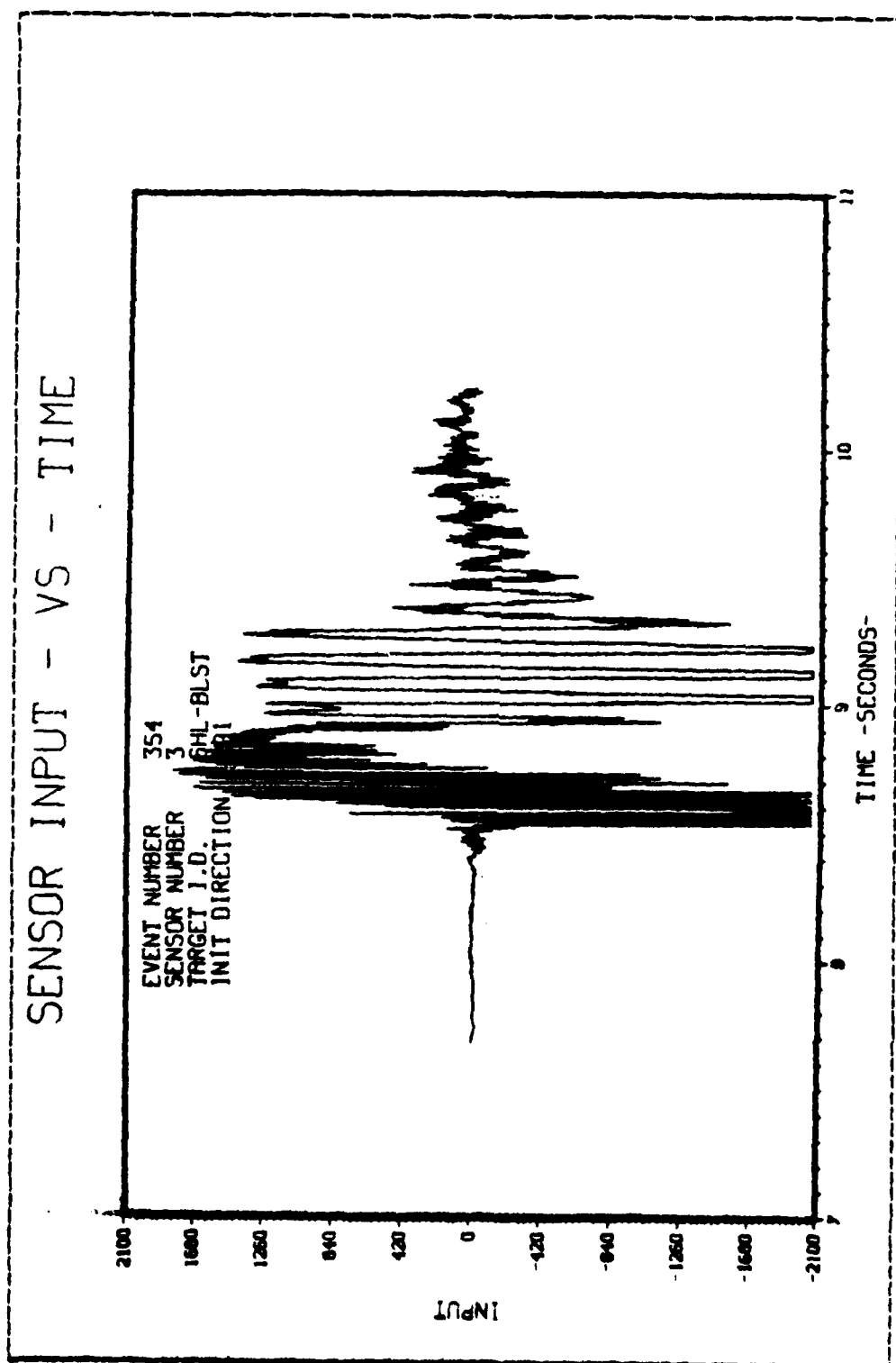


Figure 6.61 Amplitude Response for Event 354 (7 - 11sec)

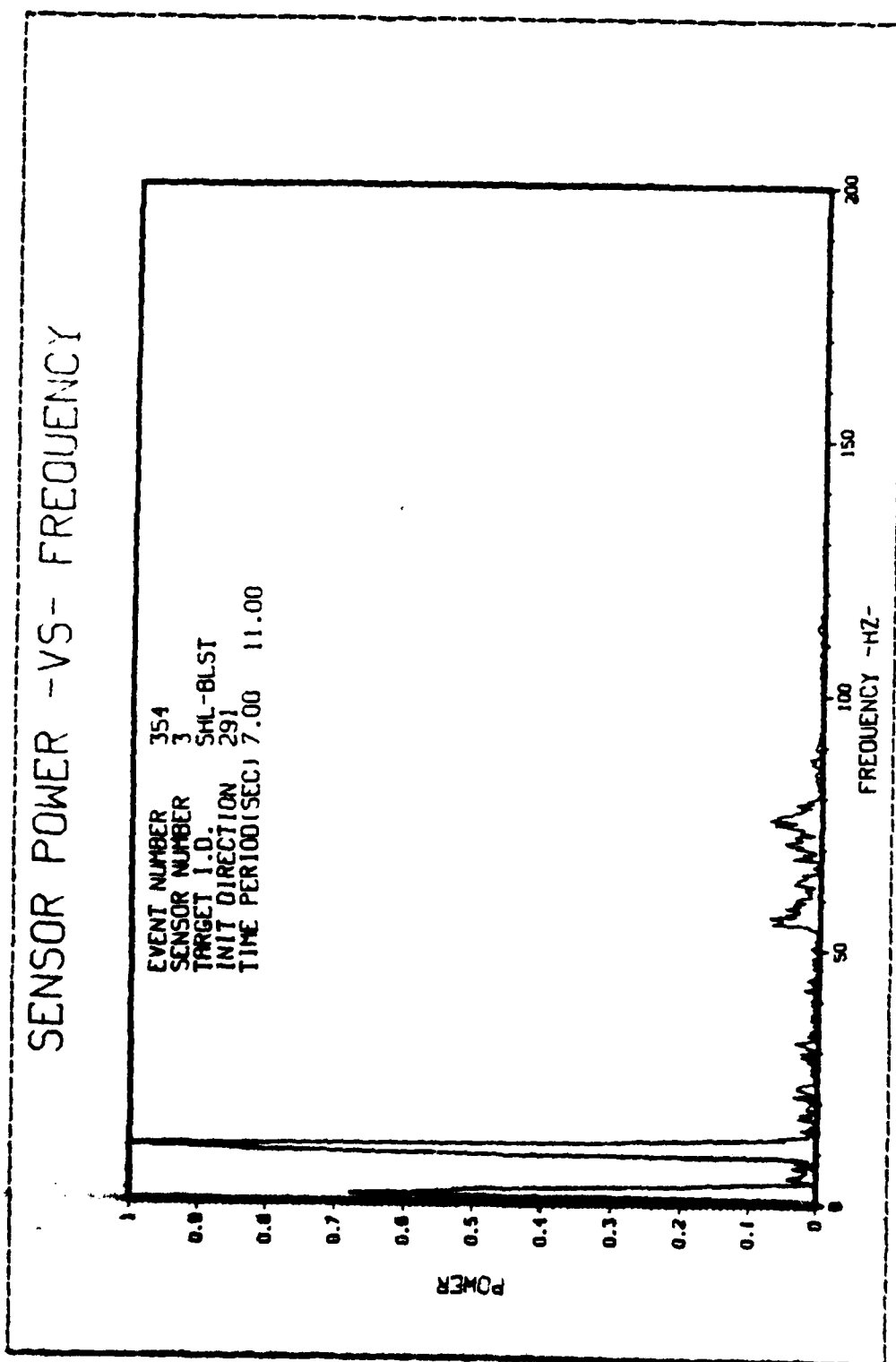


Figure 6.62 Frequency Response for Event 354 (7 - 11sec)

LEAST MEAN SQUARES POLYNOMIAL

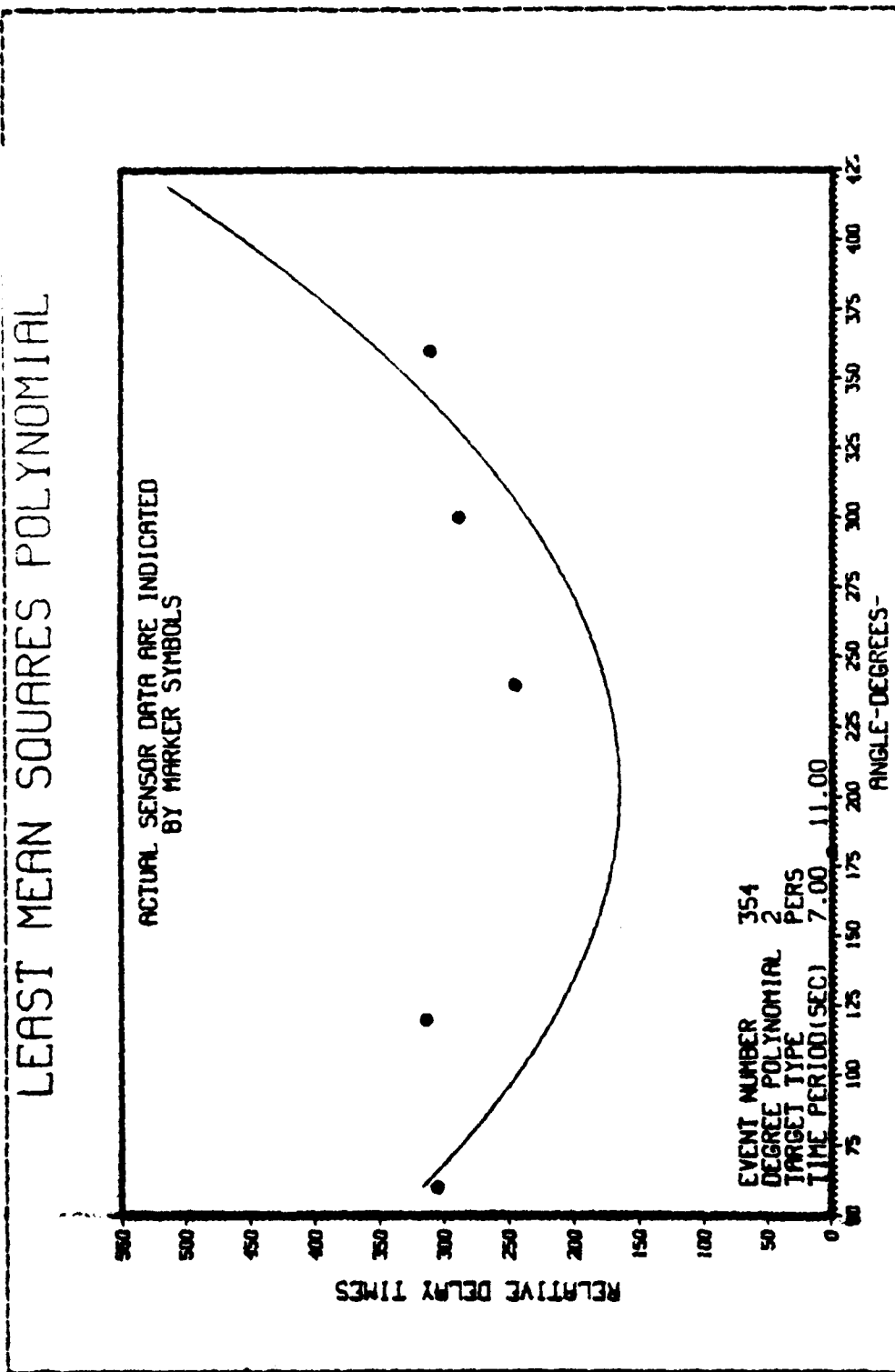


Figure 6.63 LMSP Matched Filter Direction for Event 354 (7 - 11sec)

MULTIPLE TARGET - MATCHED FILTER OUTPUT

EVENT NUMBER	354
TIME PERIOD(SEC)	7.00 11.00
TRACKED VEHICLE	DIRECTION - 291.00
WHEELED VEHICLE	DIRECTION - 291.00
SHELL BLAST	DIRECTION - 291.00
PERSONNEL	DIRECTION - 204.00
SIMULATED TRKD VEHICLE	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED WHEELED VEHICLE	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED HELICOPTER	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000
SIMULATED PERSONNEL	TARGET FREQUENCY 0.00
AMPLITUDE	0.0000
DIRECTION	0.0000

Figure 6.64 LMSP Multiple Target Summary Event 354 (7 - 11sec)

VII. CONCLUSIONS AND RECOMMENDATIONS

The ability of the digital matched filter to detect and correctly identify actual discrete single target types was excellent. The adaptive enhancement of the matched filter scheme was found to sharpen the matched filter responses. The complications of multiple targets and continuous targets proved to be less successful. Lack of high signal to noise ratio sample signals for use as filters reduced the ability of the filters to match the signals. As would be expected, the simulated target identification and direction finding operations met with success for both single and multiple targets.

The simulated data validated the usefulness of the least mean squares curve fitting method for target direction finding. This algorithm was noted to be useful in both the relative peak amplitude response method for recoil/blast targets and the matched filter peak position method for all targets. The highest accuracies were found using second degree polynomials. This was due to the reduced noise sensitivity of lower degree polynomials. Conversely, significant errors were found in the directions determined by the phase difference algorithm. These errors were possibly due to round-off error sensitivity in the software/hardware implementation. Experimental data could not be used to effectively crosscheck this finding since most of the experimental data targets were at zero degrees and the phase difference routine seemed to seek zero degrees.

A significant result was the accuracy of the blast/recoil target direction found using only the peak amplitude responses. Directions could be found using only the relative amplitude peak positions for the array sensors and the

least mean squares curve fitting routine. This result indicated a possible counter-fire application using a greatly simplified system. This finding is felt to be significant since artillery type targets are the highest priority target type. The discrete nature of the blast/recoil seismic signals would also allow for ready separation of even a large number of combined hostile and friendly signal sources. This would allow for observerless adjustment of fire onto hostile targets. Artillery and mortar targets may, in fact, be the only target types detectable at the ranges specified for a long range seismic system.

It is recommended that further study be made in to the possible implementation of the least mean square curve fitting of the peak sensor amplitudes responses in a counter-fire system. Digital matched filters for seismic target identification and direction finding may also prove effective after further experimentation with optimum sample signal filters for the various target types. Additionally, the matched filter response may possibly be enhanced by preprocessing the seismic signals through adaptive noise cancellors.

Acoustic vice seismic matched filtering with directional microphones may be useful in target identification. Once the target has been identified, a matched filter/least mean squares based direction finding scheme could then be attempted.

APPENDIX A
USERS MANUAL

The software developed provided for interactive program operation. However, further information must be provided for an initial system setup and correct program operation. To begin with, the seismic data must be transferred from magnetic tape to the IBM 3033's Mass Storage System (MSS). The data must then be transferred to the DISPLA user's disk for analysis of the program. These data transfers may be accomplished using Job Control Language (JCL) procedures.

The magnetic tape volume must first be scanned to determine the storage format of its files. The JCL procedure TSCAN provides this information. A sample TSCAN job follows:

```
//JLJV1677 JOB (3026,0304),'SMC-1677 JOHNSTON',CLASS=F
// EXEC TSCAN,VOLIN=PARK1,DCBIN='DEN=2',UNITIN='3400-4'
// EXEC TSCAN,VOLIN=PARK2,DCBIN='DEN=2',UNITIN='3400-4'
// EXEC TSCAN,VOLIN=PARK3,DCBIN='DEN=2',UNITIN='3400-4'
// EXEC TSCAN,VOLIN=PARK4,DCBIN='DEN=2',UNITIN='3400-4'
//
```

Once the tape scan is completed, the tape files and comments can be transferred to the MSS. Prior to this transfer however, space in the MSS must be made for these files. The procedure IEFBR14 is used for this purpose. A sample job follows:

```
//JLJTN74A JOB (3026,0304),'SMC1677 JOHNSTON',CLASS=A
//*MAIN ORG=HPCVH1.0131P
// EXEC PGM=IEFBR14
//DD1 DD UNIT=3330V,MSVGP=PUB4B,DISP=(NEW,CATLG),
//      SPACE=(CYL,(4,4,3)),DSN=MSS.S3026.P302
```

```

/*
// EXEC PGM=IEFBR14
//DD1 DD UNIT=3330V,MSVGP=PUB4B,DISP=(NEW,CATLG),
//      SPACE=(CYL,(4,4,3)),DSN=MSS.S3026.P314
/*
// EXEC PGM=IEFBR14
//DD1 DD UNIT=3330V,MSVGP=PUB4B,DISP=(NEW,CATLG),
//      SPACE=(CYL,(4,4,3)),DSN=MSS.S3026.P319
/*
//

```

Each event is proceeded by a comment file for that event. These comment files can be identified from the TSCAN output as a file containing only one record. The files containing nine or eighteen records are the sensor data files for the events. Files of nine records in length are events using a circular array of nine sensors designated as a type A33 array. These sensors are all vertical motion-sensing geophones. The eighteen record files contain six sensor groups of three geophones. A circular array is designated type A31 and a linear array is a type A32 array. The three geophones for each group sense either radial, transverse or vertical motion. A JCL routine, using the procedure IEBGENER, transfers the event comment and sensor data. A sample IEBGENER job follows:

```

//JLJ11677 JOB (3026,0304),'JOHNSTON SMC1677',CLASS=P
//*MAIN ORG=NPGVM1.0131P
/*
/* CP/CMS SUBMIT      IEBGENER JCL
/*
/*      COPY TAPE FILES TO MSS.S3026.P319
/*
// EXEC PGM=IEBGENER
//SYSPRINT DD SYSOUT=A
//SYSUT1 DD UNIT=3400-4,VOL=SER=PARK2,DISP=(,PASS),

```

```

//          LABEL=(1,NL,,IN),
//      DCB=(RECFM=FB,LRECL=64,BLKSIZE=2048,DEN=2,OPTCD=Q)
//SYSUT2 DD  DISP=SHR,DSN=MSS.S3026.COMMENTS(COM319)
//SYSIN DD DUMMY
//*****
//COPY  PROC FILE=,MEM=
//  EXEC PGM=IEBGENER
//SYSPRINT DD SYSOUT=A
//SYSUT1 DD UNIT=3400-4,VOL=SER=PARK2,DISP=(,PASS),
//      LABEL=(&FILE,NL,,IN),DCB=(RECFM=F,BLKSIZE=2048,DEN=2)
//SYSUT2 DD  DISP=SHR,DSN=MSS.S3026.P319(&MEM)
//SYSIN DD DUMMY
//  PEND
/*
//*****
//  EXEC COPY,FILE=342,MEM=SEN1
//  EXEC COPY,FILE=343,MEM=SEN2
//  EXEC COPY,FILE=344,MEM=SEN3
//  EXEC COPY,FILE=345,MEM=SEN4
//  EXEC COPY,FILE=346,MEM=SEN5
//  EXEC COPY,FILE=347,MEM=SEN6
//  EXEC COPY,FILE=348,MEM=SEN7
//  EXEC COPY,FILE=349,MEM=SEN8
//  EXEC COPY,FILE=350,MEM=SEN9
//  EXEC COPY,FILE=351,MEM=SEN10
//  EXEC COPY,FILE=352,MEM=SEN11
//  EXEC COPY,FILE=353,MEM=SEN12
//  EXEC COPY,FILE=354,MEM=SEN13
//  EXEC COPY,FILE=355,MEM=SEN14
//  EXEC COPY,FILE=356,MEM=SEN15
//  EXEC COPY,FILE=357,MEM=SEN16
//  EXEC COPY,FILE=358,MEM=SEN17
//  EXEC COPY,FILE=359,MEM=SEN18
//  EXEC COPY,FILE=360,MEM=SEN19
/*

```

//

Transfer of the desired event data from MSS to the DISSPLA user's disk may now be performed. A batch fortran job with the appropriate FILEDEFS to denote the various geophone's data is submitted. The RSCS/NET feature is used to send the output of this routine to the user's reader. A sample fortran job for nine sensors follows:

```
//JLJ83026 JOB (3026,0304),'JOHNSTON',CLASS=A
```

```
//*MAIN ORG=NPGVM1.0090P
```

```
// EXEC FOFTXCG,REGION.GO=1024K
```

```
//PORT.SYSIN DD *
```

C

```
LOGICAL*1 INFO1(8),INFO2(8),INFO3(8),INFO4(8),INFO5(8)
```

```
LOGICAL*1 INFO6(8),INFO7(8),INFO8(8),INFO9(8)
```

```
INTEGER*2 DATA1(1020),DATA2(1020),DATA3(1020),DATA4(1020)
```

```
INTEGER*2 DATA5(1020),DATA6(1020),DATA7(1020),DATA8(1020)
```

```
INTEGER*2 DATA9(1020),DATB1(4096)
```

C

```
DO 30 J=1,4
```

```
READ(1,100) INFO1,DATA1
```

```
READ(2,100) INFO2,DATA2
```

```
READ(3,100) INFO3,DATA3
```

```
READ(4,100) INFO4,DATA4
```

```
READ(8,100) INFO5,DATA5
```

```
READ(9,100) INFO6,DATA6
```

```
READ(10,100) INFO7,DATA7
```

```
READ(11,100) INFO8,DATA8
```

```
READ(12,100) INFO9,DATA9
```

```
100 FORMAT(8A1,102(10A2))
```

```
DO 10 I = 10,1020,10
```

```
WRITE(6,101) DATA1(I - 9),DATA1(I - 8),DATA1(I - 7),
```

```
1DATA1(I - 6),DATA1(I - 5),DATA1(I - 4),DATA1(I - 3),
```

```
2DATA1(I - 2),DATA1(I - 1),DATA1(I)
```

```

10  CONTINUE
    DO 20 I = 10,1020,10
      WRITE(6,101) DATA2(I - 9),DATA2(I - 8),DATA2(I - 7),
1DATA2(I - 6),DATA2(I - 5),DATA2(I - 4),DATA2(I - 3),
2DATA2(I - 2),DATA2(I - 1),DATA2(I)
20  CONTINUE
    DO 40 I = 10,1020,10
      WRITE(6,101) DATA3(I - 9),DATA3(I - 8),DATA3(I - 7),
1DATA3(I - 6),DATA3(I - 5),DATA3(I - 4),DATA3(I - 3),
2DATA3(I - 2),DATA3(I - 1),DATA3(I)
40  CONTINUE
    DO 50 I = 10,1020,10
      WRITE(6,101) DATA4(I - 9),DATA4(I - 8),DATA4(I - 7),
1DATA4(I - 6),DATA4(I - 5),DATA4(I - 4),DATA4(I - 3),
2DATA4(I - 2),DATA4(I - 1),DATA4(I)
50  CONTINUE
    DO 60 I = 10,1020,10
      WRITE(6,101) DATA5(I - 9),DATA5(I - 8),DATA5(I - 7),
1DATA5(I - 6),DATA5(I - 5),DATA5(I - 4),DATA5(I - 3),
2DATA5(I - 2),DATA5(I - 1),DATA5(I)
60  CONTINUE
    DO 70 I = 10,1020,10
      WRITE(6,101) DATA6(I - 9),DATA6(I - 8),DATA6(I - 7),
1DATA6(I - 6),DATA6(I - 5),DATA6(I - 4),DATA6(I - 3),
2DATA6(I - 2),DATA6(I - 1),DATA6(I)
70  CONTINUE
    DO 80 I = 10,1020,10
      WRITE(6,101) DATA7(I - 9),DATA7(I - 8),DATA7(I - 7),
1DATA7(I - 6),DATA7(I - 5),DATA7(I - 4),DATA7(I - 3),
2DATA7(I - 2),DATA7(I - 1),DATA7(I)
80  CONTINUE
    DO 90 I = 10,1020,10
      WRITE(6,101) DATA8(I - 9),DATA8(I - 8),DATA8(I - 7),
1DATA8(I - 6),DATA8(I - 5),DATA8(I - 4),DATA8(I - 3),
2DATA8(I - 2),DATA8(I - 1),DATA8(I)

```



```

90     CONTINUE
      DO 91 I = 10,1020,10
        WRITE(6,101) DATA9(I - 9),DATA9(I - 8),DATA9(I - 7),
1DATA9(I - 6),DATA9(I - 5),DATA9(I - 4),DATA9(I - 3),
2DATA9(I - 2),DATA9(I - 1),DATA9(I)
91     CONTINUE
101    FORMAT(I6,I6,I6,I6,I6,I6,I6,I6,I6,I6)
30     CONTINUE
      STOP
      END

```

```

/*
//GO.FT01F001 DD DISP=SHR,DSN=MSS.S3026.P383(SEN1)
//GO.FT02F001 DD DISP=SHR,DSN=MSS.S3026.P383(SEN2)
//GO.FT03F001 DD DISP=SHR,DSN=MSS.S3026.P383(SEN3)
//GO.FT04F001 DD DISP=SHR,DSN=MSS.S3026.P383(SEN4)
//GO.FT08F001 DD DISP=SHR,DSN=MSS.S3026.P383(SEN5)
//GO.FT09F001 DD DISP=SHR,DSN=MSS.S3026.P383(SEN6)
//GO.FT10F001 DD DISP=SHR,DSN=MSS.S3026.P383(SEN7)
//GO.FT11F001 DD DISP=SHR,DSN=MSS.S3026.P383(SEN8)
//GO.FT12F001 DD DISP=SHR,DSN=MSS.S3026.P383(SEN9)
//GO.SYSIN DD *
/*
//

```

A sample job for a six sensor group array follows:

```

//JLJ83026 JOB (3026,0304),'JOHNSTON',CLASS=A
//*MAIN ORG=NPGVM1.0090P
// EXEC PORTXCG,REGION.GO=1024K
//PORT.SYSIN DD *

```

C

```

LOGICAL*1 INFO1(8),INFO2(8),INFO3(8),INFO4(8),INFO5(8)
LOGICAL*1 INFO6(8),INFO7(8),INFO8(8),INFO9(8)
INTEGER*2 DATA1(1020),DATA2(1020),DATA3(1020),DATA4(1020)
INTEGER*2 DATA5(1020),DATA6(1020),DATA7(1020),DATA8(1020)
INTEGER*2 DATA9(1020),DATB1(4096)

```

C

```

DO 30 J=1,4

    READ(1,100) INFO1,DATA1
    READ(2,100) INFO2,DATA2
    READ(3,100) INFO3,DATA3
    READ(4,100) INFO4,DATA4
    READ(8,100) INFO5,DATA5
    READ(9,100) INFO6,DATA6
100  FORMAT(8A1,102(10A2))
    DO 10 I = 10,1020,10
        WRITE(6,101) DATA1(I - 9),DATA1(I - 8),DATA1(I - 7),
1DATA1(I - 6),DATA1(I - 5),DATA1(I - 4),DATA1(I - 3),
2DATA1(I - 2),DATA1(I - 1),DATA1(I)
10  CONTINUE
    DO 20 I = 10,1020,10
        WRITE(6,101) DATA2(I - 9),DATA2(I - 8),DATA2(I - 7),
1DATA2(I - 6),DATA2(I - 5),DATA2(I - 4),DATA2(I - 3),
2DATA2(I - 2),DATA2(I - 1),DATA2(I)
20  CONTINUE
    DO 40 I = 10,1020,10
        WRITE(6,101) DATA3(I - 9),DATA3(I - 8),DATA3(I - 7),
1DATA3(I - 6),DATA3(I - 5),DATA3(I - 4),DATA3(I - 3),
2DATA3(I - 2),DATA3(I - 1),DATA3(I)
40  CONTINUE
    DO 50 I = 10,1020,10
        WRITE(6,101) DATA4(I - 9),DATA4(I - 8),DATA4(I - 7),
1DATA4(I - 6),DATA4(I - 5),DATA4(I - 4),DATA4(I - 3),
2DATA4(I - 2),DATA4(I - 1),DATA4(I)
50  CONTINUE
    DO 60 I = 10,1020,10
        WRITE(6,101) DATA5(I - 9),DATA5(I - 8),DATA5(I - 7),
1DATA5(I - 6),DATA5(I - 5),DATA5(I - 4),DATA5(I - 3),
2DATA5(I - 2),DATA5(I - 1),DATA5(I)
60  CONTINUE

```

```

DO 70 I = 10,1020,10
WRITE(6,101) DATA6(I - 9),DATA5(I - 8),DATA6(I - 7),
1DATA6(I - 6),DATA6(I - 5),DATA6(I - 4),DATA6(I - 3),
2DATA6(I - 2),DATA6(I - 1),DATA6(I)
70    CONTINUE
101    FORMAT(I6,I6,I6,I6,I6,I6,I6,I6,I6,I6)
30    CONTINUE
      STOP
      END

```

```

/*
//GO.FT01F001 DD DISP=SHR,DSN=MSS.S3026.P350(SEN3)
//GO.FT02F001 DD DISP=SHR,DSN=MSS.S3026.P350(SEN6)
//GO.FT03F001 DD DISP=SHR,DSN=MSS.S3026.P350(SEN9)
//GO.FT04F001 DD DISP=SHR,DSN=MSS.S3026.P350(SEN12)
//GO.FT08F001 DD DISP=SHR,DSN=MSS.S3026.P350(SEN15)
//GO.FT09F001 DD DISP=SHR,DSN=MSS.S3026.P350(SEN18)
//GO.SYSIN DD *
/*
//

```

Two files will be returned to user's reader. The first file is the listing and diagnostics file and should be purged. The second file should be named SEN DATA. SEN DATA must now be edited. Delete the first seven lines of the file and issue the command LREC 80 to set the proper file record length.

The interactive program can be run with the complete collection of files listed in the exec MATCH. The MATCH EXEC follows:

```

STRACE OFF
PORTGI MPILTER
GLOBAL TITLIB PORTMOD2 MOD2EEH IMSLSP NONIMSL
FILEDEF 10 TERMINAL
FILEDEF 05 DISK SEN DATA (PERM)
FILEDEF 07 DISK COM DATA (PERM)

```

FI 4 DISK FILTER DATA (RECFM VS PERM

FILEDEF 18 DISK DISSPLA METAFILE T4 (RECFM VBS LRECL

19065 BLOCK 19069

EXEC DISSPLA MFILTER

Entry parameters, such as the number of sensors in the array and the sampling rate, can be found in the NOSC data log for the event under study. All other interactive entries are user selected options or are self explanatory. [Ref. 12]

APPENDIX B
SAMPLE INTERACTIVE PROGRAM SESSION

match
G1 COMPILER ENTERED
SOURCE ANALYZED
PROGRAM NAME = MAIN
* NO DIAGNOSTICS GENERATED
SOURCE ANALYZED
PROGRAM NAME = ANGLE
* NO DIAGNOSTICS GENERATED
SOURCE ANALYZED
PROGRAM NAME = TIMOUT
* NO DIAGNOSTICS GENERATED
SOURCE ANALYZED
PROGRAM NAME = FREQOT
* NO DIAGNOSTICS GENERATED
SOURCE ANALYZED
PROGRAM NAME = MATCH
* NO DIAGNOSTICS GENERATED
SOURCE ANALYZED
PROGRAM NAME = MYDATA
* NO DIAGNOSTICS GENERATED
SOURCE ANALYZED
PROGRAM NAME = MAXMIN
* NO DIAGNOSTICS GENERATED
SOURCE ANALYZED
PROGRAM NAME = SPCTRM
* NO DIAGNOSTICS GENERATED
SOURCE ANALYZED
PROGRAM NAME = MULTI
* NO DIAGNOSTICS GENERATED
SOURCE ANALYZED

PROGRAM NAME = MLTPLT
* NO DIAGNOSTICS GENERATED
SOURCE ANALYZED
PROGRAM NAME = RMS
* NO DIAGNOSTICS GENERATED
SOURCE ANALYZED
PROGRAM NAME = AVG
* NO DIAGNOSTICS GENERATED
SOURCE ANALYZED
PROGRAM NAME = SIMULT
* NO DIAGNOSTICS GENERATED
SOURCE ANALYZED
PROGRAM NAME = LMS
* NO DIAGNOSTICS GENERATED
SOURCE ANALYZED
PROGRAM NAME = PLT
* NO DIAGNOSTICS GENERATED
SOURCE ANALYZED
PROGRAM NAME = SOLV
* NO DIAGNOSTICS GENERATED
 STATISTICS NO DIAGNOSTICS THIS STEP
DISK 'T' NOT ACCESSED.
B (126) R/O
C (127) R/O
E (128) R/O

... Your Fortran program is now being loaded ...

... execution will soon follow ...

EXECUTION BEGINS...

ENTER EVENT RECORDING NUMBER-13-
383

EVENT NUMBER 383

383. A33. 10 SEPT61 5KM EOD SHOT. SET B. NO DELAY. C141 ON FINAL

AT END OF TAPE."

ENTER SAMPLE RATE IN HERTZ-REAL-
120.

ENTER LOW LOOK ANGLE IN DEGREES-I3-
-100

ENTER HIGH LOOK ANGLE IN DEGREES-I3-
200

ENTER MATCH FILTER THRESHOLD. RANGE OF 0. TO 1.0
.9

ENTER PLOT SCALING FACTOR-REAL-
.85

ENTER NUMBER OF SENSORS IN RING 6 OR 9 ONLY-I1-
9

ENTER SENSOR NUMBER FOR DISPLAY-I1-
4

ENTER NOISE THRESHOLD LEVEL -REAL-
1000.

ENTER DATA WINDOW SIZE FOR DIRECTION FINDING-I4-
0400

FOR COMPRS OUTPUT ENTER -1-. FOR TEK618 ENTER -2-
1

TO CREATE SIMULATED TARGETS ENTER -1- ELSE -2-
1

ENTER THE FOUR SIMULATION FREQUENCIES-REAL-

ENTER FREQUENCY
0

ENTER FREQUENCY
0

ENTER FREQUENCY

0

ENTER FREQUENCY

120.

ENTER AMPLITUDES FOR EACH FREQUENCY-REAL-

ENTER AMPLITUDE

0

ENTER AMPLITUDE

0

ENTER AMPLITUDE

0

ENTER AMPLITUDE

4000.

ENTER TARGET ANGLE FOR EACH FREQUENCY-13-

SIX SENSORS ALLOWABLE ANGLES; 0, 60, 120, 180, 240, 300

FOR NINE SENSORS; 0, 40, 80, 120, 160, 200, 240, 280, 320

ENTER ANGLE

0

ENTER ANGLE

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ENTER ANGLE

0

ENTER ANGLE

120

TO MODIFY AMPLITUDE OF SIGNAL ABOVE NOISE THRESHOLD

ENTER - 1 -, ELSE ENTER - 2 -

2

ENTER DEGREE OF POLYNOMIAL DESIRED -11-

FOR SIX SENSORS ENTER 2 - 4, FOR NINE ENTER 2 - 7

4

>> USING A PRE-ALLOCATED DATASET FOR UNIT FT17F001.

>> USING A PRE-ALLOCATED DATASET FOR UNIT FT18F001.

CATALOG AS TARGET? IF YES TYPE - 1 -, ELSE - 2 -

2

FOR MATCH FILTER DIREC FINDING ENTER - 1 - ELSE -2-

1

FOR PHASE DELAY METHOD ENTER -1, FOR LMS ENTER -2

2

ENTER DEGREE OF POLYNOMIAL DESIRED -11-

FOR SIX SENSORS ENTER 2 - 4, FOR NINE ENTER 2 - 7

4

TO VIEW OTHER SENSORS ENTER - 1 -

1

ENTER - 2 - TO CONTINUE TO NEXT TIME FRAME

2

PROGRAM LISTING

[illegible]

[illegible]

INTERACTIVE INITIALIZATION OF PROGRAM

ENTER EVENT RECORDING NUMBER-13-01

Electrical Testing

[illegible]

CD 777 = 1.60 LEAD. COMPEN
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[illegible][illegible]

FORM 1-08, 16A4)

ENTER SAMPLE RATE IN HERTZ-REAL-1)

1000

[illegible]

NOT RECORDED

1997

FILE NAME: C:\WINDOWS\TEMP\1077
FORMAT: 11-17-97
REASON: ENTER FILTER THRESHOLD. RANGE:

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[illegible]

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160 1DATA7(1) = 11, DATA7(1) - 41, DATA7(1) - 31, DATA7(1) - 21,
    2DATA7(1) = 11, DATA7(1)
    DO 10 I=1, 10
    1DATA8(1) = 5, 11, DATA8(1) - 81, DATA8(1) - 71, DATA8(1) - 61,
    2DATA8(1) = 11, DATA8(1)
    DO 10 I=1, 10
    1DATA9(1) = 5, 11, DATA9(1) - 81, DATA9(1) - 71, DATA9(1) - 61,
    2DATA9(1) = 11, DATA9(1)
    CONTINUE
180 CONVERT THE INTEGER DATA TO REAL AND COMPLEX VALUES
    CALL MYLATA1(1, TIME, DATA1, N, RA1, A1)
    CALL MYLATA2(1, TIME, DATA2, N, RA2, A2)
    CALL MYLATA3(1, TIME, DATA3, N, RA3, A3)
    CALL MYLATA4(1, TIME, DATA4, N, RA4, A4)
    CALL MYLATA5(1, TIME, DATA5, N, RA5, A5)
    CALL MYLATA6(1, TIME, DATA6, N, RA6, A6)
    CALL MYLATA7(1, TIME, DATA7, N, RA7, A7)
    CALL MYLATA8(1, TIME, DATA8, N, RA8, A8)
    CALL MYLATA9(1, TIME, DATA9, N, RA9, A9)
    IF (NUNSEN) GO TO 86
    IF SIMULATION IS REQUESTED, CALL THE SIMULATION ROUTINE
    AF(1) = 1, EQ, 11, CALL SIMULT(RA1, RA2, RA3, RA4, RA5, RA6,
    1RA7, RA8, RA9, A1, A2, A3, A4, A5, A6, A7, A8, A9, V, A, ND, INOISE, TIME, JL, P,
    2REDUCE, RUMSEN)
    C CALCULATE THE FFT AND SPECTRAL FREQUENCY POWER FOR EACH SENSORS DATA
    CALL FFT2C(A1, IMX1)
    CALL SPECTRA1(TIME, RA1, FREQ, A1, N, PWR1, PDB1, PH1)
    CALL FFT2C(A2, IMX2)
    CALL SPECTRA2(TIME, RA2, FREQ, A2, N, PWR2, PDB2, PH2)
    CALL FFT2C(A3, IMX3)
    CALL SPECTRA3(TIME, RA3, FREQ, A3, N, PWR3, PDE3, PH3)
    CALL FFT2C(A4, IMX4)
    CALL SPECTRA4(TIME, RA4, FREQ, A4, N, PWR4, PDE4, PH4)
    CALL FFT2C(A5, IMX5)

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113      CALL MATHMIN(RA2,1024,K,RAGMAX,L,RIN)
        IF (ABSC(RAGMAX)11.GT.(11ACISE)) GO TO 113
        NR1=1
        GO TO 112
        CALL MATHMIN(RA2,NTYP,SCALE,ICRR,DIRECT,THRES,BCAT,TESTNO,JL,C,N,
        1TS,TIME,RAS)
        CALL FREQC1(FREC,PWR2,DIRECT,SCALE,ICRR,NTYP,TESTNO,TS,TF,C)
        CALL FREQC1(FREC,PWR2,DIRECT,SCALE,ICRR,NTYP,TESTNO,TS,TF,C)
        C
        C
        C 445
        IF (110OFF) NE.(13) GO TO 446
        CALL MATHMIN(RA3,1024,K,RAGMAX,L,RIN)
        IF (ABSC(RAGMAX)11.GT.(11NOISE)) GO TO 114
        NR2=1
        GO TO 112
        CALL MATHMIN(RA3,NTYP,SCALE,ICRR,DIRECT,THRES,BCAT,TESTNO,JL,C,N,
        1TS,TIME,RAS)
        CALL FREQC1(FREC,PWR3,DIRECT,SCALE,ICRR,NTYP,TESTNO,TS,TF,C)
        CALL FREQC1(FREC,PWR3,DIRECT,SCALE,ICRR,NTYP,TESTNO,TS,TF,C)
        C
        C
        C 446
        IF (110OFF) NE.(14) GO TO 447
        CALL MATHMIN(RA4,1024,K,RAGMAX,L,RIN)
        IF (ABSC(RAGMAX)11.GT.(11ACISE)) GO TO 115
        NR3=1
        GO TO 112
        CALL MATHMIN(RA4,NTYP,SCALE,ICRR,DIRECT,THRES,BCAT,TESTNO,JL,C,N,
        1TS,TIME,RAS)
        CALL FREQC1(FREC,PWR4,DIRECT,SCALE,ICRR,NTYP,TESTNO,TS,TF,C)
        CALL FREQC1(FREC,PWR4,DIRECT,SCALE,ICRR,NTYP,TESTNO,TS,TF,C)
        C
        C
        C 447
        IF (110OFF) NE.(15) GO TO 448
        CALL MATHMIN(RA5,1024,K,RAGMAX,L,RIN)
        IF (ABSC(RAGMAX)11.GT.(11NOISE)) GO TO 116
        NR4=1
        GO TO 112
        CALL MATHMIN(RA5,NTYP,SCALE,ICRR,DIRECT,THRES,BCAT,TESTNO,JL,C,N,
        1TS,TIME,RAS)
        CALL FREQC1(FREC,PWR5,DIRECT,SCALE,ICRR,NTYP,TESTNO,TS,TF,C)
        CALL FREQC1(FREC,PWR5,DIRECT,SCALE,ICRR,NTYP,TESTNO,TS,TF,C)
        C
        C
        C 448
        IF (110OFF) NE.(16) GO TO 449

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[illegible]

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837 WRITE(6,637)
      FORN1(1,C),FOR MATCH FILTER DIREC FNCING ENTER - 1 - ELSE -2-
      REAC(1,C,15) MNV
      IF((MNV).EQ.(1))CALL MULTI(RA1,RA2,RA3,RA4,RA5,RA6,RA7,RA8,
1RA9,NTYP,SCALE,1DOR,THRES,BCAT,TESTNO,JL,C,IS,IF,V,A,ND,
2NW,NUMSEN,TIME,1TRGET,DIRCT,DIR,DIRC)
      IF((MNV).NE.(1))GC TC 454
      WRITE(6,44)
      FORMAT(10,0),FOR PHASE DELAY METHOD ENTER -1,FOR LMS ENTER -2,
44 READ(1,C,35)NGL
      FORMAT(11)
593 DO 666 I21 = 1,5
      DO 666 I21 = 1,5
      NTAR(I21) = C
      CONTINUE
      IF((NGL).EQ.(1))GO TO 101
      DO 678 IW = 1,5
      NTAR(IW) = ITRGET(1,IW)
      NTAR(2) = ITRGET(2,IW)
      NTAR(3) = ITRGET(3,IW)
      NTAR(4) = ITRGET(4,IW)
      NTAR(5) = ITRGET(5,IW)
      NTAR(6) = ITRGET(6,IW)
      IF(NUMSEN.EQ.6)GC TC 646
      IF((NGL).EQ.(1))GO TO 101
      DO 678 IW = 1,5
      NTAR(IW) = ITRGET(7,IW)
      NTAR(6) = ITRGET(8,IW)
      NTAR(5) = ITRGET(5,IW)
      NG = IW
      IF((NTAR(1)).EQ.(C))GC TC 678
      CALL LMS (NTAR,DIRC,NUMSEN,C,SCALE,TESTNO,TIME,NG,Y,
1VX,X,TT,INDEX,NCR,TRIX,B,WKAREA)
      CALL PL1(C,SCALE,TESTNO,NUMSEN,TIME,Y,VX,X,TT,INDEX,NOR,DIRC,
1NG)
      DIR(NG) = DIRC
      CCNTINUE
      GO TO 6C1
      C
      C
      C
101 PHASE DELAY METHOD
      DO 60 J3 = 1,5
      TC = C
      DC 7C K3 = 1,NUMSEN
      TCEN(K3) = FLOAT(ITARGET(K3,J3))
      TC = FLCAT(ITARGET(K3,J3)) + 7C
      CCNTINUE
      TC = TC/FLOAT(NUMSEN)
      CE = C.
      DO 84C N = 1,NUMSEN
      DO 84C N = 1,NUMSEN

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MF104610
MF104620
MF104630
MF104640
MF104650
MF104660
MF104670
MF104680
MF104690
MF104700
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MF104990
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MF105090
MF105100
MF105110
MF105120
MF105130
MF105140
MF105150
MF105160
MF105170
MF105180
MF105190
MF105200
MF105210
MF105220
MF105230
MF105240
MF105250
MF105260
MF105270
MF105280

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DO SC M = 1, NUMSEN
  IF ((M).LE.(N)) GO TO 90
  RE = ((TC - TCEN(M)) * X(N) - (TC - TCEN(N)) * X(M))
  AA = ((TC - TCEN(N)) * Y(M) - (TC - TCEN(P)) * Y(N))
  IF ((AA).EQ.(0)) GO TO 90
  IF ((BB).LT.(.000001)) GO TO 90
  DIVIDE = DIVIDE + 1.
  CE = ATAN(BB/AA) + QB
  CONTINUE
90  CONTINUE
840 NUM = NUMSEN - 1
    QB = (CE / (DIVIDE)) * (180.0 / 3.141593)
    DIR(J3) = CB
    CONTINUE
60  CALL MULTPLT(C, SCALE, TESTNO, TS, IF, DIR, V, A, ND, ITRGET)
C    SELECT TC EVALUATE OTHER SENSORS IN THIS TIME PERIOD OR CONTINUE
C    TC THE NEXT TIME PERIOD
454 WRITE(C, 51)
    WRITE(C, 52)
51  FORMAT(10.0, 'TO VIEW OTHER SENSORS ENTER - 1 - ')
52  FORMAT(10.0, 'ENTER - 2 - TO CONTINUE TO NEXT TIME FRAME.')
    READ(C, 53)
551 FORMAT(11)
    IF ((NS).NE.(1)) GO TO 10
    WRITE(C, 57)
57  FORMAT(10.0, 'ENTER NEW SENSOR NUMBER FOR DISPLAY-11-')
    READ(C, 55) ICRR
    GO TO 55
    CCNT INLE
    CALL DCNEPL
    STOP
    END
C    SUBROUTINE ANGLE COMPUTES A ROUGH DIRECTION TO THE PRIMARY TARGET
C    FOR USE IN DETERMINING WHETHER THE TARGET IS IN
C    THE SECTOR OF INTEREST
C    TIME DOMAIN ANALYSIS IS USED.
C    SUBROUTINE ANGLE (DIRC, R1, R2, R3, R4, R5, R6, R7, R8, R9, NUMSEN, TESTING,
1  TIME, C, SCALE, NTAR)
C    INTEGER M(5), TESTING, C, NTAR(9), NUMSEN, C
C    REAL R1(1024), R2(1024), R3(1024), R4(1024), R5(1024), R6(1024),
1R7(1024), R8(1024), R5(1024), TIME(1024), CIR(5), DINC
C    DO 12 I = 1, 9
C      NTAR(I) = 0

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CONTINUE
CALL MAXMIN(R1,1024,M(1),RMAX,L,RMIN)
CALL MAXMIN(R2,1024,M(2),RMAX,L,RMIN)
CALL MAXMIN(R3,1024,M(3),RMAX,L,RMIN)
CALL MAXMIN(R4,1024,M(4),RMAX,L,RMIN)
CALL MAXMIN(R5,1024,M(5),RMAX,L,RMIN)
CALL MAXMIN(R6,1024,M(6),RMAX,L,RMIN)
IF(1,NUMSEN,EQ,16) GO TO 10
CALL MAXMIN(R7,1024,M(7),RMAX,L,RMIN)
CALL MAXMIN(R8,1024,M(8),RMAX,L,RMIN)
CALL MAXMIN(R9,1024,M(9),RMAX,L,RMIN)
DO 11 12 = 1,NUMSEN
  11 12 = 1,M(12)
CONTINUE
RETURN
END

SUBROUTINE TIMOUT PLCTS AMPLITUDE VERSUS
ENSCR
1C)
RE AL RZ(1024), TIME(1024)
INTEGER NTYP(6),C,NV,DIRECT,TESTNO
IF(1,C,1) CALL COMPRS
IF(1,C,NE,1) CALL TEK618
CALL PAGER(11,0,E,5)
CALL NCRCR(SCALE)
CALL BLCK2C(9,C,6,C)
CALL FRAME
CALL XNAME('TIME-SECONDS-',14)
CALL YNAME('AMPLI',5)
CALL HESAG('SENSOR NUMBER',14,2,0,5,7)
CALL HESAG('EVENT NUMBER',12,2,0,5,7)
CALL INTNOG(DRR,4,C,5,5)
CALL INTAG(TESTNO,4,0,5,7)
CALL HESAG('TARGET I.D.',11,2,0,5,3)
XPCS = 80
IF(1,NTYP(1),NE,0) XPCS = XPCS + 1.
IF(1,NTYP(1),NE,0) XPCS = XPCS + 1.
IF(1,NTYP(2),NE,0) XPCS = XPCS + 1.
IF(1,NTYP(2),NE,0) XPCS = XPCS + 1.
IF(1,NTYP(3),NE,0) XPCS = XPCS + 1.
IF(1,NTYP(3),NE,0) XPCS = XPCS + 1.

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1C) REAL PWR(1024), FREQ(1024)
      INT NTYPE(6), C, DIRECT, IESTNG
      IF ((C).EQ.(1)) CALL CLMPAS
      IF ((C).NE.(1)) CALL TEK618
      CALL PAGE(11.0, 8.5)
      CALL NCERDR
      CALL BLCHUP(SCALE)
      CALL AFREQ(9.0, 6.0)
      CALL XNAME('FREQUENCY -HZ-', 14)
      CALL YNAME('POWER', 5)
      CALL HEADING('SENSOR NUMBER', 13, 2.0, 5.7)
      CALL MESSAGE('EVENT NUMBER', 14, 2.0, 5.5)
      CALL INTC('TESTING', 4.0, 5.7)
      CALL MESSAGE('SENSOR NUMBER', 14, 2.0, 5.5)
      CALL INTC('CURR', 4.0, 5.5)
      CALL MESSAGE('TARGET', 1.0, 11, 2.0, 5.3)
      XPCS = 2.80
      IF ((NTYPE(1)).NE.(0)) XPCS = XPCS + 1.2
      IF ((NTYPE(1)).NE.(0))
      1CALL MESSAGE('TRKD-VEH', 8, XPCS, 5.3)
      IF ((NTYPE(2)).NE.(0)) XPCS = XPCS + 1.2
      IF ((NTYPE(2)).NE.(0))
      1CALL MESSAGE('WHLD-VEH', 8, XPCS, 5.3)
      IF ((NTYPE(3)).NE.(0)) XPCS = XPCS + 1.2
      IF ((NTYPE(3)).NE.(0))
      1CALL MESSAGE('SHL-BLST', 8, XPCS, 5.3)
      IF ((NTYPE(4)).NE.(0)) XPCS = XPCS + 1.2
      IF ((NTYPE(4)).NE.(0))
      1CALL MESSAGE('HELICFTR', 8, XPCS, 5.3)
      IF ((NTYPE(5)).NE.(0)) XPCS = XPCS + 1.2
      IF ((NTYPE(5)).NE.(0))
      1CALL MESSAGE('PERSONNEL', 9, XPCS, 5.3)
      CALL MESSAGE('INIT DIRECTION', 14, 2.0, 5.1)
      CALL INTC('DIRECT', 4.0, 5.1)
      CALL MESSAGE('TIME PERIOD(SEC)', 16, 2.0, 4.9)
      CALL REALNCTS(2.3, 9, 4.9)
      CALL REALNCTF(2.4, 7, 4.9)
      XMAX = 1.0
      DO 39 I = 2, 500, 20
      IF ((XMAX).NE.(1.0)) GO TO 39
      IF ((I).GE.(IFIX(FREQ(512)))) XMAX = FLUAT(I)
      CONTINUE
      XSTEP = XMAX/4.00
      CALL MAXMIN(PWR, 1024, KP, PWRMAX, LP, PWRMIN)
      NS = IFIX(XSTEP)
      IF ((XSTEP).GT.(20.0)) NS = IFIX(FLCAT(NS)/10.0)

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MF106250
 MF106260
 MF106270
 MF106280
 MF106290
 MF106300
 MF106310
 MF106320
 MF106330
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 MF106370
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MF107060
MF107070
MF107080
MF107090
MF107100
MF107110
MF107120
MF107130
MF107140
MF107150
MF107160
MF107170
MF107180
MF107190
MF107200

CALL IATAXS (0.)
CALL YAXANG(NS)
CALL YTICKS(10)
CALL GRAF(0,XSTEP,XMAX,0.,.1,PWRMAX)
CALL CLFVE(FREQ,PWR,512,0.)
CALL RESET('ALL')
CALL RECDPL(0)
CONTINUE
RETURN
END

38
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C
C
C
C
C
SUBROUTINE MATCH PERFORMS THE MATCHED FILTERING OF THE SELECTED
SENSORS DATA. TARGET CLASSIFICATIONS FOUND ARE RETURNED, AS ARE
THEIR RESPECTIVE PEAK SIGNAL DETECTION POSITIONS.

C
C
C
C
SUBROUTINE MATCHP(RMAG,NTYP,SCALE,ICFR,DIRECT,THRES,ICAT,TESTNO,
JL,C,NT,TS,TF,TIME)
REAL RPAC(NT),CAT(1024),UNIT(11264),PLATE(2048),SCAT(1024),
1RMAT(11264),THRES,XRAY4(2),YRAY(2),TIME(1024),
2XRAY1(2),XRAY2(2),XRAY3(2),ICAT(512)
INTEGER NTYP(6),C,DIRECT,TESTNO

C

DATA XRAY1/2048,.2C48./
DATA XRAY2/4096,.4C56./
DATA XRAY3/6144,.6144./
DATA XRAY4/8192,.8192./
DATA YRAY/-1.0,1.0/
DO 33 J = 1,1024
PLATE(J) = 0.
PLATE(J + 1024) = 0.
SCAT(J) = 0.
CAT(J) = 0.

CONTINUE
NH = NT/2
N2 = NT*4
N4 = NT*4
N6 = NT*6
N8 = NT*8
N10 = NT*10
N11 = NT*11
DO 600 PM = 1,N11
RMAT(PM) = 0.0
UNIT(PM) = MM
CONTINUE
DO 201 K = N2,N10,N2
NM = K - NT
DO 202 KK = 1,NT

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202 RMAT(NM + KK) = RMAG(KK)
201 CONTINUE
C CONTINUE
C NTYP(1) = 0
C NTYP(2) = 0
C NTYP(3) = 0
C NTYP(4) = 0
C NTYP(5) = 0
C
C NEW FILTER SAMPLE SIGNALS ARE LOADED AFTER EACH SUCCESSIVE 2048
C ELEMENTS OF THE WORKING ARRAY 'RPAT' ARE CONVOLVED
C
111 DO 400 I = 1, N10
C IF(((1).EQ.(1)).OR.((1).EQ.(N2)).OR.((1).EQ.(N4)).OR.
C 1((1).EQ.(N6)).OR.((1).EQ.(N8)))
C 2GO TO 55
C GO TO 55
C IF(((1).EQ.(1)).OR.((1).EQ.(N2)).OR.((1).EQ.(N4)).OR.
C 98 IF(((1).EQ.(N6)).OR.((1).EQ.(N8))) MARK = 1025
C IF(((1).EQ.(N2)).OR.((1).EQ.(N4)).OR.((1).EQ.(N6)).OR.
C IF(((1).EQ.(N8))) MARK = 3073
C IF(((1).EQ.(N6)).OR.((1).EQ.(N8))) MARK = 4097
C IF(((1).EQ.(N8))) MARK = 5121
C
C FILTER DATA REVERSED
C
C DO 777 LL = 1, NT
C CAT(LL) = TCAT(-LL + MARK)
C CONTINUE
777
C
C EXECUTION TIME MAY BE REDUCED BY SELECTING SMALLER WINDOWS OR
C SEGMENTS AROUND THE PEAK AMPLITUDES OF THE FILTER AND SEISMIC DATA
C
C CALL MAXMINCAT, 1024, LL, CATM, KK, CATMIN)
C DO 10 11 = 1, NT
C IF((LL - NH).LE.(0)) LL = NH
C IF((LL).GT.(1024 - NH)) LL = 1024 - NH
C SCAT(11) = CAT(11 - NH + LL)
C CONTINUE
10 CALL RMS(SCAT, NT, CATMAX)
99 CALL RMS(RMAG, NT, RAGMAX)
C CALL AVG(SCAT, NT, ACAT)
C RMAT(11) = 0
C
C REMOVAL OF CC EIAS AND NORMALIZATION OF THE FILTER AND SEISMIC DATA
C
C CC 2CC KN = 1, NT
C IF(((CATMAX).EQ.(0)).OR.((RAGMAX).EQ.(0))) GO TO 200
C

```

```

MF107210
MF107220
MF107230
MF107240
MF107250
MF107260
MF107270
MF107280
MF107290
MF107300
MF107310
MF107320
MF107330
MF107340
MF107350
MF107360
MF107370
MF107380
MF107390
MF107400
MF107410
MF107420
MF107430
MF107440
MF107450
MF107460
MF107470
MF107480
MF107490
MF107500
MF107510
MF107520
MF107530
MF107540
MF107550
MF107560
MF107570
MF107580
MF107590
MF107600
MF107610
MF107620
MF107630
MF107640
MF107650
MF107660
MF107670
MF107680

```

MF107690
MF107700
MF107710
MF107720
MF107730
MF107740
MF107750
MF107760
MF107770
MF107780
MF107790
MF107800
MF107810
MF107820
MF107830
MF107840
MF107850
MF107860
MF107870
MF107880
MF107890
MF107900
MF107910
MF107920
MF107930
MF107940
MF107950
MF107960
MF107970
MF107980
MF107990
MF108000
MF108010
MF108020
MF108030
MF108040
MF108050
MF108060
MF108070
MF108080
MF108090
MF108100
MF108110
MF108120
MF108130
MF108140
MF108150
MF108160

```

C C DISCRETE CONVOLUTION PERFORMED
      RMAT(I) = ((RMAT(I) + KN1)/(ABS(RAGMAX))) *
      1((SCAT(KN) - ACAT)/(CATMAX)) + RMAT(I)
      CONTINUE
      CONTINUE
      CALL MAXMIN(RMAT,N11,L2,KATMAX,K2,RATMIN)
      DO 390 NN = 1,N11
      IF ((RATMAX).EQ.(0)) GO TO 390
      RMAT(NN) = RMAT(NN)/(ABS(RATMAX))
      CONTINUE
C C CLASSIFICATION OF TARGETS BY TARGET TYPE
      DO 302 J1 = N2,N10,N2
      DO 301 I1 = 1,N2
      PLATE(I1) = RPAT(J1 - N2 + 11)
      CONTINUE
      CALL MAXMIN(PLATE,N2,L2,PMAX,LS,PMIN)
      IF ((ABS(PMAX)).GE.(THRES)) NTYP(J1/N2) = LP
      CONTINUE
C C MATCHED FILTER OUTPUT PLCTING
      IF ((IOPF).EQ.(0)) GC TC 391
      IF ((C).EQ.(1)) CALL CCMPS
      IF ((C).NE.(1)) CALL TEK618
      CALL PAGE(11.0,E.5)
      CALL NCERDR
      CALL BLCWUP(SCALE)
      CALL AREA2D(9,C.6,C)
      CALL FRAME
      CALL XNAME('POSITION',8)
      CALL YNAME('AMPLITUDE',8)
      CALL HEADIN('MATCHED FILTER RESPONSE',23.2,C,1)
      CALL MESSAG('TRACKED VEH',11.25,5.8)
      CALL MESSAG('WHEELED VEH',11.2,0.5,8)
      CALL MESSAG('SHELL BLAST',11,4.00,5.8)
      CALL MESSAG('SHELL',4,6.0,5.8)
      CALL MESSAG('PERSONNEL',9,7.55,5.8)
      CALL MESSAG('PERSON NUMBER',13.25,25.25)
      CALL MESSAG('SERIAL',2.5,25)
      CALL MESSAG('EVENT',12.25,45)
      CALL MESSAG('ESTIMATED TIME PERIOD',16.25,C5)
      TS = FLCAT(IFIX(TIME(1)))
      TF = FLCAT(IFIX(TIME(1024)) + 1.1)
      CALL REALND(TS,2.2,4.05)

```

```

CALL REFLNC(TF,2,3,2,0,05)
CALL INTAXS
CALL YPARAMS(0,)
CALL XTICKS(10,)
CALL YTICKS(10)
CALL GRAVE(0,2000,0,10240,0,-1,0,0,2,1,0)
CALL CLAVE(UNIT,RMAI,10240,0)
CALL CASE
CALL CLFVE (XRAY1,YRAY,2,0)
CALL CLFVE (XRAY2,YRAY,2,0)
CALL CLFVE (XRAY3,YRAY,2,0)
CALL CLFVE (XRAY4,YRAY,2,0)
CALL ENCEPL(0)
RETURN
END
391
C
C SUBROUTINE MYDATA SUBTRACTS OFF THE SENSOR BIAS AND CREATES FROM THE
C INTEGER DATA ARRAY PASSED TO IT, REAL AND COMPLEX ARRAYS.
C
SUBROUTINE MYDATA(I,DATA,N,RA,A)
REAL T(N),RA(N)
INTEGER *2 DATA(N)
COMPLEX A(N)
DO 10 I=1,N
IDA=(DATA(I)-2048
IF (IDA.GT.2047 .OR. IDA.LT.-2048) IDA=0
RA(I)=FLCAT(IDA)
CONTINUE
SUM=0.0
DO 20 K=1,N
AIM=C*(CMPLX(RA(K),AIM))
SUM=SUM+RA(K)
CONTINUE
AVG=SUM/FLOAT(N)
CORRECTION FOR SWITCHING SPIKES IN DATA
RA(1) = 0
RA(2) = 0
RA(3) = C
RA(4) = C
RA(1021) = 0
RA(1022) = 0
RA(1023) = 0
RA(1024) = 0
RETURN
END

```

```

C
C
C
C
C
C
SUBROUTINE MAXMIN FINDS THE MAXIMUM AND MINIMUM VALUES FOR THE ARRAY
PASSED TO IT. IT ALSO RETURNS THE ELEMENT NUMBERS ASSOCIATED WITH
THE MAXIMUM AND MINIMUM VALUES.
C
C
C
C
SUBROUTINE MAXMIN(A,N,K,AMAX,L,AMIN)
REAL A(N)
AMAX = A(1)
AMIN = A(1)
K=1
L=1
DO 10 I=2,N
  IF ((A(I)).LE.(AMAX)) GO TO 20
  IF
    AMAX=A(I)
    K=I
  IF ((A(I)).GE.(AMIN)) GO TO 10
  IF
    AMIN=A(I)
    L=I
CONTINUE
IF ((ABS(AMIN)).GT.(ABS(AMAX))) AMAX = AMIN
RETURN
END
C
C
C
C
SUBROUTINE SPCTRM COMPUTES THE NORMALIZED POWER FOR THE COMPLEX ARRAY
PASSED TO IT.
C
C
C
C
SUBROUTINE SPCTRM(T,RA,F,A,N,PWR,PWRDB,PHASE)
REAL T(N),RA(N),F(N),PWR(N),PHASE(N)
COMPLEX A(N)
DO 10 K=1,N
  PWR(K)=CABS(A(K))*2
  AR=REAL(A(K))
  AI=AIMAG(A(K))
  IF (AR.EQ. C.O .AND. AI.EQ. 0.O) GC TO 40
  PPHASE(K)=ATAN2(AI,AR)
  GC TC 50
  PHASE(K)=0.O
CONTINUE
CONTINUE
N2=N/2
CALL MAXMIN(PWR,N,KF,PMAX,LP,PMIN)
NCRMALIZE POWER SPECTRUM
DO 60 J=1,N
  PWR(J)=PWR(J)/AES(PMAX)
CONTINUE
RETURN
END
C
C
C
C

```

```

C
C
C
C
C
SUBROUTINE MULT1 IS THE MULTIPLE TARGET DIRECTION ROUTINE.
DIRECTIONS ARE COMPUTED FOR UP TO FIVE TARGET CLASSES. FOR SINGLE
TARGETS, CAPTIVE MATCHED FILTERING IS PERFORMED ALLOWING FOR
IMPROVED ACCURACY.
      SUBROUTINE MULT1(RMAG1,RMAG2,RMAG3,RMAG4,RMAG5,RMAG6,RMAG7,
      1RMAG8,RMAG9,NTYP,SCALE,ICORR,THRES,ECAT,TESTNO,JL,C,TS,TF,
      2V,A,ND,AN,NUMSEN,TIME,ITARGET,DIRECT,CIA,DIRC)
      1INTEG,NTYP(6),C,NUMSEN
      1INTR,ND(4),TESTNO,C,NUMSEN
      1REAL,RMAG1(1024),RMAG2(1024),RMAG3(1024),RMAG4(1024),RMAG5(1024),
      1RMAG6(1024),RMAG7(1024),RMAG8(1024),RMAG9(1024),
      2BCAT(512),CCAT(512),TIME(1024),DIR(5),DIRC,
      3X(9),Y(5),THETA(9),T(9),V(4),A(4),RMAG(1024)
      NSEN = 1
      NH = NH/2
      DO 6 I1 = 1,512C
      CCAT(I1) = BCAT(I1)
      CONTINUE
      ADAPTIVE MATCHED FILTERING FOR A SINGLE TARGET IS ACCOMPLISHED
      BY USING THE SELECTED SENSOR'S SEISMIC DATA AS THE MATCHED FILTER
      WHEN COMPUTING THE IDENTIFIED TARGET'S DIRECTION.
      NTR = C
      DO 888 I14 = 1,4
      IF((V(I14)).NE.(0)) GC TC 5
      CONTINUE
      DO 377 JV = 1,5
      IF((NTYP(JV)).NE.(0)) NTR = NTR + 1
      CONTINUE
      IF((NTYP(1)).EQ.(0)) GC TU 1
      IF((NTR).NE.(1)) GC TC 1
      NTR = NTR + 1
      DO 151 I2 = 1,1024
      CCAT(I2) = RMAG1(I2)
      CONTINUE
      IF((NTYP(2)).EQ.(0)) GC TO 2
      IF((NTR).NE.(1)) GC TO 2
      NTR = NTR + 1
      DO 102 I3 = 1,1024
      CCAT(I3) = RMAG2(I3)
      CONTINUE
      IF((NTYP(3)).EQ.(0)) GC TO 3
      IF((NTR).NE.(1)) GC TO 3
      NTR = NTR + 1

```

```

PF105120
PF105140
PF105150
PF105160
PF105170
PF105180
PF105190
PF105200
PF105210
PF105220
PF105230
PF105240
PF105250
PF105260
PF105270
PF105280
PF105290
PF105300
PF105310
PF105320
PF105330
PF105340
PF105350
PF105360
PF105370
PF105380
PF105390
PF105400
PF105410
PF105420
PF105430
PF105440
PF105450
PF105460
PF105470
PF105480
PF105490
PF105500
PF105510
PF105520
PF105530
PF105540
PF105550
PF105560
PF105570
PF105580
PF105590
PF105600

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PF11C090
 PF11C100
 PF11C110
 PF11C120
 PF11C130
 PF11C140
 PF11C150
 PF11C160
 PF11C170
 PF11C180
 PF11C190
 PF11C200
 PF11C210
 PF11C220
 PF11C230
 PF11C240
 PF11C250
 PF11C260
 PF11C270
 PF11C280
 PF11C290
 PF11C300
 PF11C310
 PF11C320
 PF11C330
 PF11C340
 PF11C350
 PF11C360
 PF11C370
 PF11C380
 PF11C390
 PF11C400
 PF11C410
 PF11C420
 PF11C430
 PF11C440
 PF11C450
 PF11C460
 PF11C470
 PF11C480
 PF11C490
 PF11C500
 PF11C510
 PF11C520
 PF11C530
 PF11C540
 PF11C550
 PF11C560

```

112
NTYP(5) = 0
CALL MTIME(RMAG,NTYP,SCALE,II,DIRECT,THRES,CCAT,TESTNO,JL,C,NW,
1ITS,TF,TIME)
ITARGET(1,1) = NTYP(1)
ITARGET(1,2) = NTYP(2)
ITARGET(1,3) = NTYP(3)
ITARGET(1,4) = NTYP(4)
ITARGET(1,5) = NTYP(5)
DO 112 18 = 1,NH
RMAG(18) = RMAG2(LM - NH + 18)
CONTINUE = 0
NTYP(1) = 0
NTYP(2) = 0
NTYP(3) = 0
NTYP(4) = 0
NTYP(5) = 0
CALL MTIME(RMAG,NTYP,SCALE,II,DIRECT,THRES,CCAT,TESTNO,JL,C,NW,
1ITS,TF,TIME)
ITARGET(2,1) = NTYP(1)
ITARGET(2,2) = NTYP(2)
ITARGET(2,3) = NTYP(3)
ITARGET(2,4) = NTYP(4)
ITARGET(2,5) = NTYP(5)
DO 113 19 = 1,NH
RMAG(19) = RMAG3(LM - NH + 19)
CONTINUE = 0
NTYP(1) = 0
NTYP(2) = 0
NTYP(3) = 0
NTYP(4) = 0
NTYP(5) = 0
CALL MTIME(RMAG,NTYP,SCALE,II,DIRECT,THRES,CCAT,TESTNO,JL,C,NW,
1ITS,TF,TIME)
ITARGET(3,1) = NTYP(1)
ITARGET(3,2) = NTYP(2)
ITARGET(3,3) = NTYP(3)
ITARGET(3,4) = NTYP(4)
ITARGET(3,5) = NTYP(5)
DO 114 1A = 1,NH
RMAG(1A) = RMAG4(LM - NH + 1A)
CONTINUE = 0
NTYP(1) = 0
NTYP(2) = 0
NTYP(3) = 0
NTYP(4) = 0
NTYP(5) = 0
CALL MTIME(RMAG,NTYP,SCALE,II,DIRECT,THRES,CCAT,TESTNO,JL,C,NW,
1ITS,TF,TIME)

```

MF11C570
MF11C580
MF11C590
MF11C600
MF11C610
MF11C620
MF11C630
MF11C640
MF11C650
MF11C660
MF11C670
MF11C680
MF11C690
MF11C700
MF11C710
MF11C720
MF11C730
MF11C740
MF11C750
MF11C760
MF11C770
MF11C780
MF11C790
MF11C800
MF11C810
MF11C820
MF11C830
MF11C840
MF11C850
MF11C860
MF11C870
MF11C880
MF11C890
MF11C900
MF11C910
MF11C920
MF11C930
MF11C940
MF11C950
MF11C960
MF11C970
MF11C980
MF11C990
MF11C100
MF11C101
MF11C102
MF11C103
MF11C104
MF11C105

```

115      ITRGETI(4,1) = NIYP(1)
      ITRGETI(4,2) = NIYP(2)
      ITRGETI(4,3) = NIYP(3)
      ITRGETI(4,4) = NIYP(4)
      ITRGETI(4,5) = NIYP(5)
      DO 115 I8 = 1, NW
      RMAG(I8) = RMAG5(LM - NH + I8)
      CONTINUE
      NTYP(1) = 0
      NTYP(2) = 0
      NTYP(3) = 0
      NTYP(4) = 0
      NTYP(5) = 0
      CALL MATCH(RMAG,NTYP,SCALE,II,DIRECT,THRES,CCAT,TESTNO,JL,C,NW,
      1 I8,TIME)
      ITRGETI(5,1) = NIYP(1)
      ITRGETI(5,2) = NIYP(2)
      ITRGETI(5,3) = NIYP(3)
      ITRGETI(5,4) = NIYP(4)
      ITRGETI(5,5) = NIYP(5)
      DO 116 I16 = 1, NW
      RMAG(I16) = RMAG6(LM - NH + I16)
      CONTINUE
      NTYP(1) = 0
      NTYP(2) = 0
      NTYP(3) = 0
      NTYP(4) = 0
      NTYP(5) = 0
      CALL MATCH(RMAG,NTYP,SCALE,II,DIRECT,THRES,CCAT,TESTNO,JL,C,NW,
      1 I16,TIME)
      ITRGETI(6,1) = NIYP(1)
      ITRGETI(6,2) = NIYP(2)
      ITRGETI(6,3) = NIYP(3)
      ITRGETI(6,4) = NIYP(4)
      ITRGETI(6,5) = NIYP(5)
      IF((NUMSEN).EQ.16) GO TO 101
      DO 117 I17 = 1, NW
      RMAG(I17) = RMAG7(LM - NH + I17)
      CONTINUE
      NTYP(1) = 0
      NTYP(2) = 0
      NTYP(3) = 0
      NTYP(4) = 0
      NTYP(5) = 0
      CALL MATCH(RMAG,NTYP,SCALE,II,DIRECT,THRES,CCAT,TESTNO,JL,C,NW,
      1 I17,TIME)
      ITRGETI(7,1) = NIYP(1)
      ITRGETI(7,2) = NIYP(2)

```


MF111C50
MF111C60
MF111C70
MF111C80
MF111C90
MF111100
MF111110
MF111120
MF111130
MF111140
MF111150
MF111160
MF111170
MF111180
MF111190
MF111200
MF111210
MF111220
MF111230
MF111240
MF111250
MF111260
MF111270
MF111280
MF111290
MF111300
MF111310
MF111320
MF111330
MF111340
MF111350
MF111360
MF111370
MF111380
MF111390
MF111400
MF111410
MF111420
MF111430
MF111440
MF111450
MF111460
MF111470
MF111480
MF111490
MF111500
MF111510
MF111520

```

118 IIRGET(7,3) = NIYP(3)
IIRGET(7,4) = NIYP(4)
IIRGET(7,5) = NIYP(5)
DO 118 IE = 1,NA
RMAG(IE) = RMAG(LM - NH + IE)
CONTINUE
NTYP(1) = 0
NTYP(2) = 0
NTYP(3) = 0
NTYP(4) = 0
NTYP(5) = 0
CALL MATCH(RMAG,NTYP,SCALE,II,DIRECT,THRES,CCAT,TESTNO,JL,C,NW,
1 IIS,TIME)
IIRGET(8,1) = NIYP(1)
IIRGET(8,2) = NIYP(2)
IIRGET(8,3) = NIYP(3)
IIRGET(8,4) = NIYP(4)
IIRGET(8,5) = NIYP(5)
DO 119 IF = 1,NA
RMAG(IF) = RMAG9(LM - NH + IF)
CONTINUE
NTYP(1) = 0
NTYP(2) = 0
NTYP(3) = 0
NTYP(4) = 0
NTYP(5) = 0
CALL MATCH(RMAG,NTYP,SCALE,II,DIRECT,THRES,CCAT,TESTNO,JL,C,NW,
1 IIS,TIME)
IIRGET(9,1) = NIYP(1)
IIRGET(9,2) = NIYP(2)
IIRGET(9,3) = NIYP(3)
IIRGET(9,4) = NIYP(4)
IIRGET(9,5) = NIYP(5)
NTYP(1) = 0
NTYP(2) = 0
NTYP(3) = 0
NTYP(4) = 0
NTYP(5) = 0
DC 675 IF = 1,5
DIR(IN) = 955.
CCNTINCE
CCRETURN
END
679
101
C
C
C
MULTIPLE DIRECTIONAL OUTPUT PLACING
SUBROUTINE MULTPLTIC SCALE,TESTNO,IS,IF,CIR,V,A,WD,IIRGET)
INTEGER C,NC(4),IIRGET(5,5)

```

```

REAL CIF(5),V(4),A(4)
IF((C).EQ.(1)) CALL CCMPRS
IF((C).NE.(1)) CALL TEK618
CALL PACERDR
CALL BLCDRUP(SCALE)
CALL AREA2D(10,C,6,C)
CALL FRAME
CALL HEADSAG(,EVENT NUMBER,35,2.0,1)
CALL INTNG(,TESTING,5.0,5.8)
CALL MESSAG(,TIME,PERIOD(SEC),16,3.0,5.4)
CALL REALNG(ITS,2.4,5.5,4)
CALL REALNG(ITS,2.5,7.5,4)
IF(((1)TRGET(1,1)).NE.(0)).AND.((DIR(1,1)).NE.(999.1))
1 IF(((1)TRGET(1,1)).NE.(0)).AND.((DIR(1,1)).NE.(999.1))
1 IF(((1)TRGET(1,2)).NE.(0)).AND.((DIR(2,1)).NE.(999.1))
1 IF(((1)TRGET(1,2)).NE.(0)).AND.((DIR(2,1)).NE.(999.1))
1 IF(((1)TRGET(1,3)).NE.(0)).AND.((DIR(3,1)).NE.(999.1))
1 IF(((1)TRGET(1,3)).NE.(0)).AND.((DIR(3,1)).NE.(999.1))
1 IF(((1)TRGET(1,4)).NE.(0)).AND.((DIR(4,1)).NE.(999.1))
1 IF(((1)TRGET(1,4)).NE.(0)).AND.((DIR(4,1)).NE.(999.1))
1 IF(((1)TRGET(1,5)).NE.(0)).AND.((DIR(5,1)).NE.(999.1))
1 IF(((1)TRGET(1,5)).NE.(0)).AND.((DIR(5,1)).NE.(999.1))
1 CALL MESSAG(,SIMULATED TRK VEHICLE TARGET FREQUENCY,35,3.0,2.8)
CALL REALNG(,1,2,6.0,2.8)
CALL REALNG(,AMPLITUDE,5.4,0.2,6)
CALL MESSAG(,DIRECTION,9,4.0,2.4)
CALL REALNG(,1,4,5.5,2.6)
D = FLCAT(NE(1))
CALL REALNG(,4,5.5,2.4)
CALL MESSAG(,SIMULATED WHLD VEHICLE TARGET FREQUENCY,35,3.0,2.2)
CALL REALNG(,2,2,8.0,2.2)
CALL MESSAG(,AMPLITUDE,9,4.0,2.0)
CALL MESSAG(,DIRECTION,9,4.0,1.80)
D = FLCAT(NE(2))
CALL REALNG(,4,5.5,1.8)

```

```

MF1111530
MF1111540
MF1111550
MF1111560
MF1111570
MF1111580
MF1111590
MF1111600
MF1111610
MF1111620
MF1111630
MF1111640
MF1111650
MF1111660
MF1111670
MF1111680
MF1111690
MF1111700
MF1111710
MF1111720
MF1111730
MF1111740
MF1111750
MF1111760
MF1111770
MF1111780
MF1111790
MF1111800
MF1111810
MF1111820
MF1111830
MF1111840
MF1111850
MF1111860
MF1111870
MF1111880
MF1111890
MF1111900
MF1111910
MF1111920
MF1111930
MF1111940
MF1111950
MF1111960
MF1111970
MF1111980
MF1111990
MF1112000

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C C C
CALL MESSAG('SIMULATED HELICOPTER TARGET FREQUENCY',37,3.0,1.0)
CALL REALNDC(3,2,8.0,1.6)
CALL MESSAG('AMPLITUDE',9,4.0,1.4)
CALL MESSAG('DIRECTION',9,4.0,1.2)
CALL REALNDC(3,4,5.5,1.4)
D = FLOAT(NDC(3))
CALL REALNDC(4,5,5.1,2)
CALL MESSAG('SIMULATED PERSONNEL TARGET FREQUENCY',36,3.0,1.0)
CALL REALNDC(4,2,8.0,1.0)
CALL MESSAG('AMPLITUDE',9,4.0,0.8)
CALL MESSAG('DIRECTION',9,4.0,0.6)
CALL REALNDC(4,4,5.5,0.8)
D = FLOAT(NDC(4))
CALL REALNDC(4,5,5.0,6)
CALL ENCLPL(0)
RETURN
END

C C C
SUBROUTINE RMS(COMPUTES THE ROOT MEAN SQUARE VALUE OF THE N-ARRAY
DATA PASSED TO IT)
  SUBROUTINE RMS(A,N,TMAX)
    REAL A(N)
    TMAX = C
    DO 200 I = 1,N
      TMAX = (A(I))**2 + TMAX
    CONTINUE
    TMAX = SQRT(TMAX)
    RETURN
  END

200

C C C
SUBROUTINE AVG(COMPUTES THE AVERAGE VALUE OF THE N-ARRAY DATA PASSED
TO IT)
  SUBROUTINE AVG(A,N,AAVG)
    REAL A(N)
    AAVG = C
    DO 100 I = 1,N
      AAVG = A(I) + AAVG
    CONTINUE
    AAVG = AAVG/FLOAT(N)
    RETURN
  END

100

C C C
SUBROUTINE SIMULATE ALLOWS FOR THE MODIFICATION OF EXPERIMENTAL
DATA TO INCLUDE UP TO FOUR SIMULATED TARGETS, AND MODIFICATION OF
SEISMIC SIGNAL AMPLITUDES ABOVE THE SELECTED NOISE LEVEL.

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SUBROUTINE SIMULT(RA1, RA2, RA3, RA4, RA5, RA6, RA7, RA8, RA9,
 1A1, A2, A3, A4, A5, A6, A7, A8, A9,
 2V, A, INC, TIME, JL, P, REDUCE, NUMSEN)
 IN, TEG, RND(4), NUMSEN
 1RA7(1024), RA8(1024), RA2(1024), RA3(1024), RA4(1024), RA5(1024), RA6(1024),
 2V(4), A(4), TIME(1024), P(9,4), REDUCE
 COMPLEX A1(1024), A2(1024), A3(1024), A4(1024), A5(1024),
 1A6(1024), A7(1024), A8(1024), A9(1024)
 WRITE(6,10) *ENTER THE FOUR SIMULATION FREQUENCIES-REAL-*)
 10 FORMAT(10,1) *ENTER FREQUENCY*)
 WRITE(6,11) *ENTER FREQUENCY*)
 11 FORMAT(10,12)V(1)
 READ(6,12)V(2)
 WRITE(6,13)V(2)
 READ(6,14)V(3)
 WRITE(6,15)V(3)
 READ(6,16)V(4)
 WRITE(6,17)V(4)
 12 FORMAT(10,12)V(4)
 WRITE(6,18) *ENTER AMPLITUDES FOR EACH FREQUENCY-REAL-*)
 13 FORMAT(10,14) *ENTER AMPLITUDE*)
 14 FORMAT(10,15)A(1)
 READ(6,16)A(2)
 WRITE(6,17)A(2)
 READ(6,18)A(3)
 WRITE(6,19)A(3)
 READ(6,20)A(4)
 WRITE(6,21)A(4)
 15 FORMAT(10,15)A(4)
 WRITE(6,22) *ENTER TARGET ANGLE FOR EACH FREQUENCY-13-*)
 16 FORMAT(10,17) *ENTER ANGLE*)
 18 FORMAT(10,17) *SIX SENSORS ALLOWABLE ANGLES:0,60,120,180,240,300,*)
 19 FORMAT(10,17) *FOR NINE SENSORS:0,40,80,120,160,200,240,280,320,*)
 17 FORMAT(10,17) *ENTER ANGLE*)
 READ(6,25)ND(1)
 WRITE(6,26)ND(1)
 READ(6,27)ND(2)
 WRITE(6,28)ND(2)
 READ(6,29)ND(3)
 WRITE(6,30)ND(3)
 READ(6,31)ND(4)
 WRITE(6,32)ND(4)

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IF(AC(I)).NE.(320)) GO TO 41
P(1:I) = -21.3
P(2:I) = -76.7
P(3:I) = -138.7
P(4:I) = -180.
P(5:I) = -138.7
P(6:I) = -76.7
P(7:I) = -21.3
P(8:I) = 0.
P(9:I) = 0.

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ADDITION OF THE SIMULATED SIGNALS TO EACH SENSORS DATA

```

DO 2 J = 1,1024
  RA1(J) = 1.024
  RA2(J) = RA1(J)
  RA3(J) = RA2(J)
  RA4(J) = RA3(J)
  RA5(J) = RA4(J)
  RA6(J) = RA5(J)
  IF(ALMSEN).EQ.(6)) GO TO 2
  RA7(J) = RA6(J)
  RA8(J) = RA7(J)
  RA9(J) = RA8(J)
CONTINUE
CONTINUE

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SUPPRESSION OR ENHANCEMENT OF SEISMIC SIGNALS ABOVE THE NOISE LEVEL

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DO 3 K = 1,1024
  IF(AES(RA1(K))) .GT. (TNOISE)) RA1(K) = RA1(K)/REDUCE
  IF(AES(RA2(K))) .GT. (TNOISE)) RA2(K) = RA2(K)/REDUCE
  IF(AES(RA3(K))) .GT. (TNOISE)) RA3(K) = RA3(K)/REDUCE
  IF(AES(RA4(K))) .GT. (TNOISE)) RA4(K) = RA4(K)/REDUCE
  IF(AES(RA5(K))) .GT. (TNOISE)) RA5(K) = RA5(K)/REDUCE
  IF(AES(RA6(K))) .GT. (TNOISE)) RA6(K) = RA6(K)/REDUCE
  IF(ALMSEN).EQ.(6)) GO TO 3
  IF(AES(RA7(K))) .GT. (TNOISE)) RA7(K) = RA7(K)/REDUCE
  IF(AES(RA8(K))) .GT. (TNOISE)) RA8(K) = RA8(K)/REDUCE
  IF(AES(RA9(K))) .GT. (TNOISE)) RA9(K) = RA9(K)/REDUCE

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CONTINUE
DO 20 KI = 1,1024
  ABC = 0.
  A1(KI) = CMPLX(RA1(KI),ABC)
  A2(KI) = CMPLX(RA2(KI),ABC)
  A3(KI) = CMPLX(RA3(KI),ABC)
  A4(KI) = CMPLX(RA4(KI),ABC)
  A5(KI) = CMPLX(RA5(KI),ABC)

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20  A6(K1) = CMPLX(RA6(K1),AEC)
30  IF((ALMSEN).EQ.(6)) GO TO 20
    A7(K1) = CMPLX(RA7(K1),AEC)
    A8(K1) = CMPLX(RA8(K1),AEC)
    A9(K1) = CMPLX(RA9(K1),AEC)
    CONTINUE
    RETURN
    ENC
C
C
C   LEAST MEAN SQUARES POLYNOMIAL CURVE FITTING ALGORITHM
C
SUBROUTINE LMS (NTAR,DIRC,NUMSEN,C,SCALE,TESTINC,TIME,NFRM,Y,
1 VX,X,T,INDEX,NCR,TRIX(5,9),B(5),WKAREA(108)
DIMENSION NTRIX(9),T(9),C,NHALF,MIN,NUMSEN,TESTINC,NFRM
INTEGER NTRIX(9),T(9),X(9),A(9),Y(360),
REAL P,FMIN,S,XS(16),X(9),Y(360),
1 VX(360),T(9),TIME(1024),DIRC,SCALE,C
C
C   FIND FIRST SENSOR TO MATCH THE INPUT SIGNAL
C
MIN = NTRIX(1)
MARK = 1
IF((NTRIX(1)).LT.(1)) GO TO 66
DO 51 I17 = 1,9
1(I17) = 0
IF((I17) = 0)
1(I17) = 0
WRITE(6,53) NTRIX(17)
CONTINUE
FCRMAT(17)
DO 11 I12 = 2,NUMSEN
IF((NTRIX(12)).GE.(MIN)) GO TO 11
MIN = NTRIX(12)
MARK = 12
CONTINUE
DO 46 IE = 1,360
Y(IE) = C
VX(IE) = 0.
CONTINUE
C
C   CENTER THE PARABOLA
C
IF((MIN).EQ.(0)) GO TO 66
DO 21 K1 = 1,NUMSEN
INA = 1
IF((NUMSEN).EQ.(6)) INA = 0
N = NUMSEN/2 - MARK + K1 + INA
IF((N).GT.(NUMSEN)) N = N - NUMSEN
IF((N).LT.(1)) N = N + NUMSEN

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T(N) = NTAR(K1)
CONTINUE
DO 41 L = 1, NUMSEN
  TT(L) = T(L) - MIN
CONTINUE
DO 40 I = 1, 16
  XS(I) = 0.
CONTINUE
X(1) = NUMSEN - 1
LN = LN
DO 130 I = 1, LN
  IF((NUMSEN).EQ.(6)) X(14 + I) = (FLCAT(14*60))/360.
  IF((NUMSEN).EQ.(9)) X(14 + I) = (FLCAT(14*40))/320.
CONTINUE

LOAD THE MATRICES FOR THE LEAST MEAN SQUARES POLYNOMIAL SOLUTION

IF (NUMSEN.EQ.6) X(7) = 0
IF (NUMSEN.EQ.6) X(8) = 0
IF (NUMSEN.EQ.6) X(9) = 0
NTC = 2 + NUMSEN - 2
DO 20 A1 = 1, NTC
  DO 40 K3 = 1, NUMSEN
    X(NA1) = X(K3)*#N1 + XS(N1)
CONTINUE
CONTINUE
DO 47 IZA = 1, 9
  CO 45 I3A = 1, 9
  TRIXX(I2A, I3A) = 0.
CONTINUE
CONTINUE
TRIX(1, 1) = NUMSEN
DO 50 A2 = 2, NUMSEN
  TRIX(1, A2) = XS(N2 - 1)
CONTINUE
CO 40 IIZ = 1, NUMSEN
  TRIX(I2, I5) = XS(I5) + 1
  TRIX(I3, I5) = XS(I5) + 2
  TRIX(I4, I5) = XS(I5) + 3
  TRIX(I5, I5) = XS(I5) + 4
  TRIX(I6, I5) = XS(I5) + 5
  IF((ALHSEN).EQ.(6)) GC TC 40
  TRIX(I7, I5) = XS(I5) + 6
  TRIX(I8, I5) = XS(I5) + 7
CONTINUE
DO 60 IC = 1, 9

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60      B(I6) = 0
      CONTINUE
      DO 70 I = 2, NUMSEN
      CO EC K4 = 1, NUMSEN
      B(I7) = B(I7) + (X(K4)**(I7 - 1)) * (FLCAT(T(K4)))
      CONTINUE
      DO 80 I = 1, NUMSEN
      B(I) = B(I) + (FLCAT(T(I8)))
      CONTINUE
      WRITE(6,1)
      WRITE(6,2)
      FORMAT(10)
      ENTER DEGREE OF POLYNOMIAL DESIRED -11-
      1  FORMAT(10)
      2  FOR SIX SENSORS ENTER 2 - 4, FOR NINE ENTER 2 - 7
      3  REAC(10,3) NOR
      4  FORMAT(11)
      5  NZ = NCR + 1
      6  CALL THE MATRIX SOLVER LEGT2F VIA SUBROUTINE SOLV TO FIND THE
      7  COEFFICIENTS FOR THE LEAST MEAN SQUARES POLYNOMIAL
      8  CALL SOLV (TRIX,NZ,B)
      9  PMIN = 1E20
      10  MINI = 1
      11  DIVICE = 40
      12  IF((NUMSEN).EQ.(6)) DIVIDE = 60
      13  NO = NCR + 2 NO,9
      14  DO 110 IS = NO,9
      15  B(IS) = 0
      16  CONTINUE
      17  IF((NUMSEN).EQ.(6)) INDEX = 360
      18  IF((NUMSEN).EQ.(9)) INDEX = 320
      19  CALCULATE PLCTING DATA FROM THE LEAST MEAN SQUARES POLYNOMIAL
      20  DO 120 NR = 1, INDEX
      21  Q = FLCAT(NR)/FLCAT(INDEX)
      22  P = E(1) + B(2)*(Q) + B(3)*(Q**2)
      23  1 + B(4)*(Q**3)
      24  2 + B(5)*(Q**4)
      25  3 + B(6)*(Q**5)
      26  4 + B(7)*(Q**6)
      27  5 + B(8)*(Q**7)
      28  6 + B(9)*(Q**8)
      29  VX(NR) = FLOCAT(NR) + FLOCAT(MARK - NUMSEN/2 - INA)*DIVIDE -1.
      30  Y(NR) = P
      31  IF((P).GT.(PMIN)) GC TC 120
      32  MINI = NR
      33  END

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CALL YAXANG(0.)
CALL XTICKS(10)
CALL YTICKS(10)
TS = VXR(1), INDEX
TF = VXR(INDEX), INDEX, K, YMAX, L, YMIN)
CALL MAXMIN(Y, YMIN, YMAX)
IF ((YMAX).LE.(YMIN)) YMAX = - YMAX
CALL MAXMIN(TTB, NUMSEN, K, TMAX, L, TMIN)
IF ((TMAX).LE.(TMIN)) TMAX = - TMIN
IF ((TMAX).GT.(YMAX)) YMAX = TMAX
IF ((TMIN).LT.(YMIN)) YMIN = TMIN
CALL GRAPT(TS, SCALE, TF, YMIN, SCALE, YMAX)
CALL RESET (MARKER)
CALL CLEVER (VXR, Y, INDEX, 0.)
CALL MARKER(10)
CALL CLEVE (BX, TTB, NUMSEN, -1)
CALL RESET (MARKER)
CALL ENDPL (0)
RETURN
END

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CC

MATRIX SOLVING ROUTINE

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SUBROUTINE SOLV (TRIX, NZ, B)
DIMENSION TRIX(5,9), B(5), WKAREA(108)
CALL LECT2F (TRIX, 1, NZ, 5, B, 0, WKAREA, IER)
RETURN
END

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